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DEPARTMENT OF MECHANICAL ENGINEERING AND MECHANICS
SCHOOL OF ENGINEERING
OLD DOMINION UNIVERSITY
NORFOLK, VIRGINIA

THE DEVELOPMENT OF A MICROPROCESSOR-CONTROLLED
LINEARLY-ACTUATED VALVE ASSEMBLY

By

Raymond H. Wall

Robert L. Ash, Principal Investigator

Supplemental Technical Report

Prepared for the
Jet Propulsion Laboratory
California Institute of Technology
Pasadena, California 91103

Under
Contract No. 955731
Warren L. Dowler, Technical Manager

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P.O. Box 6369
Norfolk, Virginia 23508



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By

Raymond H. Wall

Under the direction of Dr. Robert L. Ash, Principal Investigator and
Dr. Ping Tchong, NASA/Langley Research Center, Hampton, VA

INTRODUCTION

The major part of this report is comprised of a master's thesis (see Appendix), written by Mr. Raymond H. Wall, which investigates the feasibility of a direct, digitally-controlled valve for space system operation. The thesis is relatively self-explanatory. It is important to note, however, that due to severe cost limitations, many of the technology issues which could be solved by modest levels of funding, could not be solved at the level of funding available for this work.

In addition to the thesis, we would like to submit the discussion that follows, concerning technology issues in space. We would also recommend, in addition to the documentation provided with this report, the inclusion of the following three reference documents for our report:

- Item 1: Document 74A39119 entitled "Pulse-Modulated Dual-Gas Control Subsystem for Space Cabin Atmosphere," by J.K. Jackson. Summarizes research on a pulse modulated valve for space system application.
- Item 2: Report No. 75N22744 entitled "Final Report: Line Fluid Activated Valve Development Program," by R. A. Lynch of the Marquardt Corporation, Van Nuys, California.
- Item 3: Document No. 75N24839 entitled "Space Shuttle Main Engine Definition (Phase B) Volume 5: Valves and Interconnects -- for Space Shuttle." This is a Pratt and Whitney Report.

These documents, which are available in the library of the California Institute of Technology, should be included as a package of material addressing technology issues for continuously controllable valve systems.

OBJECTIVES

One of the objectives of this program was to determine the feasibility of avoiding A to D conversion elements in this valve system. Based on the work that followed, it is believed that the control aspects can be accomplished directly, but the pressure measuring aspects of the system remain driven by analog systems with digital conversion.

One problem which will continue to be an issue is absolute position location using digital encoder systems. A recent development called a "gray band encoder" provides an absolute position location rather than a digital output (driven typically by counting black and white bands) and appears to be the preferred mode. Gray band encoders are presently being used by the Army in some of the helicopter development work at Ft. Eustis. While the gray band encoder is not a classified device, apparently the application is classified, and we were not able to determine the precise specifications for gray band position encoders. Digital displacement measurement devices are an essential technology for any system of this type. A digital displacement measuring device can be used both for pressure measurements as well as valve stem position measurements. These devices have not yet been developed at a level which would be acceptable for autonomous operation in space. State of the art linear encoder devices appear to be too large and too sensitive to drift in absolute position due to vibration.

OPINIONS AND CONCLUSIONS

It was clearly the consensus of the three investigators in this project that bellows valves are the preferred valve for space applications. These valves must be custom designed for each application because of the need to match both the bellows spring constant and the valve stem travel to the particular displacement system and flow application. Longer valve stem travel is needed for high flow rate applications than presently is available on commercial valve systems. The actual valve geometry must be a major consideration in the space system design. The geometry affects performance, reliability, repeatability, and survivability. In addition, proper valve design may enable a relatively high level of self calibration capability.

Continuing calibration of the valve/flow control system was identified as an essential element in a long duration valve operation. Presently there are no easy calibration procedures which could be used either continuously, or intermittently in conjunction with the valve and associated differential pressure measuring systems.

The consensus of this group was that the pressure measurements across an orifice plate which was not maintained directly in the flow stream was a viable means of calibrating intermittently the flow control system. The idea would be to have a bypass system whereby the flow could be bypassed through an orifice plate with the corresponding pressure measurements being made for short periods of time. In that way, neither corrosion nor contamination would occur at the same rate as it would occur in the continuously operated stream. By using the calibration loop intermittently, it is possible to maintain an updated calibration for the valve system.

At the conclusion of the thesis, it became apparent that the next step beyond this indirect control approach would be to develop a calibration

relation by appropriate choice of parameters which collapsed the data onto a single curve. That is, the different parametric calibration curves presented in the thesis could be collapsed onto a single curve (to some reasonable level of approximation) and greatly reduce the computation time required in the software. That approach would be feasible if variations of up to 5% in flow are tolerated by the system.

It was determined further that the flow rate measurement, or at least the flow rate calibration, should be separated from direct operation of the valve. Problems with long term degradation of valve geometry and valve performance cannot be identified if both are degrading simultaneously.

SUMMARY

In summary, no technology issues were found which prevent the development of digital valve systems for space applications. The single, most limiting technology, however, appears to be a direct digital displacement measuring system which is not affected seriously by vibration. The valve system developed in this thesis was found to work autonomously and satisfactorily. It is hoped that the information developed in this work, including the micro computer programs, will be useful at the start of a more ambitious valve research program in the future.

APPENDIX

THE DEVELOPMENT OF A MICROPROCESSOR
CONTROLLED LINEARLY ACTUATED VALVE ASSEMBLY

by

Raymond H. Wall
B.S. (Mechanical Engineering)
August 1981, Old Dominion University

A Thesis Submitted to the Faculty of
Old Dominion University in Partial Fulfillment of
the Requirements for the Degree of

MASTER OF ENGINEERING
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OLD DOMINION UNIVERSITY
May 1984

Approved by:

Ping Tcheng (Director)

Robert L. Ash

ABSTRACT

THE DEVELOPMENT OF A MICROPROCESSOR CONTROLLED LINEARLY ACTUATED VALVE ASSEMBLY

Raymond H. Wall
Old Dominion University, 1984
Director: Dr. Ping Tcheng with
Dr. Robert L. Ash

The development of a proportional fluid control valve/assembly is presented. This electro-mechanical system is needed for space applications to replace the current proportional flow controllers that use bulky hydraulic or pneumatic actuation systems. The flow is controlled by a microprocessor system that monitors the control parameters of upstream pressure and requested volumetric flow rate. The microprocessor achieves the proper valve stem displacement by means of a digital linear actuator. A linear displacement sensor is used to measure the valve stem position. This displacement is monitored by the microprocessor system as a feedback signal to close the control loop. With an upstream pressure between 15-47 psig, the developed system operates between 779 standard cm³/sec (SCCS) and 1543 SCCS. The delivered volumetric flow rates were within 5% of the requested values. Recommendations made for future work on the design include further modification to the bellows spring and the possibility of a mass flow rate controller.

ACKNOWLEDGEMENT

The author wishes to express his appreciation to his advisors, Dr. Ping Tchong of NASA Langley Research Center and Dr. R. L. Ash at Old Dominion University, for their advice and assistance in this investigation. Also, thanks to Dr. A. Sidney Roberts, Jr. for his guidance and support through the Aeronautics Program. A special thanks is expressed to Mrs. Diane Pizzeck for her help, suggestions, and the typing of this manuscript. The author offers his gratitude to his family and Ms. Robin Wilkinson for their continuing support during this development.

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LIST OF SYMBOLS

E_s	incremental optical linear encoder output
L_s	linear variable differential transformer output
P_u	upstream pressure
P_a	atmospheric pressure
Δp	differential pressure across regulating valve
Q_o	output volumetric flow rate
Q_r	requested volumetric flow rate
x_s	valve stem displacement (steps)

Chapter 1

INTRODUCTION

The control of fluid flow through a valve can be accomplished by either on-off or proportional control techniques. A survey of commercially available equipment for either method reveals limitations in the use of such technology designed for space systems. Examples of on-off type flow controllers would be solenoid valves or explosively actuated valves. Proportional flow control valves would typically be gate valves.

One limitation of an on-off controller is the inability to vary flow rate through the valve. Assuming a constant differential pressure, the fluid must be diverted through a second valve of different orifice size to change the flow rate. This requires an array of solenoid valves for a modest number of flow rate changes. This is prohibitive due to the mass and size restraints on a space system. Proportional flow controllers using gate or globe valves actuated by hydraulic systems, are also unacceptable because of the mass and size restraints on a space system. Additionally, the potential for leaks in hydraulic systems would eliminate their use in space.

A proportional pulse-modulated flow controller was reported by Jackson [1]*. The system achieved proportional control by pulsing solenoid valves at specific rates to maintain constant concentrations of nitrogen and oxygen in a Space Shuttle Orbiter cabin.

* The numbers in brackets indicate references.

While Jackson's results showed it was possible to approximate proportional flow control by discrete-time operation of the solenoid, such a device is considered unreliable on long term space missions. That is, the cycling of the valve for extended times would eventually wear down the valve seat and thereby change the flow rate characteristics.

Another, commercially available, device is a digital flow controller developed by the Porter Instrument Company. Their device uses a spring and a diaphragm. Assuming a constant upstream pressure, a change in downstream pressure is compensated by a change in the force the spring exerts on the diaphragm. Thus, changes in the differential pressure across the controller are compensated and a constant flow rate is maintained. However, the device is designed for operation in a controlled laboratory environment and could not function in a space system where temperature and pressure variations can be extreme.

The purpose of this thesis is to describe the development of a prototype electro-mechanical proportional flow control system for use in a space environment. A microprocessor was used in the system to monitor system parameters and digitally control the valve stem displacement. Currently, to the author's knowledge, there exists no similar system or design, even though the necessary technology has existed for several years. The equipment used in this development is described in Chap. 2. A description of the development is presented in Chap. 3. The results of the system final testing are presented and discussed in Chap. 4. Conclusions and recommendations on those areas of the development that need further work are in Chap. 5.

Chapter 2

DESCRIPTION OF EQUIPMENT

The control of volumetric flow rate through a regulating valve can be accomplished by: (1) An accurate and reliable means to move the valve stem; and (2) a method to measure differential pressure across the valve. The equipment listed below was used to accomplish flow control and has been divided into two categories: (1) Items which were preselected (by others) as the basic components for the preliminary investigation of the problem (designated by an asterisk); and (2) items which were added to the preliminary system during its development, which will be described by this thesis. A summary of the equipment used can be found in Table 1. A more detailed description of each component can be found in the following pages.

1. *Bellows Sealed Regulating Valve, NUPRO Model SS-4BRG. This type of valve (bellows sealed) was chosen because: (1) it eliminates the possibility of fluid leaks which can not be tolerated by space systems; (2) it eliminates the need for wad packing and the accompanying need to tighten the packing nut periodically; and finally, (3) because the bellows design readily adapts to the modifications which convert the valve movement from rotary to translational. (See Fig. 1 for a cross section view of the modified valve).
2. *Digital Linear Actuator, AIRPAX Model L92421-P2. After a survey of the commercially available linear movement

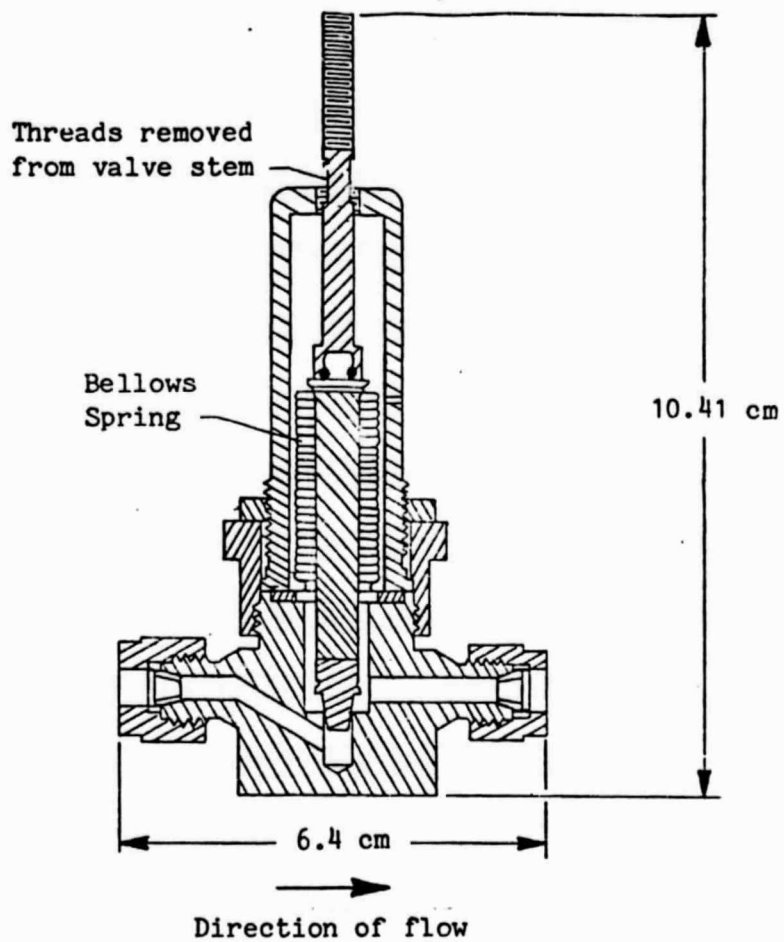


Fig. 1. Cross section of modified bellows sealed regulating valve.

actuators [2], the AIRPAX model 192421-P2 was selected on the basis of size and the magnitude of the linear force it could generate. The actuator is a modified stepping motor that has had its rotary motion converted into a linear force. The linear actuator has a resolution of 0.0051 cm/ step and a maximum linear force of 84.5 Newtons (N) at the slowest stepping rate.

3. *COSMAC Microprocessor Evaluation System, RCA CDP18S025. The RCA 1802 microprocessor was chosen to simulate the controlling computer in a spacecraft for several reasons: (1) The microprocessor evaluation system is built entirely of Complementary Metal-Oxide Silicon (CMOS) logic circuits, which means it uses considerably less power than the Transistor-Transistor Logic (TTL) circuits [3]; (2) the 1802 microprocessor is space qualified; and (3) the 1802 microprocessor is designed to be a process control device; it is not designed to be a support system for a personal computer like many other commercially available microprocessors. The Evaluation System has an eight bit processor with a 4K byte Random Access Memory (RAM) and 16 storage registers [4]. The system used in this development was modified to give it a battery backup RAM, in case of power failure, to prevent the loss of any programs stored in the memory. The RCA CDP18S021 COSMAC Microterminal is a fully assembled, compact, hand-held terminal designed for standard control, communications, and debugging functions

with CDP1802 microprocessor systems.

4. Barocel Electronic Manometer, DATAMETRICS Model 1174. The manometer measures the differential pressure across the valve and outputs a digital display of pressure (in psig) and an output in DC voltage between zero and one volt full scale.
5. Linear Variable Differential Transformer (LVDT), G. L. COLLINS CORP. Model LMT SS-207, S/N5240. By knowing the proportional relationship between flow coefficient and the displacement of the valve stem, the voltage readings from the LVDT can be used as one of the parameters necessary to control the volumetric flow rate through the valve. This was later replaced by the incremental optical linear encoder to eliminate the need for an analog to digital converter (ADC).
6. Turbometer, COX Model GM 16. This was used initially to measure volumetric flow rate through the valve but was later found to be inaccurate and was not used in the development. However, because it did not degrade the performance of the system, it was not removed from the assembly.
7. Rotameter, BROOKS CO. Model R9M-25-1. The rotameter was put in the system downstream as a means to check the accuracy of the turbometer. After the turbometer was found to be inaccurate, the rotameter was used as the only volumetric flow rate measurement device.
8. Globe Valve. Initially it was assumed that the differential pressure variation should be the control parameter in the development and so the valve was added to the downstream

assembly (see Fig. 2 for a photograph of the downstream assembly). While differential pressure is the correct control parameter, upstream pressure was substituted due to the lack of resolution in the Manometer Interface Circuit. Measurements were made with the downstream pressure tap of the manometer open to atmospheric conditions. Consequently, the valve was not necessary to the development but was left in the downstream assembly for the same reason as the turbometer.

9. Incremental Optical Linear Encoder (IOLE), DYNAMICS RESEARCH CORP. Model SST-E-4-1-0. Since the microprocessor is not suited to read the analog signal output by the LVDT, it was necessary to either build an ADC or to replace the LVDT with a digital linear measurement device. The linear encoder has a resolution of 1.016×10^{-3} cm/cycle and has a square wave quadrature output (90° out of phase) that is TTL compatible. An external circuit was necessary to read the output waveforms. The circuit diagram is shown in Fig. A.1, Appendix A, entitled IOLE Interface Circuit.
10. Linear Actuator Driver Board, AIRPAX Model K33505. This modified printed circuit board is the main component in the Interface Box (see Fig. A.2(a), Appendix A) between the microprocessor system and the linear actuator. It takes the clock signal generated from the microprocessor and then processes the signals necessary to drive the linear actuator. A more complete description of the process is

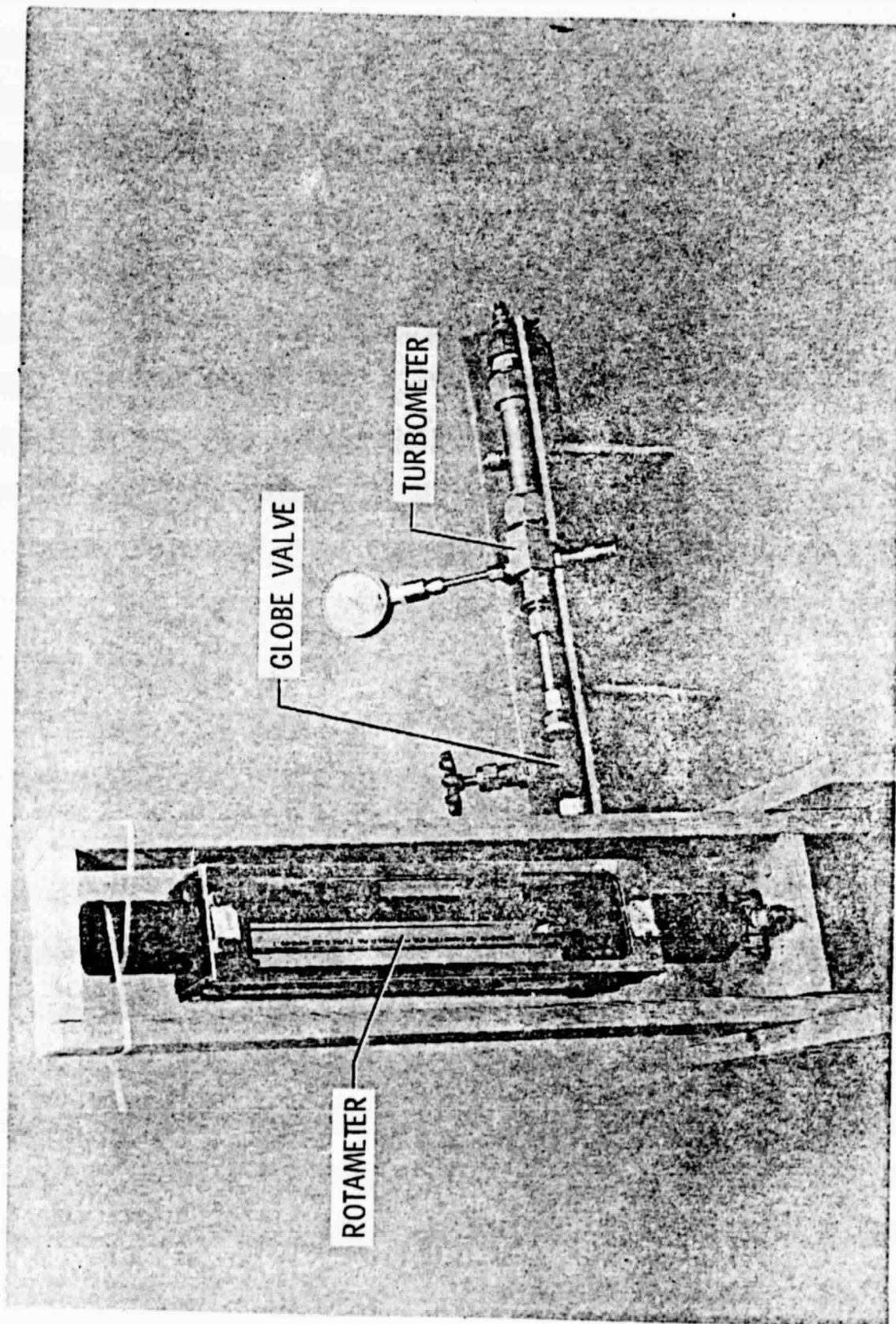


Fig. 2 Photograph of downstream assembly.

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given in sec. 3.2.1.

11. Bridgesensor, CALEX Model 166. Contained within the Model 166 Bridgesensor is a voltage to frequency (V/F) converter. By using this isolated component of the bridgesensor, in conjunction with the Manometer Interface Circuit shown in Fig. A.3, Appendix A, it was possible to convert the analog signal output from the manometer into a digital signal usable by the microprocessor.
12. Pressure Regulator, PARKER-HANNIFIN Model R2037/J5. During testing, a range of upstream pressures were desired. This required the installation of the regulator. The regulator has a 0-689.5 kilo Pascals (kPa) gage range.
13. Diverting Valve, CLIPPARD Model MVD-4J. When a signal is sent from the microprocessor system to the attached solenoids (see equipment item No. 14), the spool of the diverting valve changed the path of air flow from the exit port leading to the regulating valve, to the plugged exit port. This allowed the tubing upstream of the regulating valve to depressurize. (See Fig. 3 for a photograph of the upstream assembly).
14. Solenoids (2), CLIPPARD Model AVS-12. These "energize-to-actuate" solenoids were used to move the diverting valve's spool. By using two solenoids, on either side of the diverting valve, a hybrid "latching" solenoid was formed. This reduced power consumption.
15. Data Acquisition System (DAS), CYBORG Isaac Model 91A. The Isaac DAS was an interface device between the external

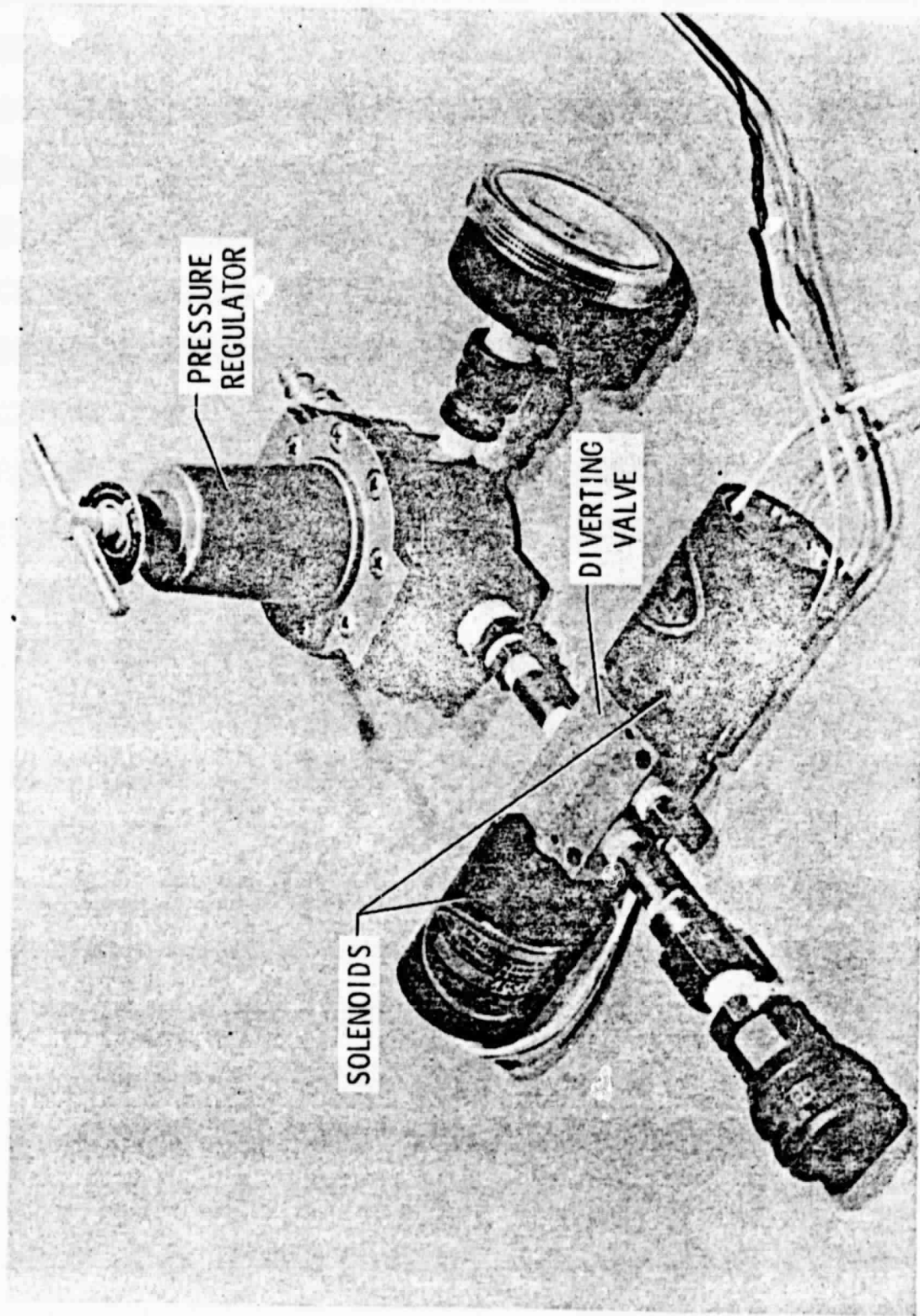


Fig. 3 Photograph of upstream assembly.

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sensors and an APPLE IIe microcomputer. During the final testing of the developed assembly, the performance of the system was continuously monitored by the DAS. The values of upstream pressure and valve stem displacement were fed into the microcomputer through the Isaac's 16 bit binary input port. Any changes in either of these variables were recorded on hard copy.

Table 1 Equipment used in development

Item No.	Equipment, Manufacturer and Model No.
1.	Bellows Sealed Regulating Valve, NUPRO Model SS-4BRG
2.	Digital Linear Actuator, AIRPAX Model L92421-P2
3.	COSMAC Microprocessor Evaluation System, RCA CDP18S025
4.	Barocel Electronic Manometer, DATAMETRICS Model 1174
5.	Linear Variable Differential Transformer, G. L. COLLINS CORP. Model LMT SS-207, S/N5240
6.	Turbometer, COX Model GM 16
7.	Rotameter, BROOKS CO. Model R9M-25-1
8.	Globe Valve, Meets Federal Specification WW-V-51A, Type 1
9.	Incremental Optical Linear Encoder, DYNAMICS RESEARCH CORP. Model SGT-E-4-1-0
10.	Linear Actuator Driver Board, AIRPAX Model K33505
11.	Bridgesensor, CALEX Model 166
12.	Pressure Regulator, PARKER-HANNIFIN Model R2037/J5
13.	Diverting Valve, CLIPPARD Model MJV-4D
14.	Solenoids (2), CLIPPARD Model AVS-12
15.	Data Acquisition System, CYBORG Isaac Model 91A

Chapter 3

DESCRIPTION OF SYSTEM DEVELOPMENT

Treating the three original components listed in Chap. 2 (i.e. the microprocessor system, the regulating valve, and the digital linear actuator) as the basic design, this work was directed toward the modification of the assembly. The modifications and the results of each are described in the following pages.

3.1 Initial Modifications

The turbometer, the globe valve, the rotameter, and an eight switch bit-setter were added to the system at the beginning of the development. The flow meters were added to allow measurements of the volumetric flow rate. The globe valve was added to permit variations of differential pressure, but was unnecessary for the reasons already described under equipment item No. 8, Chap. 2. The eight switch bit-setter was designed and built to allow direct binary input onto the microprocessor system data bus without the need for an ADC. A schematic of the design is shown in Fig. A.4, Appendix A.

3.2 Microprocessor Program Development

After the addition of the equipment listed in sec. 3.1, two microprocessor programs were developed. These programs, the Cycling Program and the Actuator Movement Program, are described in the following pages.

3.2.1 Cycling Program

The cycling program was developed to move the linear actuator/valve assembly in a controlled fashion. The purpose of the program

was to make the actuator/valve assembly cycle from the fully open position to the closed position. The listing of the program is entitled "Cycling Program" in Appendix B. The program operated by generating a clock signal (square wave) that was fed to the Interface Box. The signal was a high voltage ($V_{CLK} = +5$ volts) on bit zero (DO 0) of the output port, which was maintained while a countdown/delay subroutine was executed, and then V_{CLK} was dropped to 0 volts. The clock signal was formed by repeating this process at a set rate. The Interface Box received this square wave and utilized it to generate the signals used to drive the digital linear actuator. When this square wave was accompanied by a high voltage (V_D) on bit one (DO 1) of the output port, the actuator shaft traveled in one direction. If V_D was then brought to zero, the direction of the actuator movement was reversed. By generating continuously the square wave and periodically changing V_D , the valve stem was cycled.

The distance the actuator moved in each direction could be set by placing the desired number of steps of linear travel in a designated holding register. By varying the number of incremental steps, testing showed the maximum linear travel of the valve stem occurred during a test sequence of 80 steps; with a step size of 0.0051 cm/step, this equates to a travel of approximately .408 cm. (An uncertainty may arise because the actuator could skip a step before maximum travel was determined). This value of maximum travel was .17 cm or 71.4% greater than the maximum stem travel documented by the valve manufacturer for the unmodified valve.

3.2.2 Actuator Movement Program

The next step in the development involved developing a second program that moved the actuator/valve assembly to a user specified location. Each location was prescribed by the number of actuator steps the stem was displaced from the "valve closed" position. The eight switch bit-setter was used as the input device to the microprocessor system. The program scanned the switches for the number of displacement steps the actuator/valve assembly was to be moved. The program then generated the clock signal that was input to the actuator driver board. Each square wave corresponded to one step displacement. The driver board output moved the actuator to the user specified position. The AIRPAX Model K33505 board had been modified by removing the on-board 555 timer. This timer controlled the rate at which the actuator was stepped, by means of an externally attached potentiometer. The clock signal required by the board was generated by the microprocessor program in the same manner as described in sec. 3.2.1 and was sent via the output port to the K33505 board. This modification allowed the stepping rate of the actuator to be controlled internally by the microprocessor system. (A listing of this program, entitled "Actuator Movement Program", is given in Appendix B).

3.3 Displacement Sensor

A Linear Variable Differential Transformer (LVDT) was added to the development system at this point to verify the accuracy of the control by the Actuator Movement Program. Accuracy was defined in terms of the program's ability to move the actuator/ valve

assembly to the user specified position on a repeatable basis. Testing of the system showed the actuator shaft wobbled sufficiently to affect the LVDT readings. To eliminate this wobble, two linear bearings were added to the assembly, one at either end of the actuator shaft. Subsequent testing showed that the problem caused by the wobble was eliminated. The result of the mean LVDT calibration test is shown in Fig. 4.

3.4 Flow Rate Calibration

Using the Actuator Movement Program as the stem controller, a series of calibration curves were generated. While maintaining a constant upstream pressure and incrementing the user specified position on the eight switch bit-setter (from the closed position to the fully open setting), data recorded included the volumetric flow rate, the upstream pressure, the differential pressure, and the LVDT voltage at each location. During this testing, it was discovered that the linear actuator could not produce sufficient linear force to close the valve completely at upstream pressures above 267 kPa gage. Analysis revealed that the bellows spring required a linear force of 53.4 N to close the valve after the stem threads had been removed during modification. Since the linear actuator could produce only 84.5 N at the slowest stepping rate, it was clear that at moderate to high upstream pressures, the actuator could not close the valve.

Rather than reduce the range of upstream test pressures and thereby the flow rate resolution, an upstream assembly was added to divert the flow and eliminate the problem of insufficient linear force. A single input/dual output diverting valve was placed between

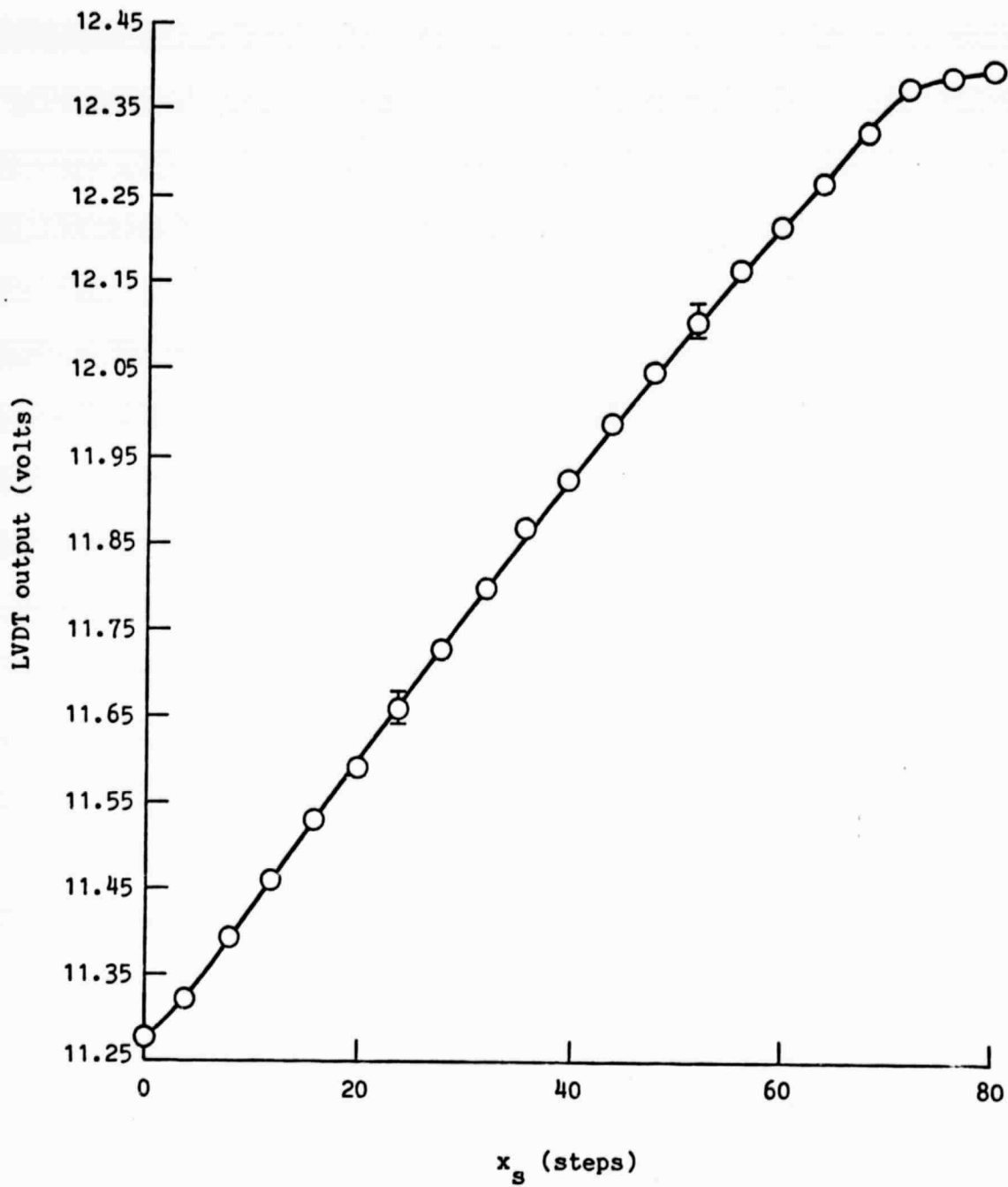


Fig. 4 LVDT calibration curve.

the regulating valve and the upstream pressure regulator, to allow the flow to be stopped during valve stem position changes. This allowed the pressurized volume upstream of the regulating valve to vent to atmosphere; reducing the amount of force required to close the valve. To reduce power usage, the upstream diverting valve was driven by two solenoids -- one on either side of the valve (see Fig. 3). Hence, the upstream flow was cut off by a single pulse of power rather than requiring a continuous current which was required normally to open or close solenoid valves. After the addition of this "latching" solenoid to the upstream assembly, the data presented in Appendix D, Table D.1, were collected without further problems due to the linear actuator. It should be noted that the upstream assembly (i.e., the two solenoids and the diverting valve) was not part of the basic design of the digital valve system, but was added to compensate for the insufficient linear force available from the actuator.

The mean flow rate calibration data are presented in Table 2. Upstream pressure was considered the controlled parameter, and is given in lb_f/in^2 (psi) because the instrumentation operated using those units. While differential pressure is a controlling parameter in the volumetric flow rate of a fluid through a valve (modeling the valve as a venturi [5]), the upstream pressure was used in this development. This decision was based on the lack of resolution in the differential pressure readings across the valve, between successive changes in the valve stem displacement. As shown by the preliminary calibration data shown in Table D.1, Appendix D, the

Table 2 Preliminary volumetric flow rate calibration data

Upstream ^a Pressure		Valve stem displacement (steps from closed position)																				
		00	04	08	12	16	20	24	28	32	36	40	44	48	52	56	60	64	68	72	76	80
Volumetric flow rate ^b																						
207 30	0.0	241 ^c	481 ^d	698	883	1001	1114	1185	1232	1274	1307	1321	1336	1350	1364	1364	1383	1383	1383	1397	1397	1397
	00	70	140	205	265	305	340	365	380	395	405	410	415	420	425	425	430	430	430	435	435	435
	0000	0046	008C	00CD	0109	0131	0154	016D	017C	018B	0195	019A	019F	01A4	01A9	01A9	01AE	01AE	01AE	01B3	01B3	01B3
276 40	0.0	293 ^c	599 ^d	883	1123	1288	1411	1501	1576	1624	1652	1685	1713	1713	1727	1746	1760	1775	1775	1775	1775	1775
	00	85	175	265	345	400	440	470	495	510	520	530	540	540	545	550	555	560	560	560	560	560
	0000	0055	00AF	0109	0159	0190	01B8	01D6	01EF	01FE	0208	0212	021C	021C	0221	0226	022B	0230	0230	0230	0230	0230
345 50	0.0	359 ^c	732	1062	1350	1562	1713	1803	1897	1959	2001	2034	2048	2077	2091	2091	2110	2124	2138	2138	2138	2138
	00	105	215	325	420	490	540	570	600	620	635	645	650	660	665	665	670	675	680	680	680	680
	0000	0069	00D7	0145	01A4	01EA	021C	023A	0258	026C	027B	0285	028A	0294	0299	0299	029E	02A3	02A8	02A8	02A8	02A8
414 60	0.0	414 ^d	897	1260	1576	1822	2001	2138	2228	2289	2336	2364	2412	2440	2440	2440	2473	2487	2501	2501	2501	2501
	00	120	270	390	495	575	635	680	710	730	745	755	770	780	780	780	790	795	800	800	800	800
	0000	0078	010E	0186	01EF	023F	027B	02A8	02C6	02DA	02E9	02F3	0302	030C	030C	030C	0316	031B	0320	0320	0320	0320
483 70	0.0	496 ^d	1034	1425	1789	2062	2275	2426	2534	2596	2657	2700	2733	2761	2794	2808	2822	2841	2841	2841	2841	2855
	00	145	315	445	565	655	725	775	810	830	850	865	875	885	895	900	905	910	910	910	915	915
	0000	0001	013B	01BD	0235	028F	02D5	0307	032A	033F	0352	0361	036B	0375	037F	0384	0389	038E	038E	038E	038E	0393

a. kilo-Pascals gage
psi gage

b. Volumetric Flow Rate (SCCS)
Rotameter 10
Rotameter 16

c. Reading below Rotameter scale, value extrapolated

d. Air spring phenomena. Steady value extrapolated

changes in differential pressure were small. Consequently, because the microprocessor system could not conveniently read fractional changes in the differential readings, upstream pressure was used as the control parameter. This was a valid assumption since the downstream pressure was expected to remain constant in this application. The validity of this was further verified by the data shown in Fig. 5, where a linear relationship between differential pressure and upstream pressure was observed. Therefore, although the use of upstream pressure was not fundamentally correct, it was valid to use as a control parameter under the system conditions.

In Table 2, the SI equivalents to the read-out pressure are listed along the left column of the table along with the psi value. Every fourth user specifiable location (from closed to fully open) was tested in the flow rate calibration. Each of the volumetric flow rate data in Table 2 are listed as three numbers: (1) the volumetric flow rate in standard cm^3/sec (SCCS); (2) the flow rate reading from the rotameter (used in the table to eliminate the need to convert from rotameter scale to SCCS); and (3) the volumetric flow rate reading from the rotameter scale in hexadecimal. A graphical representation of this table is given in Fig. 6. The correction factor between actual flow rate and flow rate at standard temperature and pressure was less than 1%. For this reason, the flow rates are listed at standard conditions. The LVDT calibration data in Fig. 4 shows that the maximum linear travel of the valve stem was not 0.408 cm as expected for a linear travel of 0.0051 cm/step. Rather, the linear distance traveled began to approach asymptotically a maximum

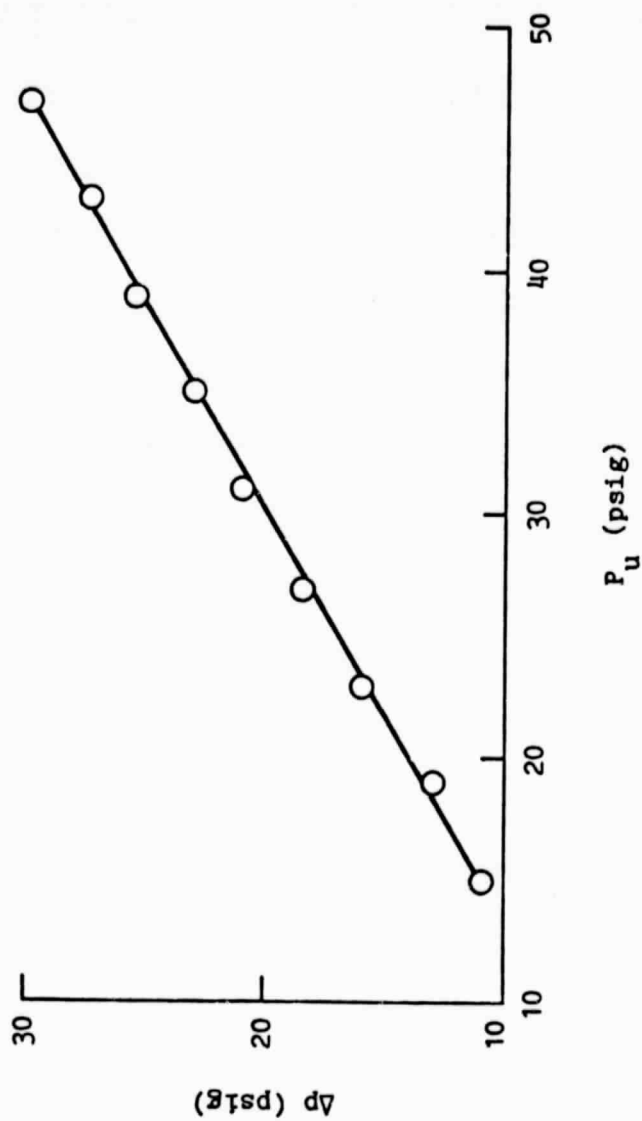


Fig. 5 Differential pressure across the valve vs upstream pressure.

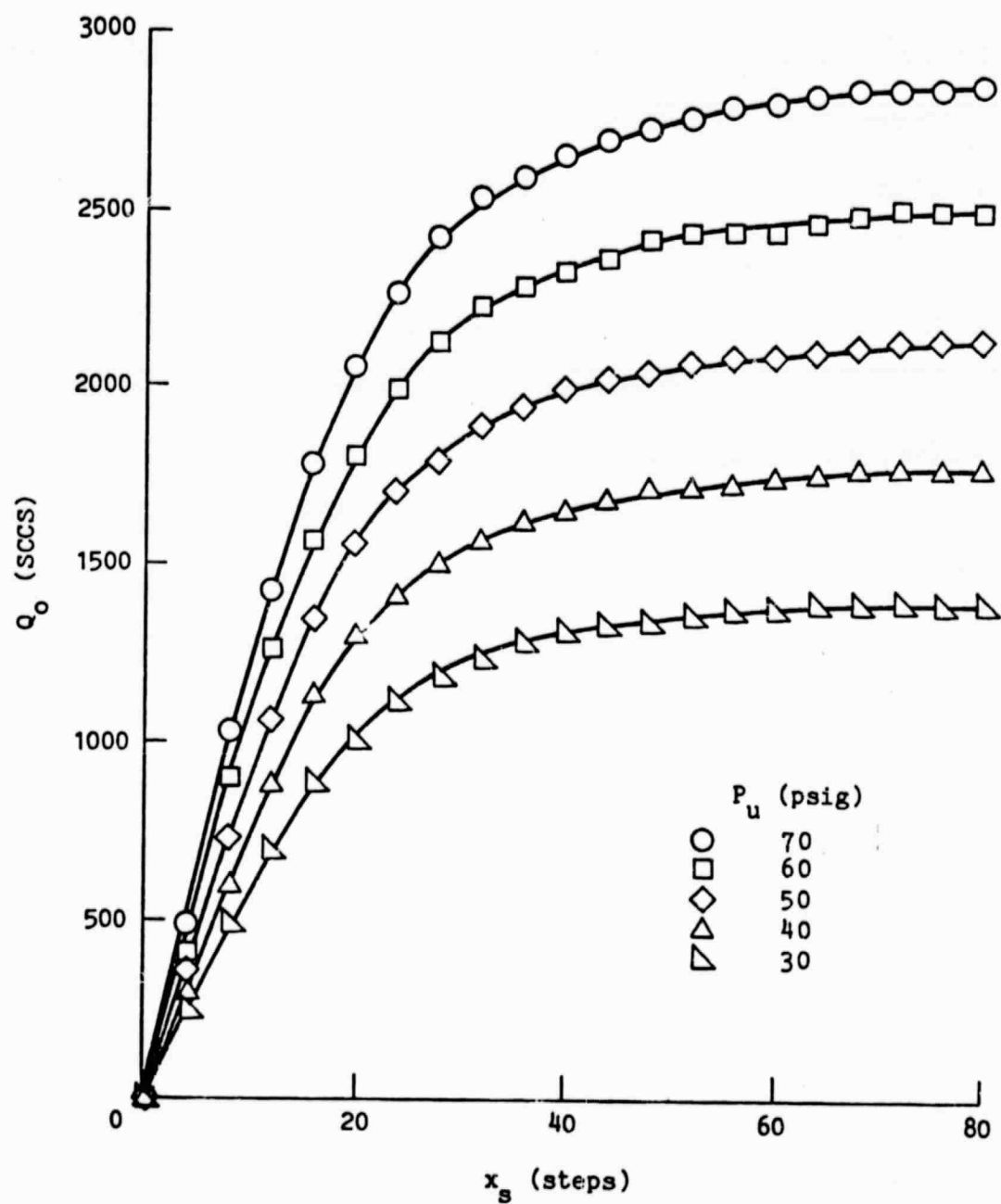


Fig. 6 Preliminary volumetric flow rate calibration curves.

value of 0.3429 cm. The nonlinearity resulted from the condition of the bellows spring, which changed from compression to tension. This tension caused the actuator to slip when the bellows was fully stretched.

3.5 Interpolation Program

The generation of Table 2 was essential in the development of the Interpolation Program for the microprocessor system. The graphical representation, Fig. 6, was used to determine how the interpolation should proceed. The interpolation process was chosen over a table look-up due to the limitation of the RAM size in the microprocessor system. It was not desirable (or even possible) to have a data table stored in the microprocessor system memory with the volumetric flow rate data for each user specifiable location. (Such a table would require 6683 memory bytes rather than the 215 memory bytes required by the data in Table 2). Therefore, because the calibration curves shown in Fig. 6 are non-linear, it became necessary for the microprocessor system to calculate the correct valve stem displacement for a requested flow rate. The calculated displacement was in the number of actuator steps from the closed position, to achieve the specified volumetric flow rate at the monitored upstream pressure. To do this, the microprocessor system was required to interpolate the correct displacement using data similar to that of Table 2. Difficulty arose at this point because the RCA CDP18S025 Evaluation System was incapable of interpolation without the addition of a RCA CDPR582CD Fixed-Point Binary Arithmetic Circuit. Without the addition of this circuit, the microprocessor

system was only capable of multiplication or division by two. To eliminate the need for this costly and time consuming modification, a special interpolation program was written. A listing of this program can be found under "Interpolation Program" in Appendix B.

The Interpolation Program consisted of two parts: (1) the collection and storage of the necessary data for the interpolation process; and (2) the interpolation of the correct valve stem displacement (in terms of the number of linear actuator steps from the closed position). A flow chart of the data collection element is shown in Fig. 7. The interpolation element of the program is represented by the flow chart in Fig. 8.

The data from Table 2 was stored in the microprocessor system RAM area in such a way as to utilize the memory address locations in the Interpolation Program. First, the data was formed into columns of constant upstream pressure with the pressure reading at the head of each column. Next, these columns were placed into the RAM in a single continuous array, starting with the lowest upstream pressure column, then followed by columns for each successively higher pressure. Because these constant pressure columns were each 43 bytes in length, flow rate data for the same displacement could be found by moving the look-up pointer by 43 counts. For example, at 40 psig the flow rate was 1576 SCCS at 32 steps displacement. By moving the look-up pointer 43 counts ahead, a flow rate of 1897 SCCS was found for the same displacement but at 50 psig.

To illustrate the interpolation process, Fig. 9 has been supplied to show the sequence of the program. The numbers in

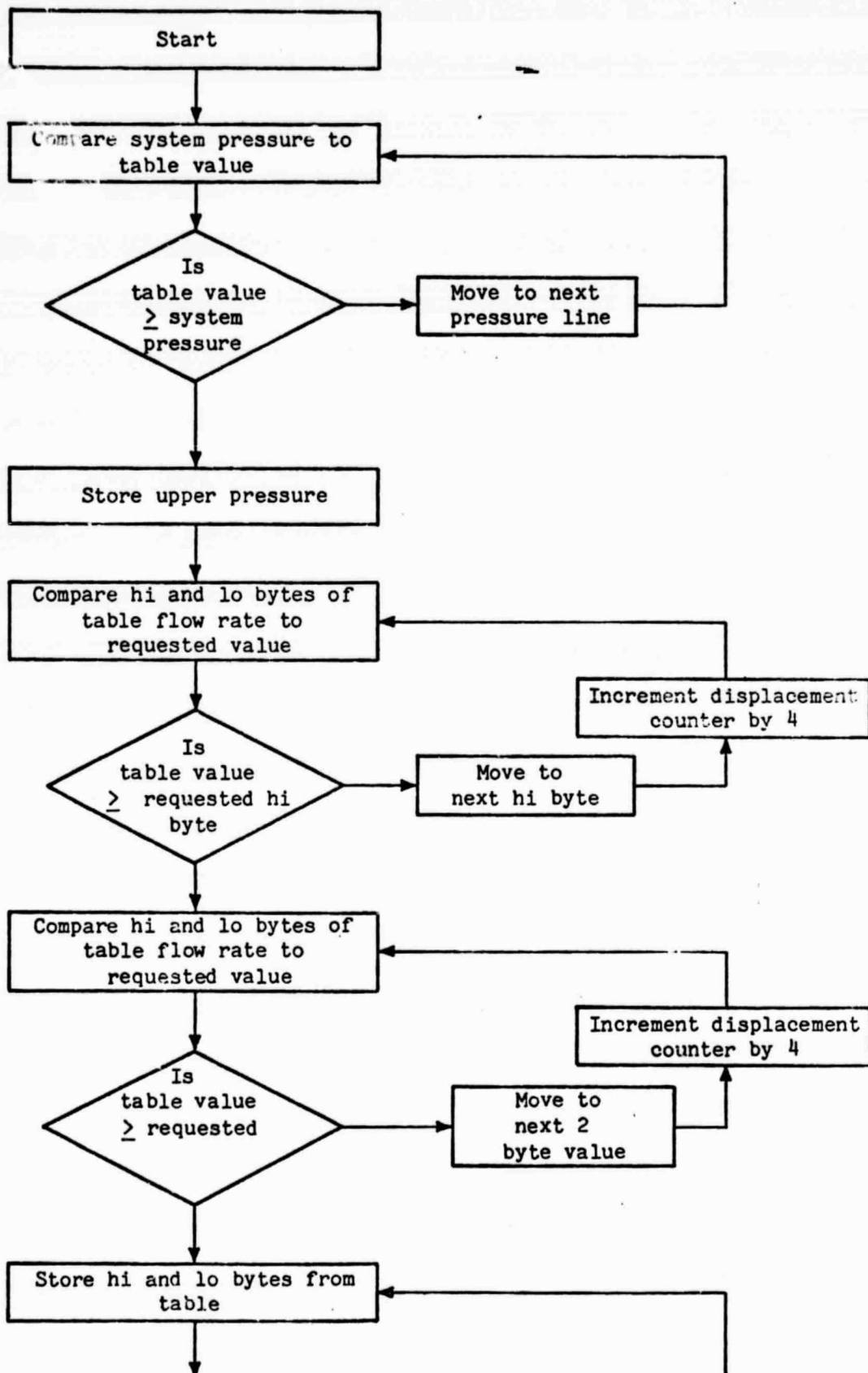


Fig. 7 Flow chart of data collection element of the Interpolation Program.

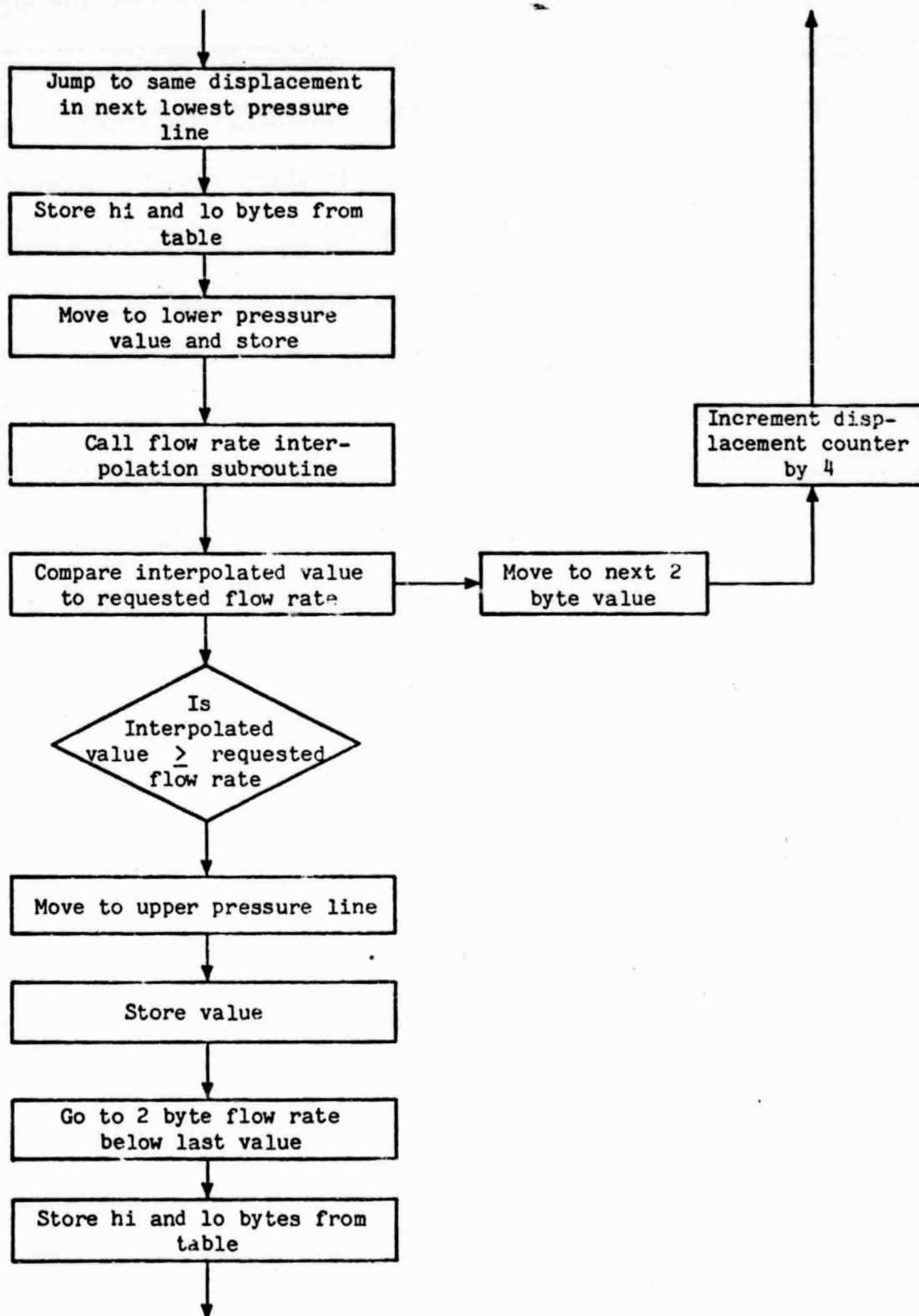


Fig. 7 Continued.

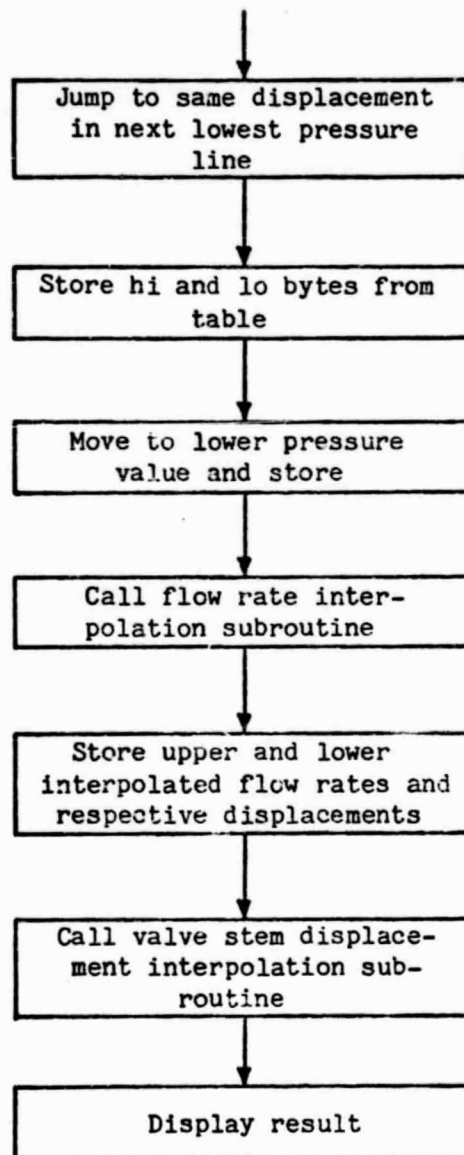


Fig. 7 Concluded.

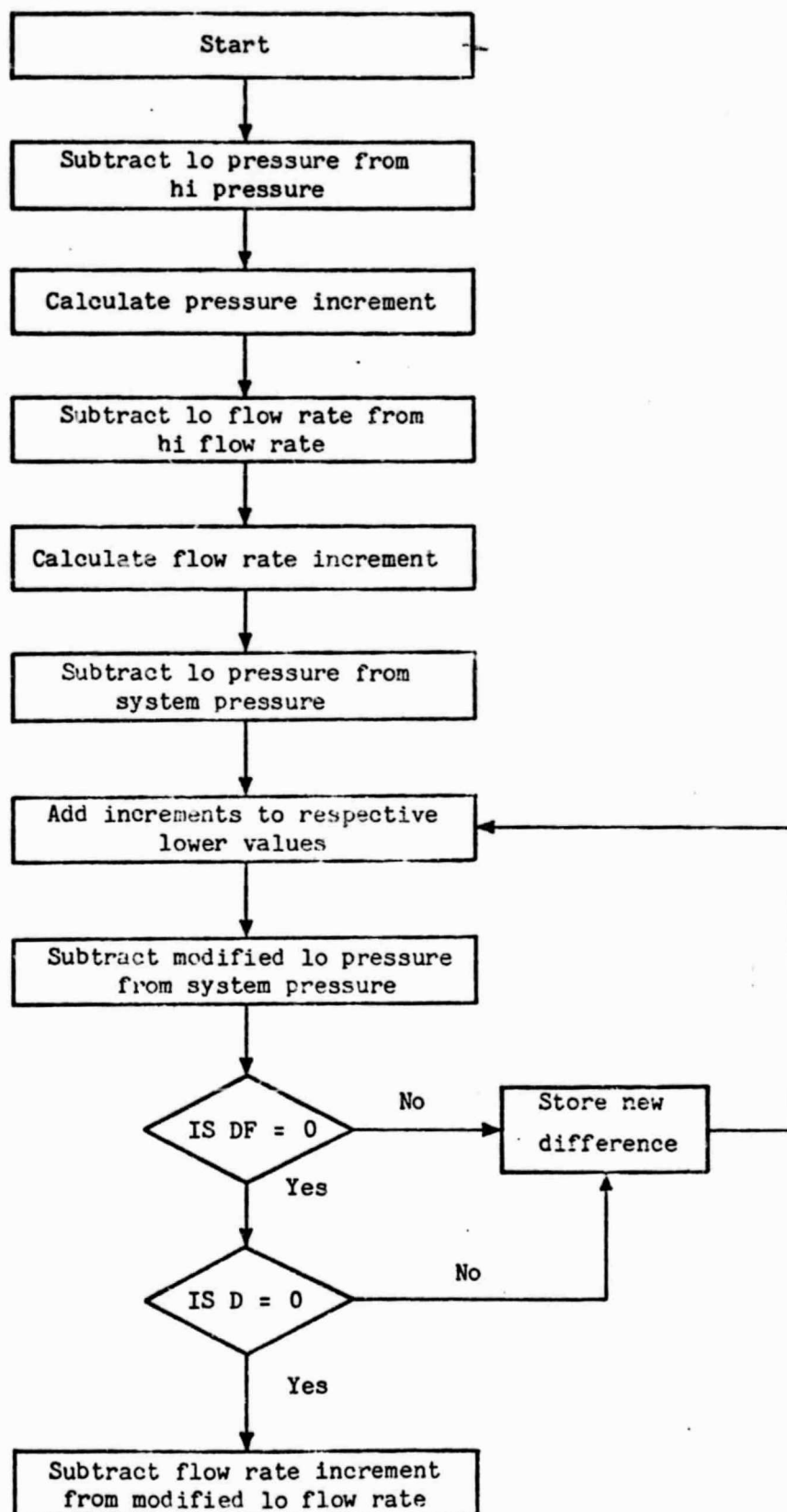


Fig. 8 Flow chart of interpolation element of the Interpolation Program.

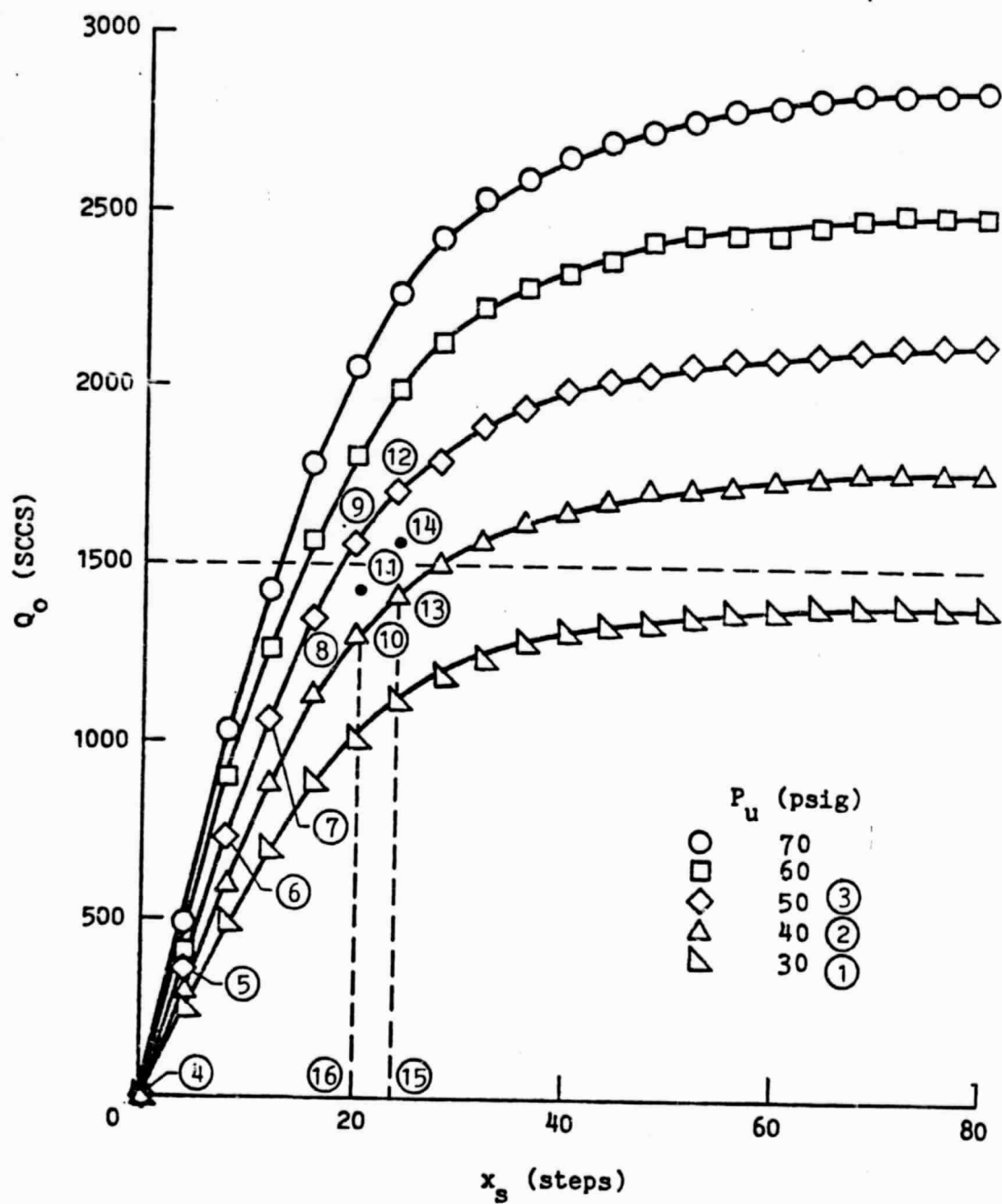


Fig. 9 Illustration of interpolation process.

parentheses () in the following text will correspond to those in Fig. 9. The conditions of the illustration are: $P_u = 45$ psig; $Q_r = 1500$ SCCS.

3.5.1 Data Collection Element

The data collection element of the Interpolation Program began by comparing the input value of upstream pressure to the first table value (in this case, 30 psig (1)). If the system pressure was not less than or equal to the memory table value, the program jumped the subsequent flow rate data to the next highest constant upstream pressure line and repeated this comparison. This process continued until the program reached a value of memory table pressure greater than or equal to the input pressure (2,3). At this point, the upper pressure range had been found for the interpolation process. The lower range would be the last value of memory table pressure not accepted (2). The program stored the upper pressure limit in a specified holding area and then compared the requested volumetric flow rate to the values in the array at that point. The two byte volumetric flow rate requested would be compared to each two byte data value in that upper pressure line until one was found that was greater than or equal to the request value (5-9). Simultaneously, a storage register was increased by increments of four as each successive table flow rate was compared. As a result, the corresponding valve stem displacement (in steps from the closed position) could be determined.

Once a table flow rate was found that was greater than or equal to the requested value, that value was stored in the specified

holding area and the program jumped back in the array to the same valve stem displacement location in the lower pressure range (10). This value of flow rate and then the lower pressure value were also stored in the specified holding area. With these four data values, the interpolation element (to be described later) was used to calculate the volumetric flow rate corresponding to the system pressure, at that valve stem displacement (11). The interpolated flow rate was compared to the requested value to determine if it was greater than or equal to the requested flow rate. If the calculated value was not greater than or equal to the requested value, then the upper boundary of flow rate along the system pressure line had not been determined. The data collection element of the program was repeated to get to the next highest flow rate pair (12,13). These are along the upper (3) and lower (2) table pressure boundaries (having been determined earlier). The process was repeated until the calculated flow rate was greater than or equal to the requested flow rate (14). At this point, the upper (14) and lower (11) boundaries of the flow rate had been determined (along the upstream pressure line of the system). The correct valve stem displacement can then be interpolated between these two values.

3.5.2 Interpolation Element

The initial step for interpolating the volumetric flow rate (14) consisted of subtracting the lower boundary pressure and flow rate values from the upper boundary values (2,13 from 3,12). The differences were then divided into increments. The increment sizes corresponded to resolution of the microprocessor

system. In the case of calculating the volumetric flow rate (along the system pressure line), the divisor is eight. This is for two reasons: (1) Since the microprocessor system can only multiply or divide by twos, the possible divisors were 2, 4, 8, or 16. A divisor of four would cause the increment size to be too large. A divisor of 16 would result in the pressure increment being zero. (2) The microprocessor system had a resolution of one psi for the monitoring of upstream pressure. It was desirable to have the increment size the same as the resolution. This reduced the round-off error.

The interpolation was simulated by adding the pressure and flow rate increments to the respective lower boundaries until the incremented pressure equaled the system pressure. At this point, the incremented lower flow rate (13) was equal to the interpolated value along the system pressure line (14). To achieve this in the programming, a test condition was set-up for the microprocessor system. As a test condition, the lower pressure boundary was subtracted from the system pressure and the result was stored. After the pressure increment was added to the lower pressure boundary, this sum was subtracted from the system pressure. This new difference (DIFF) was compared to the test criteria (TEST); if DIFF was less than TEST, the incremented lower pressure value was approaching the system pressure and another increment of pressure and flow rate were added to the respective holding areas. Subsequently, the new DIFF became the test criteria (TEST). This process continued until the new DIFF was greater than or equal to TEST. At that point, the incremented lower number pressure was diverging from the system

pressure so that the old value of incremented volumetric flow rate was the closest approximation possible (with the increment size used).

The process was similar for the valve stem displacement interpolation. The major difference was the requested volumetric flow rate became the measure by which the divergence test was made. The values stored in the specified holding area were the upper and lower interpolated flow rates (14,11) and their respective valve stem displacements (15,16). The incremental values were found by dividing the differences between the upper and lower boundaries by four (the number of steps between adjacent flow rate values on the data table). The incremental value of flow rate was added to the lower interpolated flow rate. The test condition for the microprocessor system was the difference between the lower flow rate (11) and the requested value. After the addition, the modified lower flow rate was subtracted from the requested flow rate. This new difference (DIFF) was compared to TEST; if DIFF was less than TEST, the incremented lower flow rate was approaching the requested value and another increment of flow rate and displacement were added to the respective incremented lower boundaries. As before, the new DIFF became TEST. As in the flow rate interpolation, this process continued until the divergence was detected. At that point, the previous value of displacement would give a flow rate closest the requested value (assuming the system upstream pressure was maintained). Testing of the completed program showed the accuracy of the interpolation process to be within one step in 80 or within 1.25% full scale.

3.6 Development of Overall Controlling Program

With the Interpolation Program completed, work began on a 32 switch bit-setter and the addition of four input ports to the microprocessor system. This new bit-setter was needed for the development of an overall controlling microprocessor program. This program included the Interpolation Program and parts of the Actuator Movement Program. The 32 switch bit-setter was needed to simulate the inputs, which consisted of the upstream pressure, requested volumetric flow rate, and LVDT voltage. The 32 switch bit-setter is equivalent to four eight switch bit-setters combined (see schematic, Fig. A.5, Appendix A). The 32 switch bit-setter and the four input ports were tested and found to function properly.

The execution of the Overall Program is a closed loop (see Fig. 10 for a flow chart of the Overall Program). The microprocessor system read from 16 of the input switches the requested volumetric flow rate, and from another eight the value of upstream pressure from the manometer. Those readings were then used in the Interpolation Program to calculate the correct valve stem displacement. The system then moved the actuator/valve assembly to the calculated position. A subroutine checked the LVDT voltage against a calibration table to determine whether the actuator/valve assembly was correctly positioned. In the event the LVDT voltage showed the actuator/valve assembly to be incorrectly positioned, the Overall Program would calculate the error in the number of actuator steps and correct the position by repositioning the assembly. On test cycles of the Overall Program when the actuator/valve assembly were already

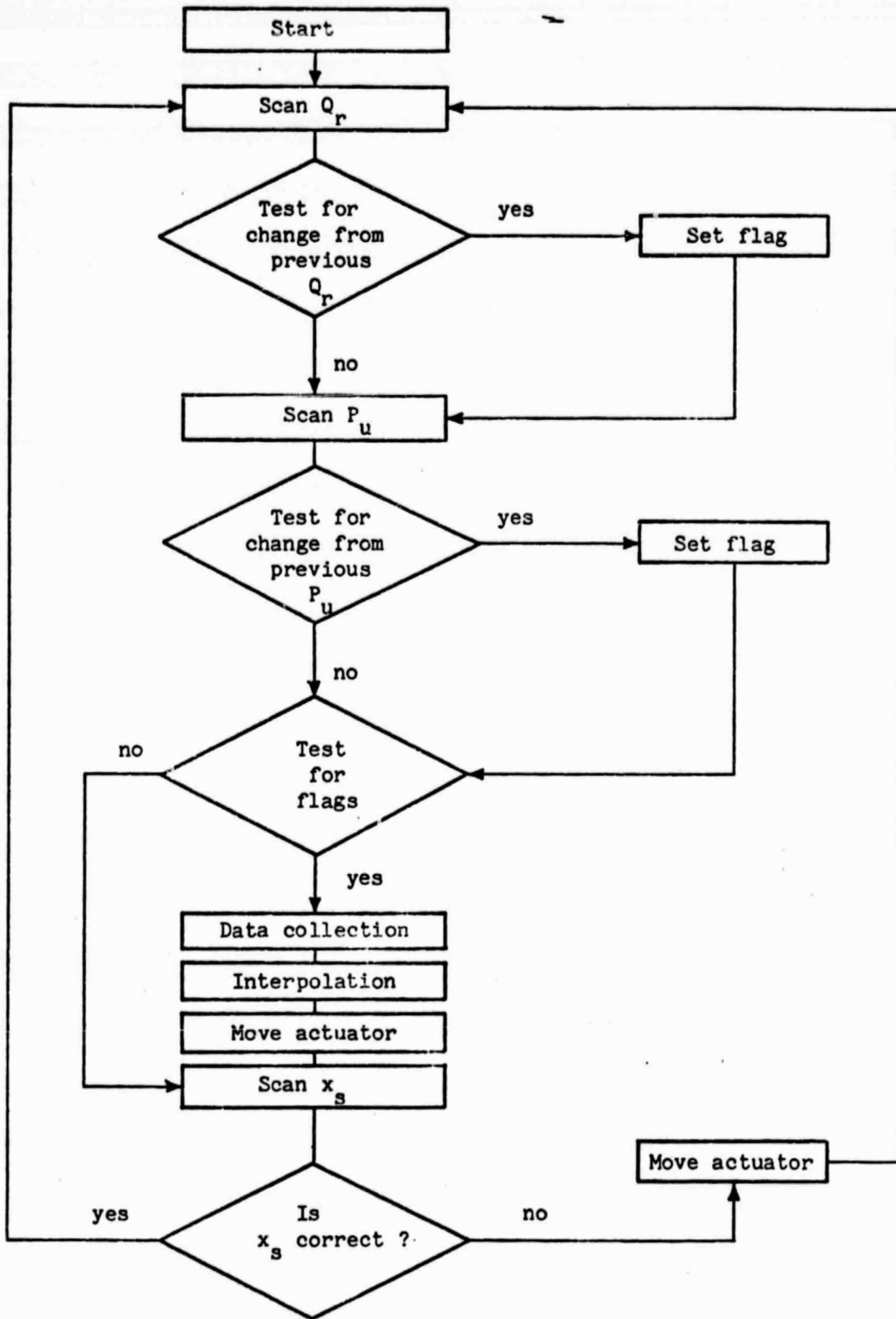


Fig. 10 Flow chart of Overall Program

positioned correctly, the LVDT subrouting was still executed to determine whether the assembly had slipped out of position. A simulation of the program was run using the 32 switch bit-setter to input the values of requested flow rate, upstream pressure, and LVDT voltage. From the results of the testing, the Overall Program was found to function correctly.

3.7 Attachment of Peripheral Sensors

With the Overall Program functional, efforts were focused on improving the peripheral sensing devices (i.e., the manometer and the position sensor) and to attach them to the microprocessor system. The electronic manometer was attached by means of the Manometer Interface Circuit (MIC) shown in Appendix A. This circuit counted the number of pulses received from the CALEX Bridgesensor V/F converter over a one second period and put the total on the data bus. The total was updated once every four seconds.

Because the analog signal from the LVDT was incompatible with the digital microprocessor system, an Incremental Optical Linear Encoder (IOLE) was used to replace the LVDT in the development and thereby eliminate the need for an ADC (see Fig. 11 for main valve assembly). Normally the IOLE has a resolution of 1.016×10^{-3} cm/cycle and a square wave quadrature output (90° out of phase) that is TTL compatible. But, a resolution of 1.016×10^{-3} cm/cycle was too sensitive for the maximum stem travel of 0.3429 cm shown by the LVDT. Such a sensitivity produced a maximum pulse count which exceeded the limiting value of 255 for one byte. Since the input was limited to one byte due to the number of available input ports, and

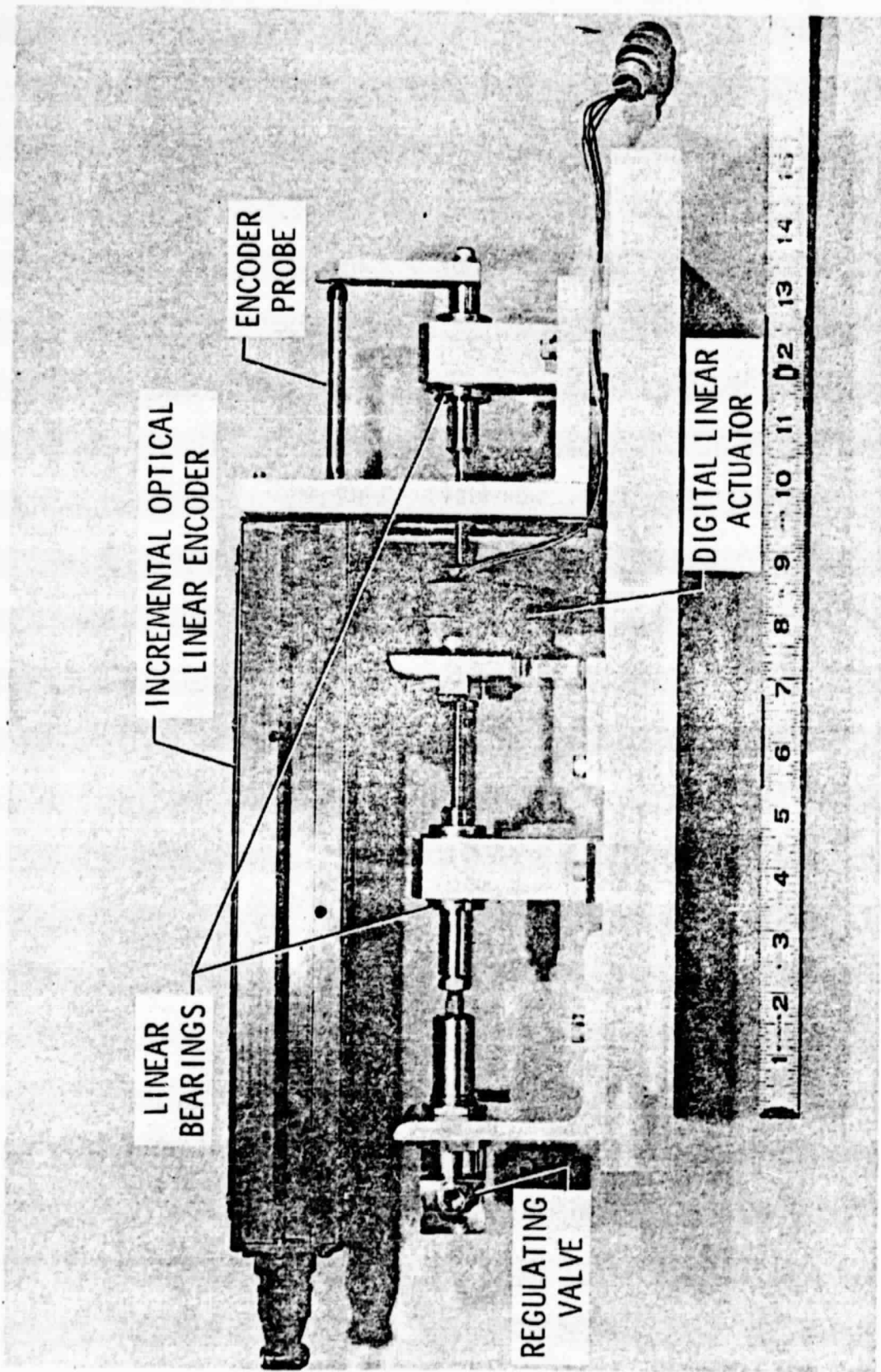


Fig. 11 Photograph of main valve assembly.

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the way the LVDT subroutine was written (the subroutine was written to manipulate eight bits not 16), an IOLE Interface Circuit (shown in Fig. A.1, Appendix A), was utilized to reduce the sensitivity of the IOLE by a factor of two. This kept the count-up total to one byte. Testing of the circuit with the IOLE, before the sensor was attached to the assembly, was successful.

With the IOLE Interface Circuit working, the IOLE was sent out to be attached to the valve assembly. Testing of the assembly after return showed the IOLE to be inoperative. Examination of the device revealed it had been damaged during the attachment and that realignment of the internal optical components was required. The manufacturer was contacted in an attempt to get the correct repair procedure. Repairs were completed and testing began to determine the ability of the IOLE to detect the valve stem displacement accurately and repeatably. The IOLE was found to be incapable of accurately measuring the displacement of the stem with the mounting configuration and the stem movement scheme as programmed. Analysis of the assembly revealed the IOLE to be adversely affected by the vibrations of the digital linear actuator during the movement phase. The errors induced by the vibrations were also cumulative, requiring the IOLE Interface Circuit to be reset each time the valve was closed. Rubber gaskets (vibration dampers) were added to the assembly in the locations shown in Fig. 12 to reduce the effect of the vibrations. Also, the Stepping Rate (SR) of linear actuator was reduced from 40.1 steps/sec to 5.41 steps/sec. However, testing showed that even with these modifications, there was still a small error caused by

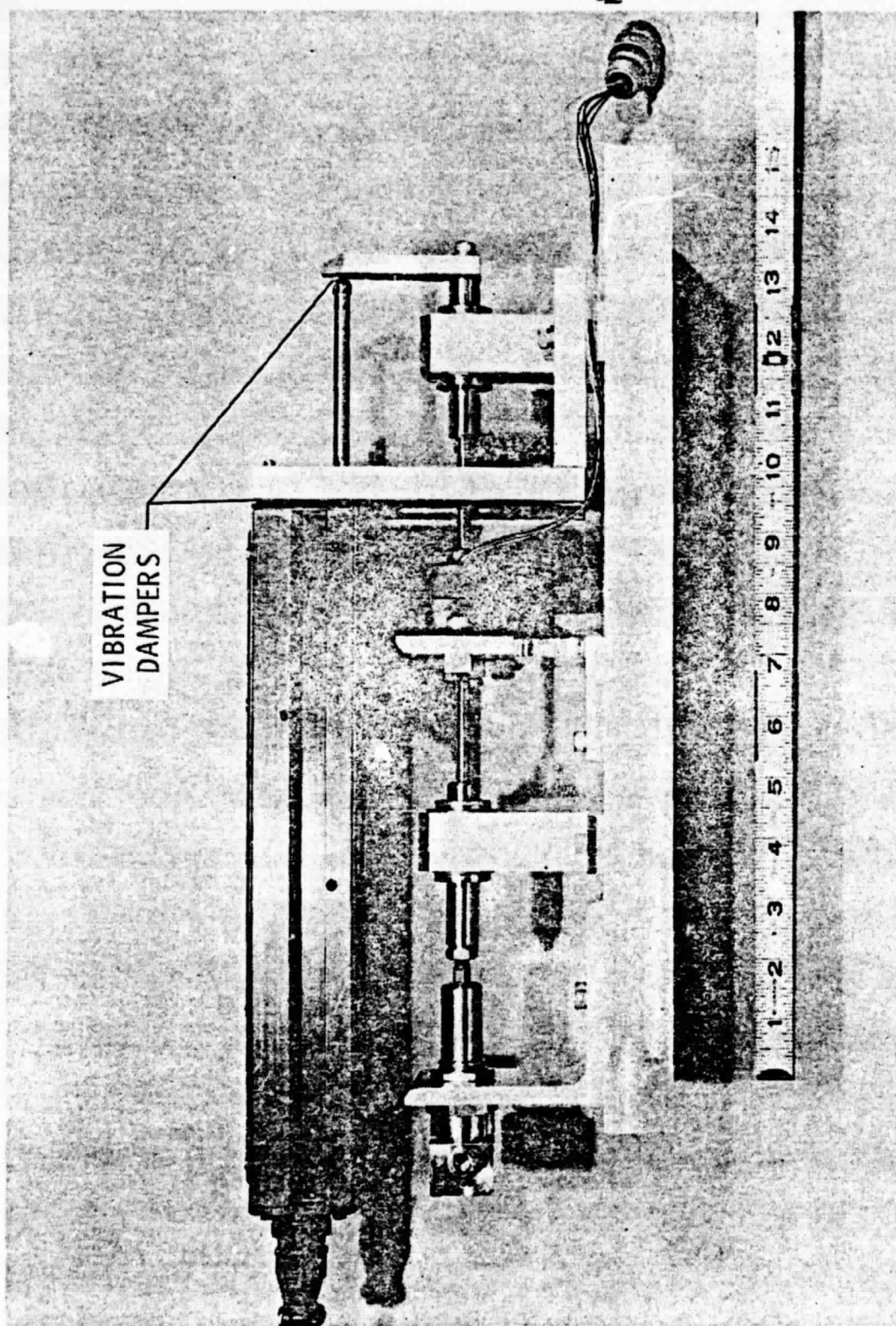


Fig. 12 Photograph of locations of vibration dampers.

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vibrations and this error was cumulative in the IOLE Interface Circuit. To eliminate this problem, the Overall Program was modified to close the valve before each repositioning and to reset the Interface Circuit each time the valve was closed. With these modifications in effect, the data table listed in Appendix D was recorded (see Linear encoder calibration data in Appendix D, Table D.2).

With these Overall Program modifications it became necessary to again simulate the assembly. This time simulation was performed using the 1802 simulator on the computer system at NASA Langley Research Center. The simulation revealed minor errors in the program that were corrected and testing began on the complete assembly.

3.8 Testing of Overall Assembly

To allow visual read-out of the binary data supplied by the interface circuits, two arrays of Light Emitting Diodes (LED's) were attached. The outputs from the interface circuits to the LED's were buffered, to prevent loading by the signals being sent to the input ports of the 32 switch bit-setter.

At this point, the overall assembly was tested. A requested volumetric flow rate of 1543 SCCS was entered in the Overall Program via the 32 switch bit-setter. While the system responded, the response was erratic and the test was stopped. Analysis of the situation showed real-time errors in the Overall Program. It was found that the program commands were not synchronized with the timing loop of the MIC. The values of upstream pressure being accepted by the microprocessor system were incorrect after two to three cycles of

the program. The problem was eliminated by altering the circuit so that it didn't begin the four second count cycle until a signal was received from the microprocessor system. After the upstream pressure was read, the microprocessor sent a command to turn the circuit off again. By placing a delay between the time the circuit was turned on and the time the upstream pressure was read by the microprocessor system, the two were synchronized.

Testing of the assembly was repeated. The resulting volumetric flow rate was within $\pm 10\%$ of the requested value, but the assembly would not maintain a steady setting for more than 30 seconds before the valve stem was repositioned. This constant cycling was due to the MIC. The output from the circuit fluctuated within plus and minus one psi of the actual upstream pressure. The reason for this fluctuation was not found. Therefore, to compensate, the Overall Program was altered to cause the microprocessor system to ignore a pressure fluctuation of one psi. When testing began again, the assembly achieved a steady position and compensated for all upstream pressure fluctuations, but the output volumetric flow rate was not the amount requested. The problem arose from the method by which the original flow rate calibration data were taken. The constant upstream pressure value had been taken from the pressure gage attached to the upstream assembly. The pressure drop across the upstream diverting valve plus head loss in the tubing meant the flow rate calibration data in Table 2 was no longer consistent. Therefore, a new set of calibration data were taken. A schematic of the overall system is shown in Fig. 13. The new location of the

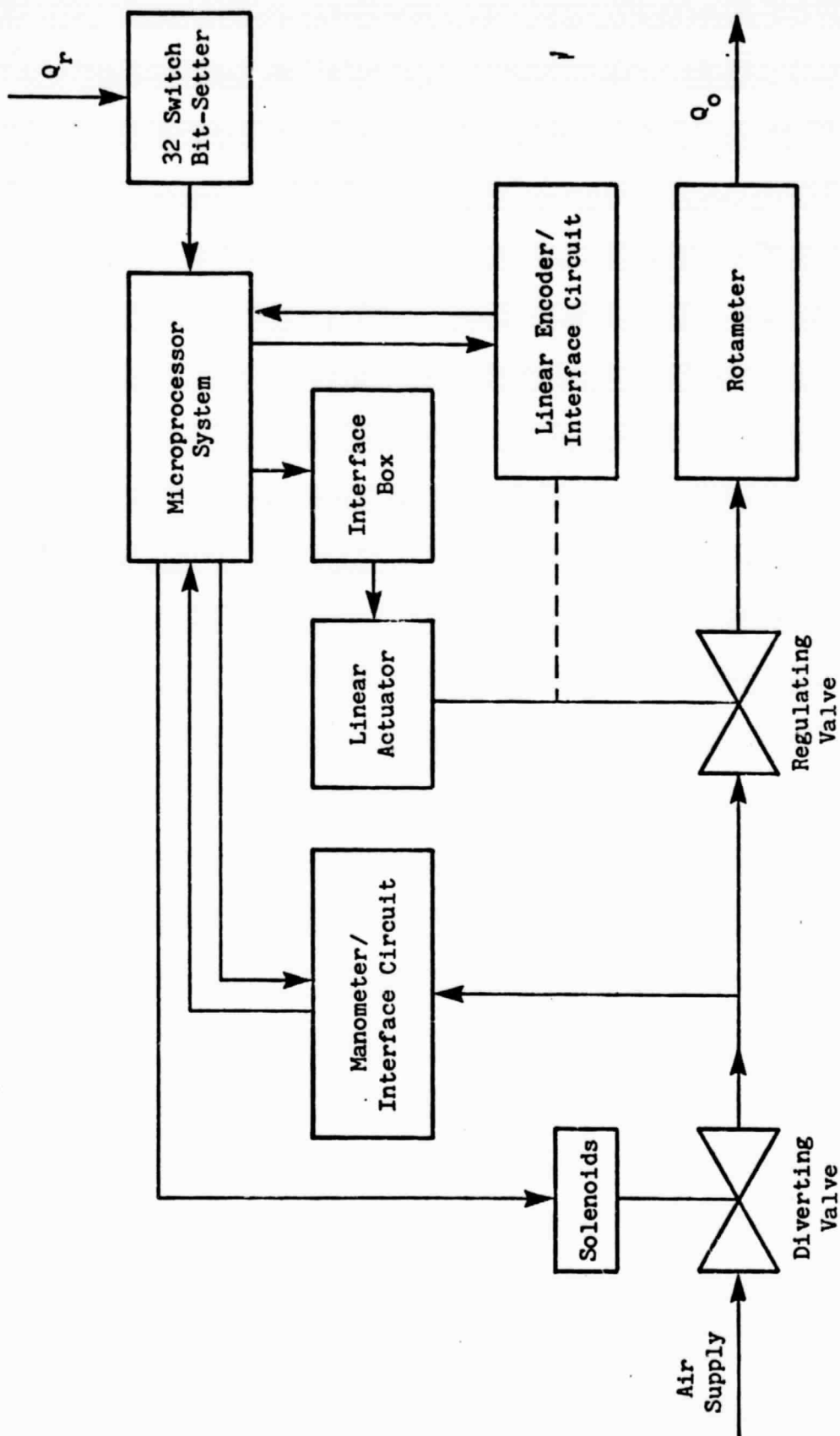


Fig. 13 Schematic of overall assembly.

pressure sensor was approximately 15 cm upstream of the regulating valve. Table 3 shows the averaged flow rate data in the same format used in Table 2. The data is graphically represented in Fig. 14. Note that the shape of the curves in Fig. 14 are the same as those in Fig. 6. Because the interpolation program was written with this curve shape in mind, the interpolation process developed in the Overall Program was still valid.

When the testing of 1543 SCCS was repeated, the resulting volumetric flow rate was within $\pm 5\%$ of the requested value. The upstream pressure was changed to reveal whether the overall assembly could compensate. All output flow rates were within $\pm 5\%$ of the 1543 SCCS requested. Further tests of the system were run with requested flow rates of 708, 858, and 1251 SCCS. The results of all these tests can be seen in Figs. 15(a) - 18(b). The data points are numbered to show the sequence of the testing. The primes (') denote the sequence of the testing with the upstream pressure recorded from the Manometer Interface Circuit. The other numbers denote the sequence of testing with the upstream pressure recorded from the digital read-out of the manometer. It should be noted that the microprocessor system accepted input from the interface circuit; therefore, the upstream pressures used in the interpolation process were frequently incorrect.

Figures 15(a) - 18(a) are the delivered flow rate versus the upstream pressure. Figures 15(b) - 18(b) are the IOLE output versus the upstream pressure. As can be seen in Fig. 15(a), the error was more than 5% at two pressure readings during the 708 SCCS testing

Table 3 Volumetric flow rate calibration data

Valve stem displacement (steps from closed position)																							
Upstream ^a Pressure		00	04	08	12	16	20	24	28	32	36	40	44	48	52	56	60	64	68	72	76	80	
		Volumetric flow rate ^b																					
103 15	0.0	241 ^c	463 ^d	632	779	901	1024	1114	1208	1270	1331	1373	1435	1482	1515	1543	1543	1543	1543	1543	1543	1543	1543
	00	70	135	185	230	270	310	340	370	390	410	425	445	460	470	480	480	480	480	480	480	480	
	0000	0046	0087	0089	00E6	010E	0136	0154	0172	0186	019A	01A9	01BD	01CC	01D6	01E0	01E0	01E0	01E0	01E0	01E0	01E0	
159 23	0.0	359 ^c	618 ^d	840	1052	1237	1373	1515	1638	1727	1822	1883	1973	2015	2058	2105	2133	2147	2147	2147	2147	2147	
	00	105	180	250	320	380	425	470	510	540	570	590	620	635	650	665	675	680	680	680	680	680	
	0000	0069	0084	00FA	0140	017C	01A9	01D6	01FE	021C	023A	024E	026C	027B	028A	0299	02A3	02A8	02A8	02A8	02A8	02A8	
214 31	0.0	481 ^c	779	1067	1331	1543	1741	1911	2044	2162	2265	2341	2426	2501	2563	2610	2657	2685	2685	2685	2685	2685	
	00	140	230	325	410	480	545	600	645	685	720	745	775	800	820	835	850	860	860	860	860	860	
	0000	008C	00E6	0145	019A	01E0	0221	0258	0285	02AD	02D0	02E9	0307	0320	0334	0343	0352	035C	035C	035C	035C	035C	
269 39	0.0	599 ^d	944	1284	1605	1850	2091	2280	2445	2563	2685	2794	2902	2978	3054	3115	3162	3195	3195	3195	3195	3195	
	00	175	285	395	500	580	660	725	780	820	860	895	930	955	980	1000	1015	1025	1025	1025	1025	1025	
	0000	00AF	011D	018B	01F4	0244	0294	02D5	030C	0334	035C	037F	03A2	03BB	03D4	03E8	03F7	0401	0401	0401	0401	0401	
324 47	0.0	732	1100	1496	1836	2147	2398	2638	2822	2978	3115	3242	3370	3464	3563	3625	3691	3705	3705	3705	3705	3705	
	00	215	335	465	575	680	765	845	905	955	1000	1040	1080	1110	1140	1160	1180	1185	1185	1185	1185	1185	
	0000	00D7	014F	C1D1	023F	02A8	02FD	034D	0389	03BB	03E8	0410	0438	0456	0474	C488	049C	04A1	04A1	04A1	04A1	04A1	

a. kilo-Pascals gage
psi gage

b. Volumetric Flow Rate (SCCS)
Rotameter 10
Rotameter 16

c. Reading below Rotameter scale, value extrapolated
d. Air spring phenomena. Steady value extrapolated

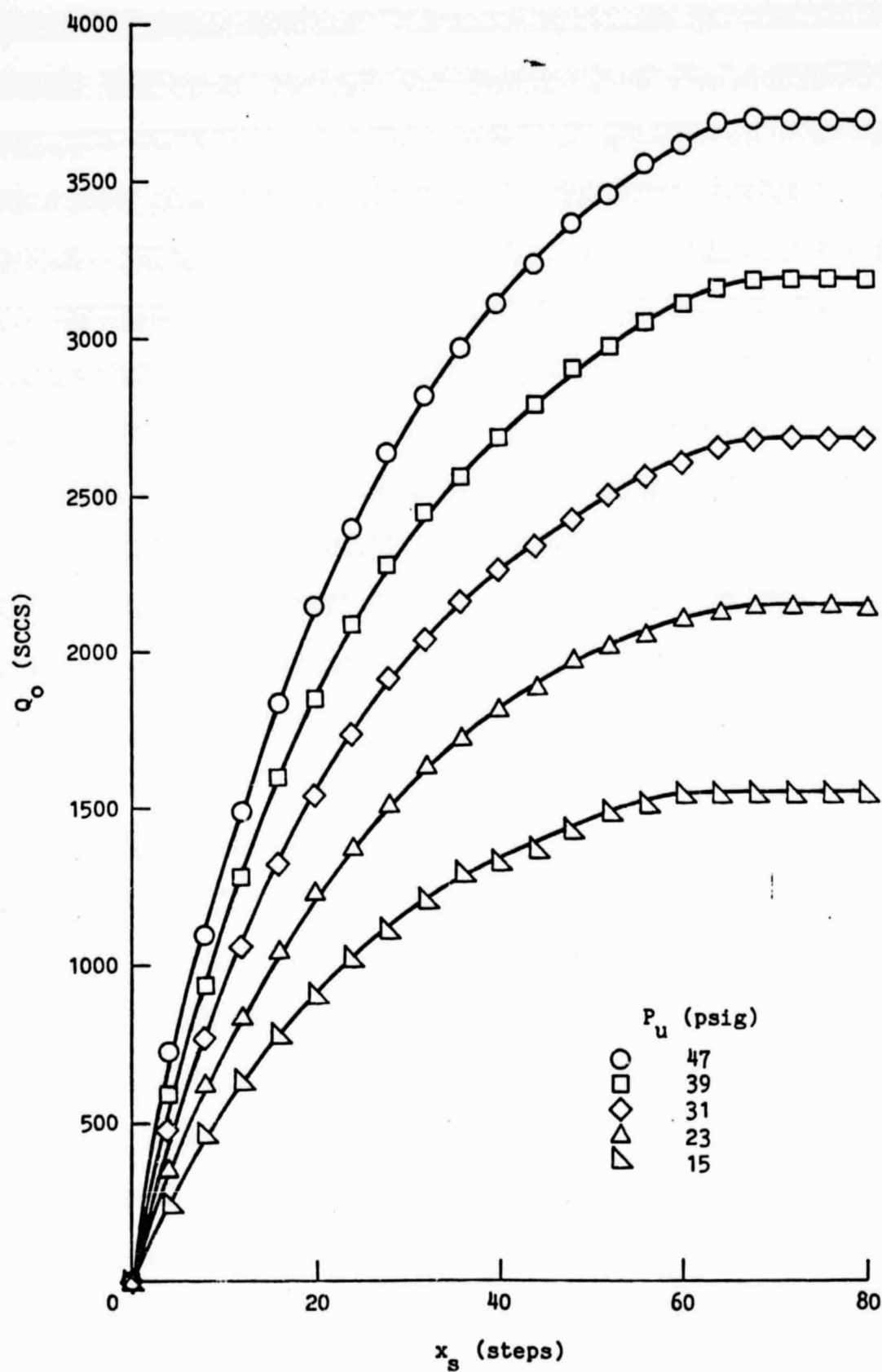


Fig. 14 Volumetric flow rate calibration curves.

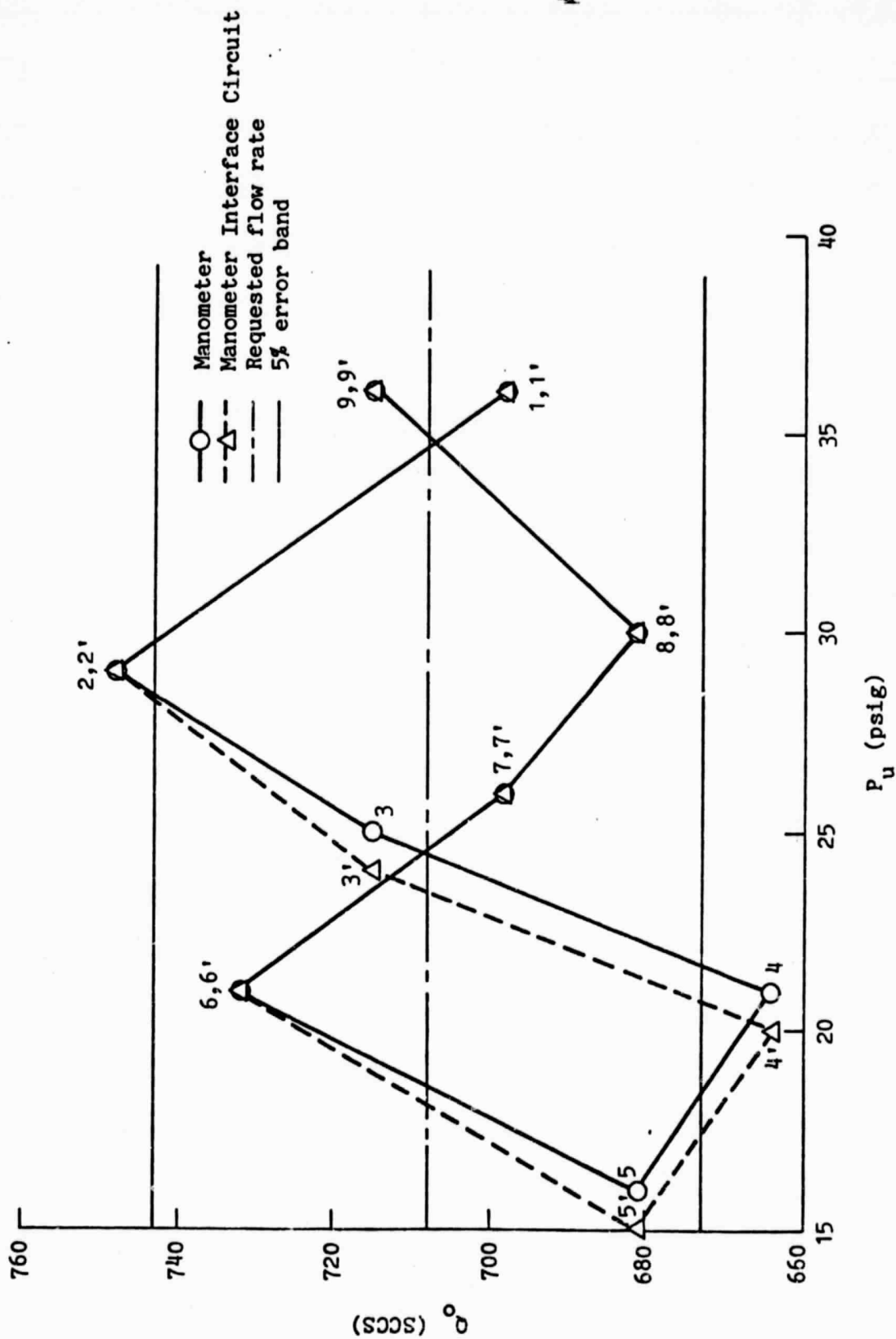


Fig. 15(a) Output volumetric flow rate vs. upstream pressure; $Q_r = 708$ SCCS.

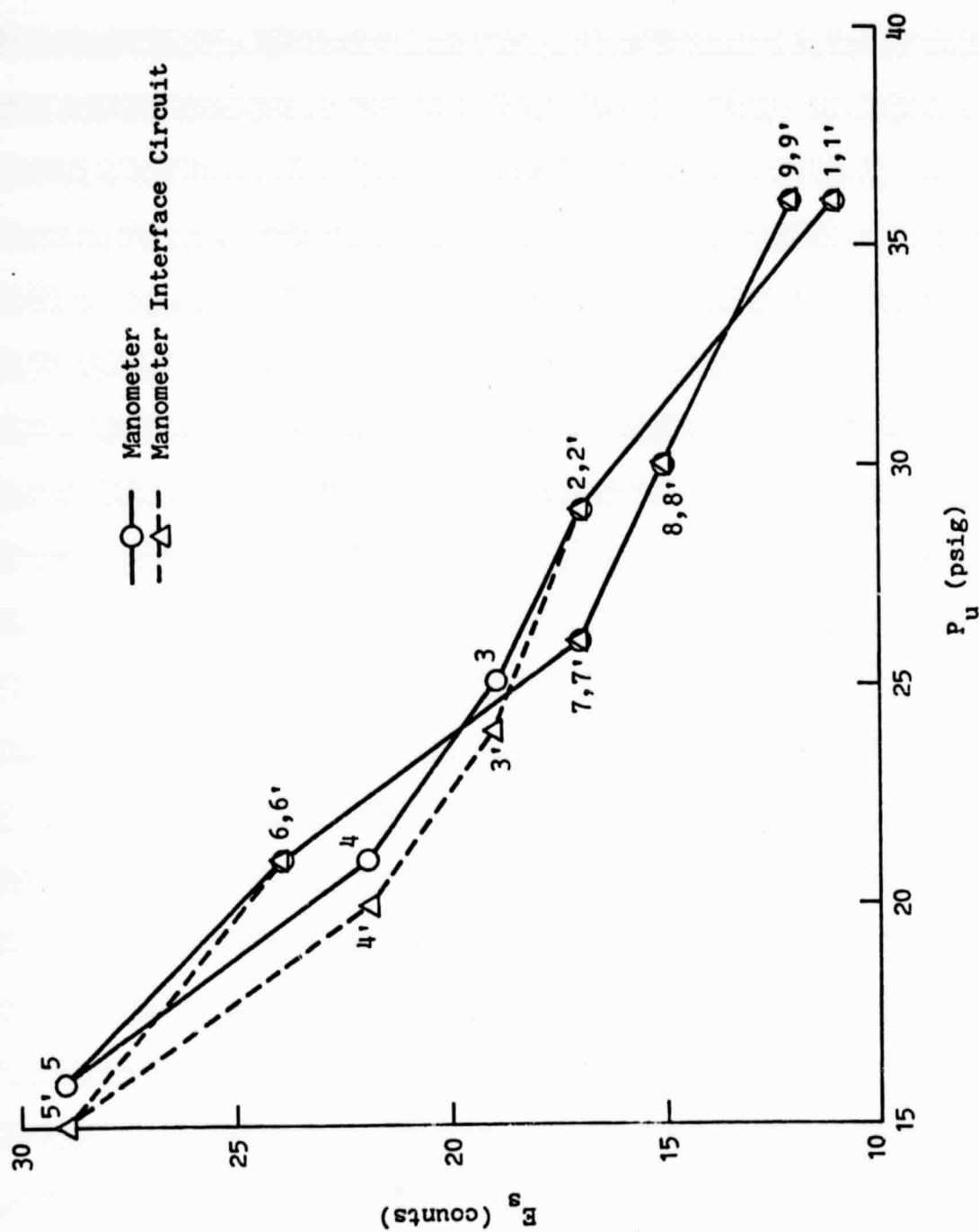


Fig. 15(b) IOLE output vs. upstream pressure; $Q_r = 708$ SCCS.

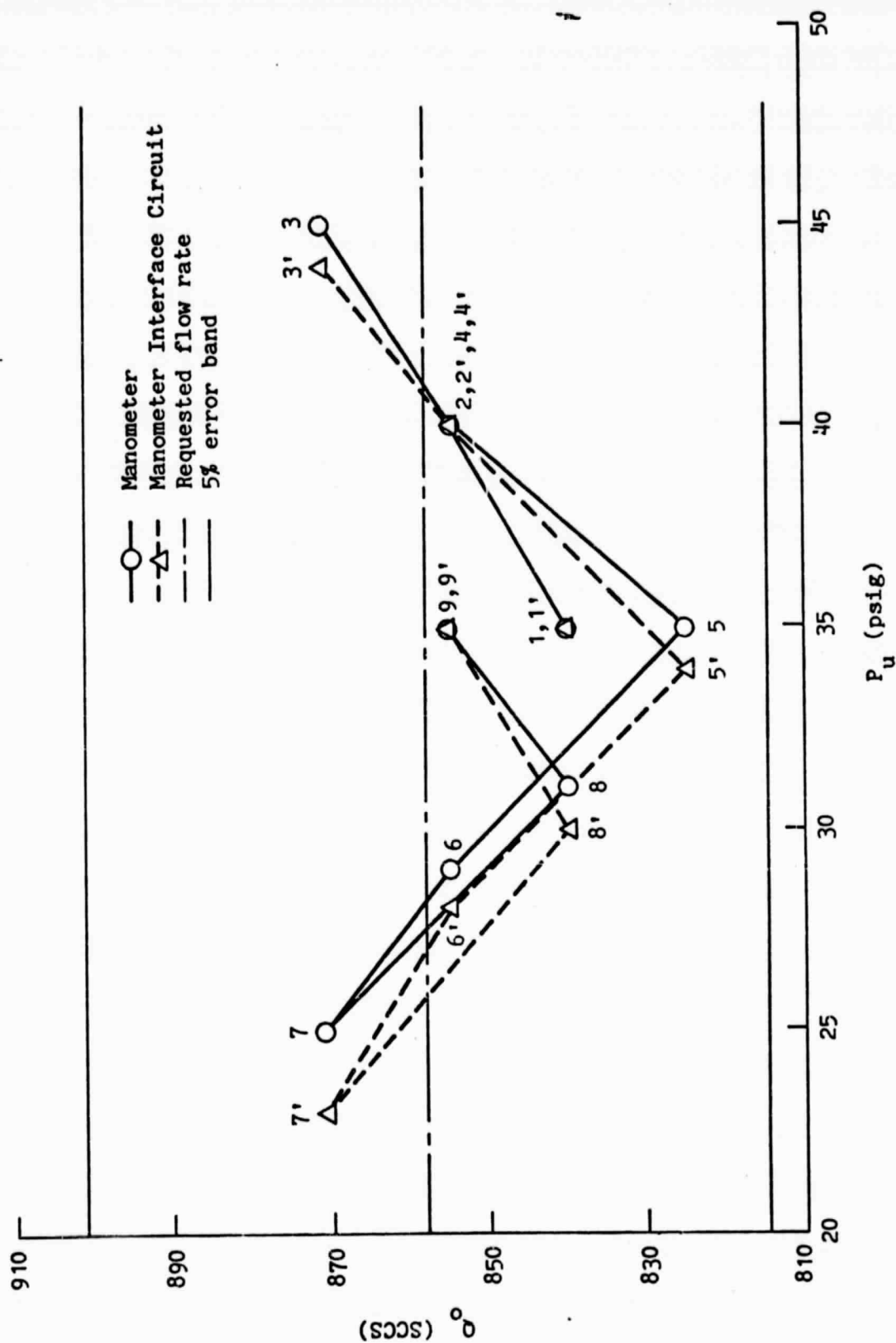


Fig. 16(a) Output volumetric flow rate vs. upstream pressure; $Q_r = 858$ SCCS.

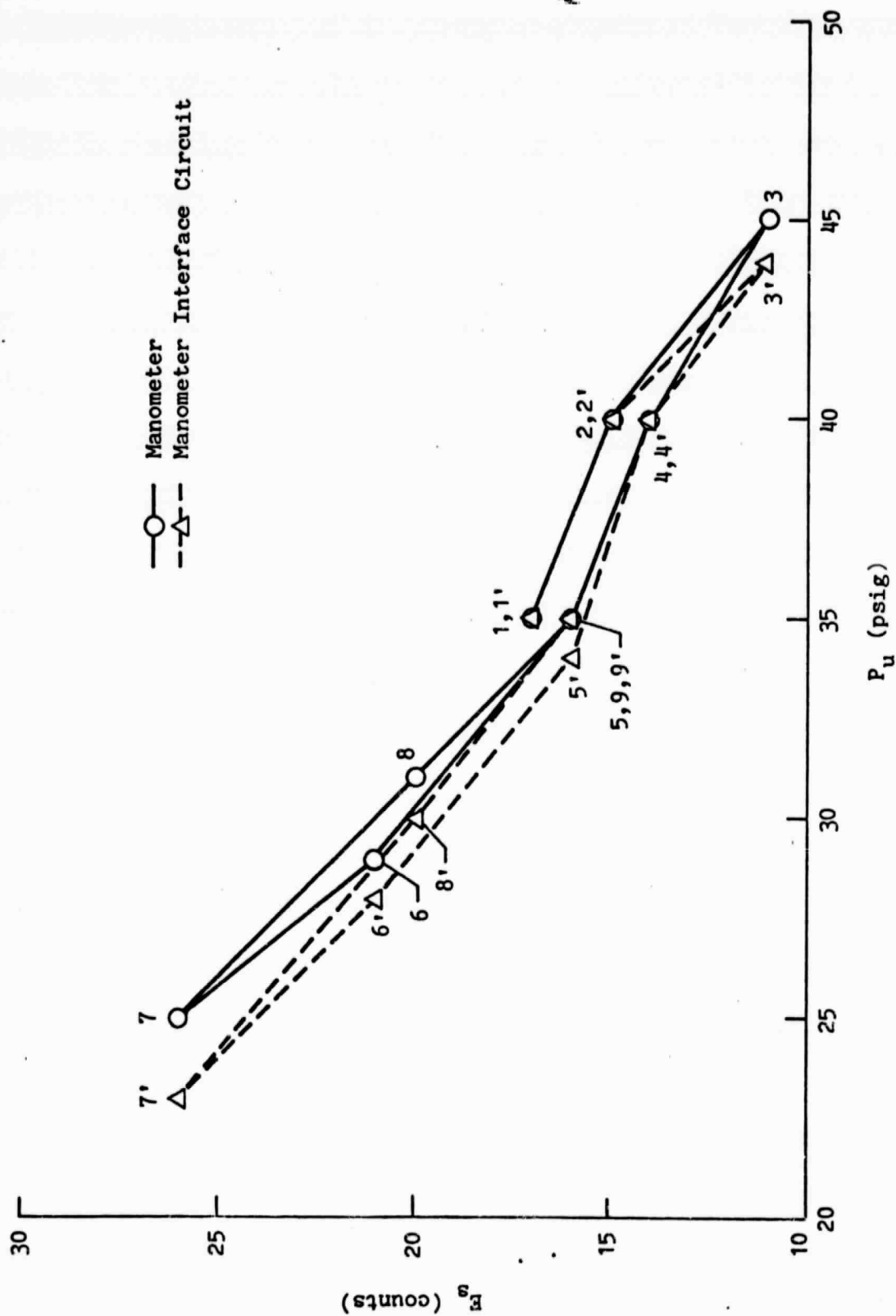


Fig. 16(b) IOLE output vs. upstream pressure; $Q_r = 858$ SCCS.

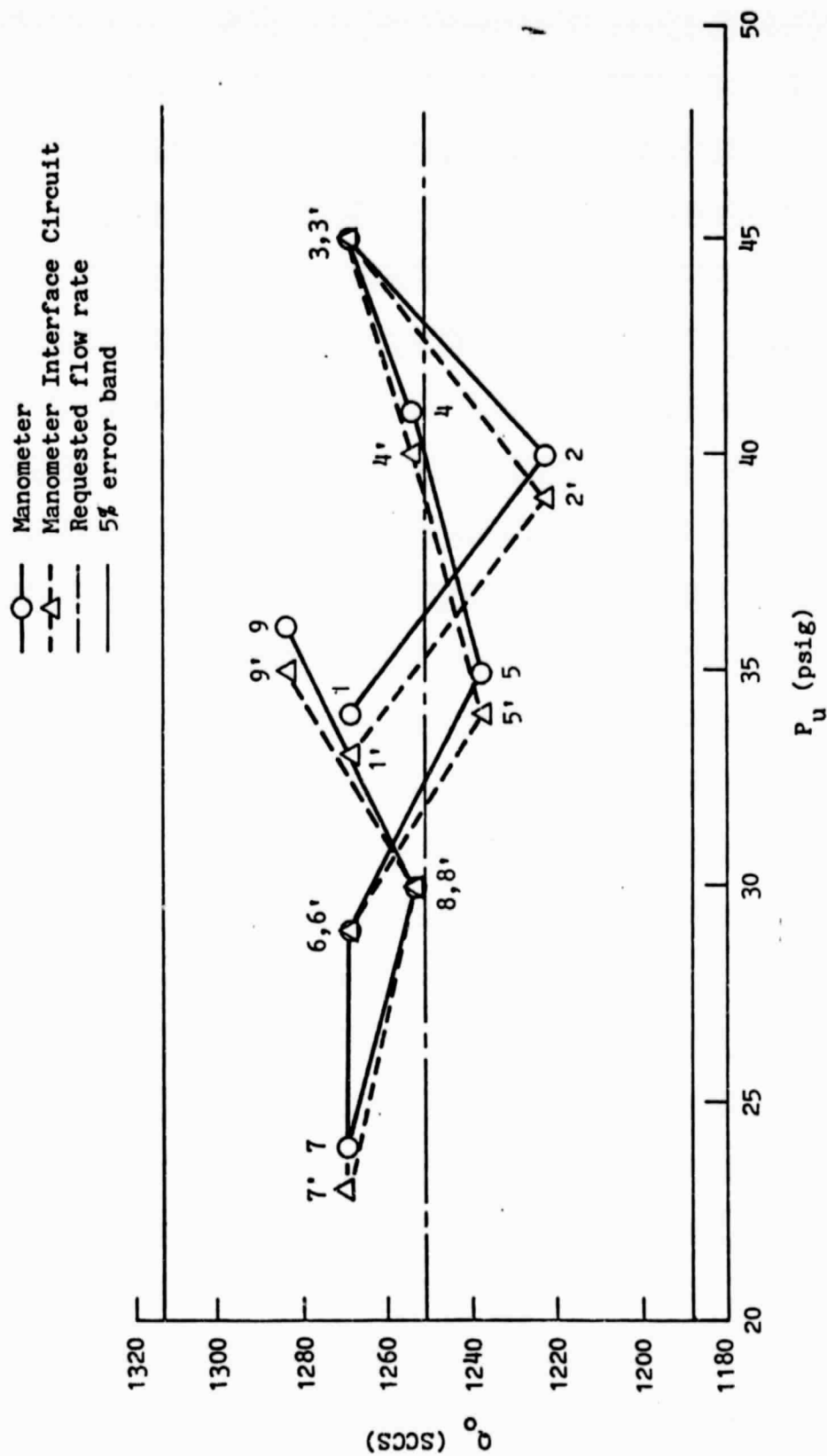


Fig. 17(a) Output volumetric flow rate vs. upstream pressure; $Q_r = 1251$ SCCS.

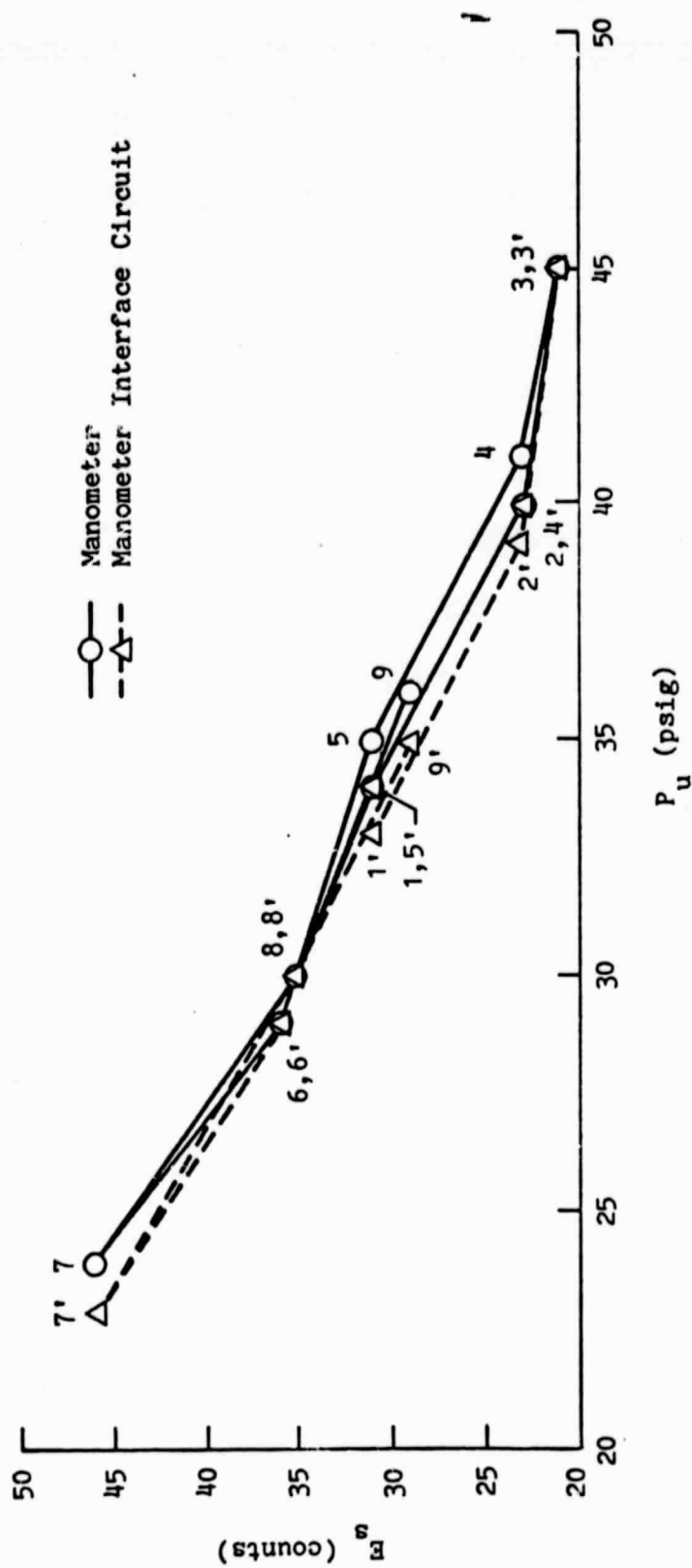


Fig. 17(b) IOLE output vs. upstream pressure; $Q_r = 1251$ SCCS.

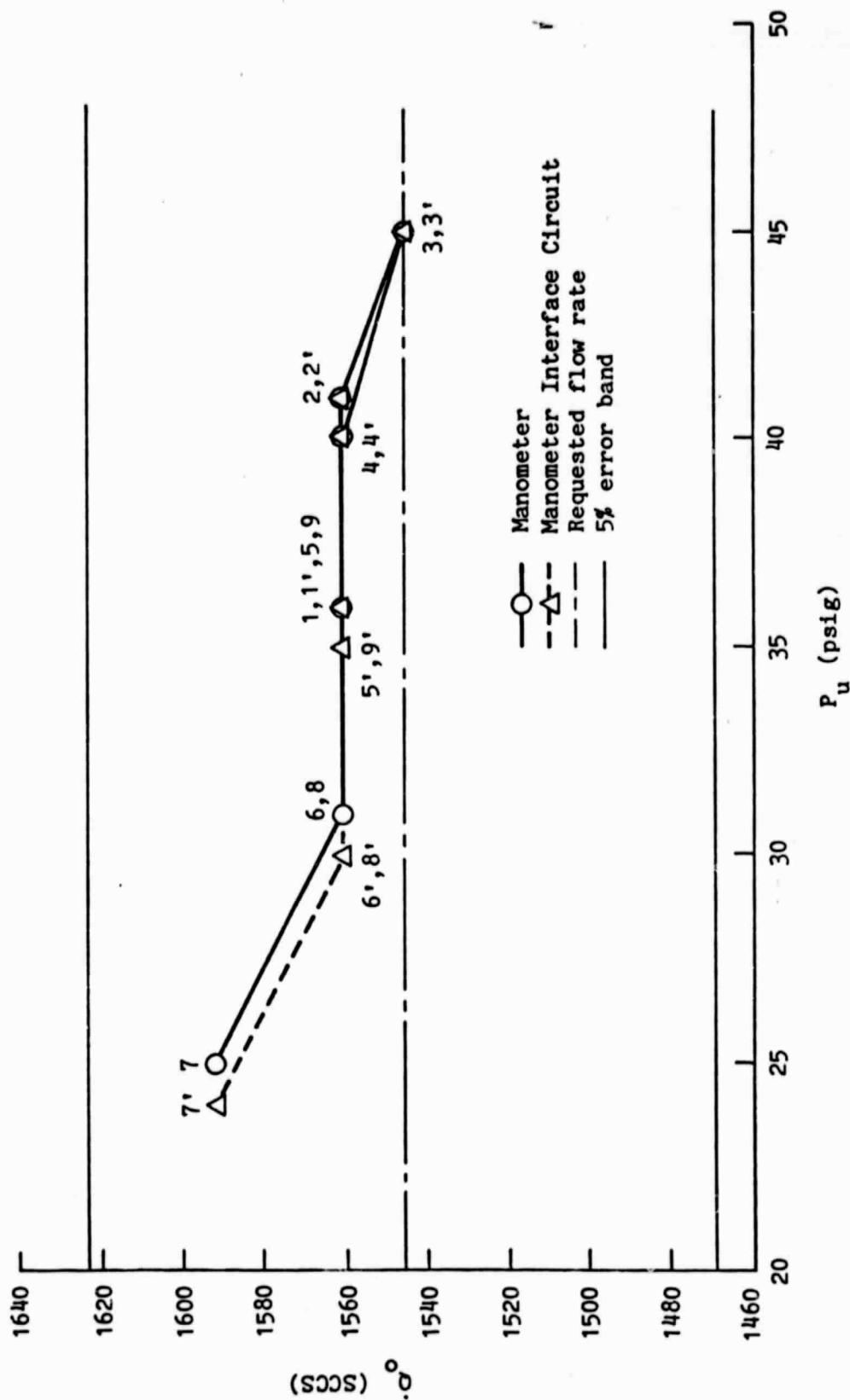


Fig. 18(a) Output volumetric flow rate vs. upstream pressure; $Q_r = 1543$ SCCS.

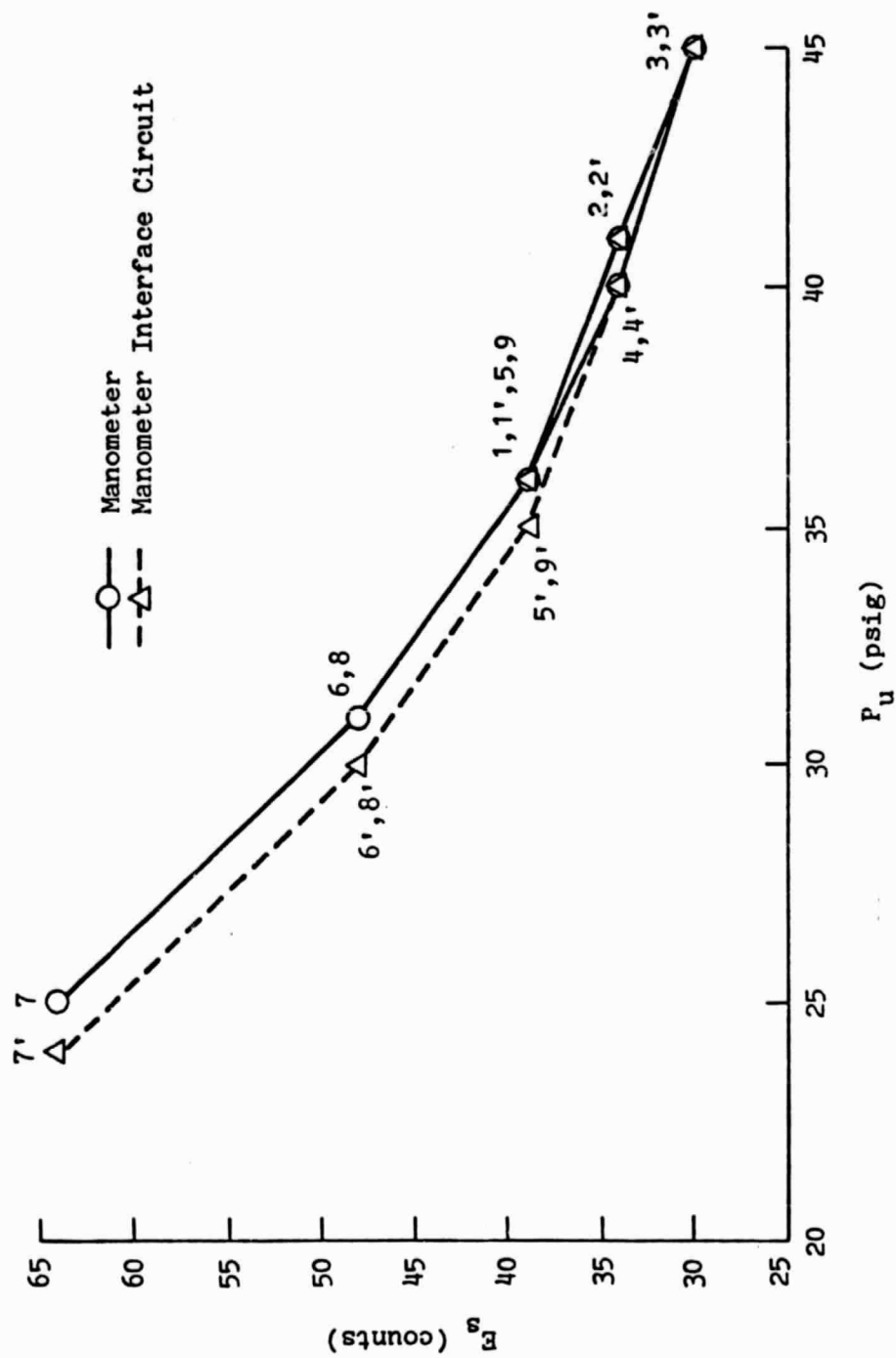


Fig. 18(b) IOLE output vs. upstream pressure; $Q_r = 1543$ SCCS.

sequence. This was due to the poor resolution for valve stem displacement near the closed position (see Fig. 14). For this reason, the lower limit of volumetric flow rate was set at 779 SCCS. Also, so the Overall Program would be valid over the entire range of possible upstream pressures, the upper limit was set at 1543 SCCS. (This is the value of the maximum flow rate at 15 psig).

3.9 Final Testing

With the testing of the overall assembly successful, the final testing of the assembly for 100 hours began. To monitor the assembly during the testing, an APPLE IIe microcomputer was interfaced with the system through a CYBORG model 91A Isaac Data Acquisition System. Four parameters were monitored by the APPLE/CYBORG System: (1) The output from the Manometer Interface Circuit; (2) the number of steps the linear actuator moved; (3) the actuator direction control bit, output on the data bus; and (4) the signal sent to the solenoid in the upstream assembly that diverted flow back to the regulating valve. Anytime the linear actuator moved, the APPLE IIe printed the date, time, and the measured parameters.

The 100 hour test was divided into two parts. Part A consisted of maintaining the upstream pressure regulator at a constant setting and varying the requested volumetric flow rates. The requested values were changed approximately every half hour during the working day. During the night, the Isaac continued to monitor the system for any changes in parameters. The requested flow rates ranged from 779 SCCS to 1543 SCCS. Part B of the testing consisted of keeping the requested flow rate constant and varying the upstream pressure. The

two requested volumetric flow rates tested were 1071 SCCS and 1251 SCCS. A presentation of the data and the discussion are in Chap. 4.

Since the testing was continuous and the system responded to the requested flow rates to within $\pm 5\%$ at all but three data points, the testing was concluded. A photograph of the complete assembly is shown in Fig. 19. Due to the limitations of certain components of the assembly (to be discussed in Chaps. 4 and 5), it was decided that this was the best possible accuracy to be expected. Recommendations for future developments and research are summarized in Chap. 5.

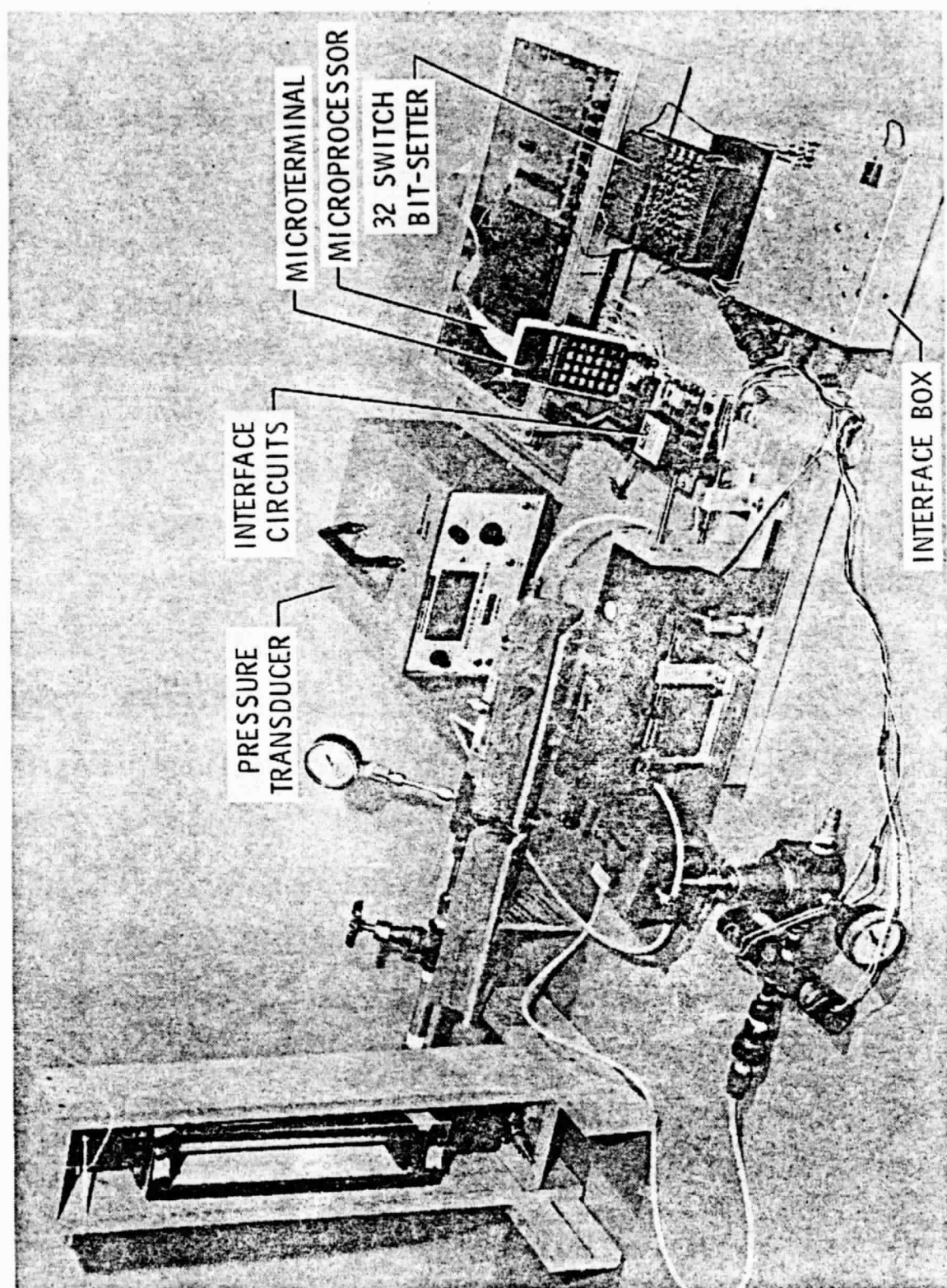


Fig. 19 Photograph of complete assembly.

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Chapter 4

RESULTS AND DISCUSSION

The discussion of the 100 hour test results can be divided into two parts: Part A, where the upstream regulating valve setting was held steady with various requested volumetric flow rates entered via the 32 switch bit-setter; and Part B, where the requested flow rate was held constant and the upstream pressure was varied.

4.1 Part A of the 100 Hour Test

The purpose of Part A of the 100 hour test was to verify that the developed assembly was capable of delivering a requested volumetric flow rate. Also, that the system was capable of operating for an extended period of time. Shown in Fig. 20(a) is the delivered volumetric flow rate (Q_o) versus the requested value (Q_r). The delivered flow rates were all within $\pm 5\%$ of the requested values. The scatter of the Q_o values was due to a combination of three factors: (1) The Manometer Interface Circuit performed the ADC of the manometer output with a resolution of only one psi. Since the manometer had an accuracy of 0.01 psi, this resulted in a round-off of the circuit generated reading. This round-off meant the actual upstream pressure was within plus or minus 0.5 psi of the circuit value. Consequently, the delivered flow rate was either higher or lower than that requested, depending on the actual upstream pressure. (2) The Manometer Interface Circuit had a constant fluctuation of plus or minus one psi. Because the source of this error

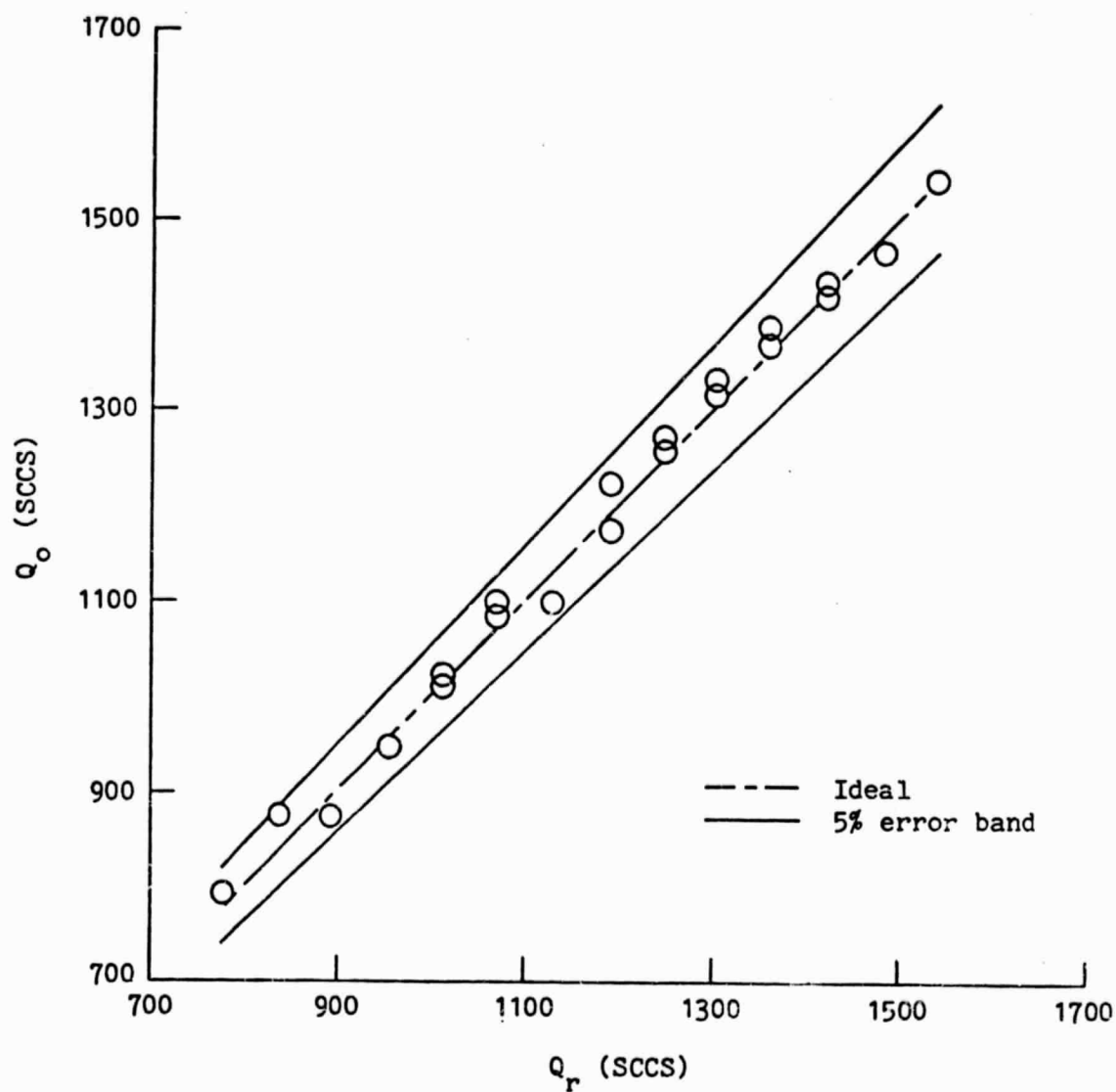


Fig. 20(a) Actual volumetric flow rate vs. requested volumetric flow rate.
100 hour test; Part A.

was never discovered, the Overall Program was altered to compensate. The programming alteration was to prevent this fluctuation from causing the assembly to cycle continuously, thereby making it unstable. The microprocessor system was programmed not to accept any upstream pressure readings, from the circuit, that were within one psi of the value from the previous scan. Because of this, the value of upstream pressure accepted by the microprocessor system could be off by one psi. Since this reading was used to calculate the correct valve stem displacement, the error could cause the stem to be incorrectly positioned. This added to the scatter of delivered flow rates. (3) The linear actuator moved a discrete distance with each step. With only 80 displacements, the number of possible volumetric flow rates was limited. This meant that although a particular flow rate may have been requested, it was not necessarily achievable. In such an instance, the system was designed to deliver the flow rate closest to the requested value. Calculated values of the correct displacements, in steps, are in agreement with these collected by the APPLE IIe microcomputer. It is believed that these errors, individually or in combination, are responsible for the scatter. The results shown in Fig. 20(b) are of the IOLE output versus the requested flow rate (Q_r). Unlike Figs. 15(b) - 18(b), the symbols used represent the direction of the valve stem movement; i.e. opening and closing. While the IOLE output had an error band of plus or minus three counts, the results show that the linear actuator moved in a controlled fashion. Calculations were again made for the correct valve stem displacements (in steps) using the control

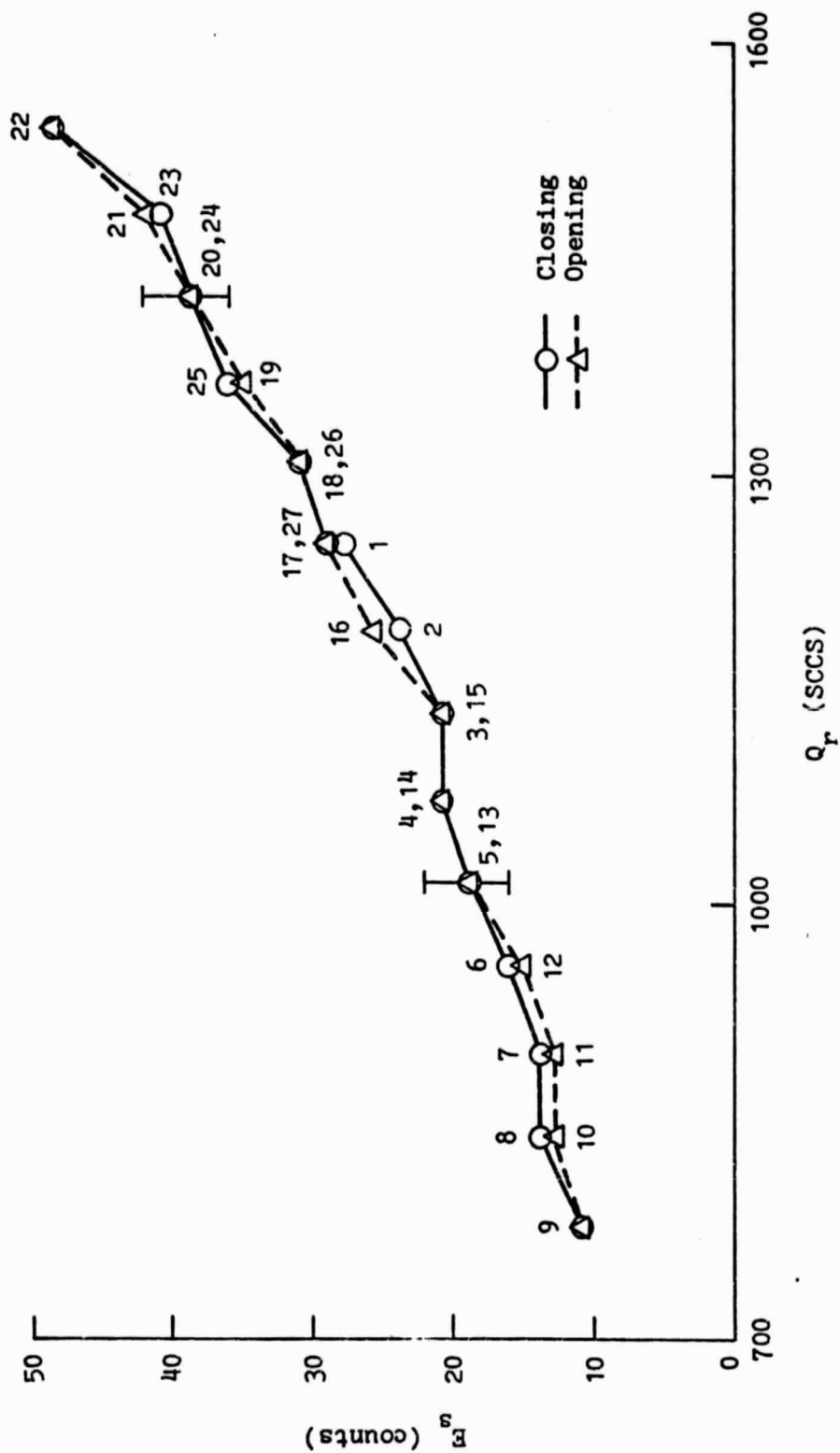


Fig. 20(b) IOLE output vs. requested volumetric flow rate. 100 hour test, Part A.

parameters of upstream pressure and requested flow rate. Comparison of these calculated displacements to those recorded by the APPLE IIe showed the microprocessor system correctly positioned the valve stem.

4.2 Part B of the 100 Hour Test

The purpose of Part B of the 100 hour test was to verify that the developed assembly was capable of maintaining a requested volumetric flow rate in the presence of upstream pressure fluctuations. The two flow rates tested were 1071 SCCS and 1251 SCCS.

Shown in Figs. 21(a) and 21(b) are the results of the 1071 SCCS test. As in Figs. 15-18, the data points are numbered to allow the sequence of testing to be followed. Again, the actual upstream pressure and the manometer signal converted by the Manometer Interface Circuit were not always equal. However, the delivered volumetric flow rates were all within $\pm 5\%$ of the requested value, and the IOLE output showed the linear actuator traveled along a repeatable path. As before, the calculated displacements were in agreement with the values recorded by the APPLE IIe microcomputer.

The results of the 1251 SCCS test are shown in Figs. 22(a) and 22(b). These graphs are similar to the previous ones except for three items: (1) In Fig. 22(a), between points two and three and between points five and six, there are two increases in Q_0 without an accompanying change in P_u . This is due to the lack of resolution in the Manometer Interface Circuit. While a change in the upstream pressure of up to one psi would change the delivered flow

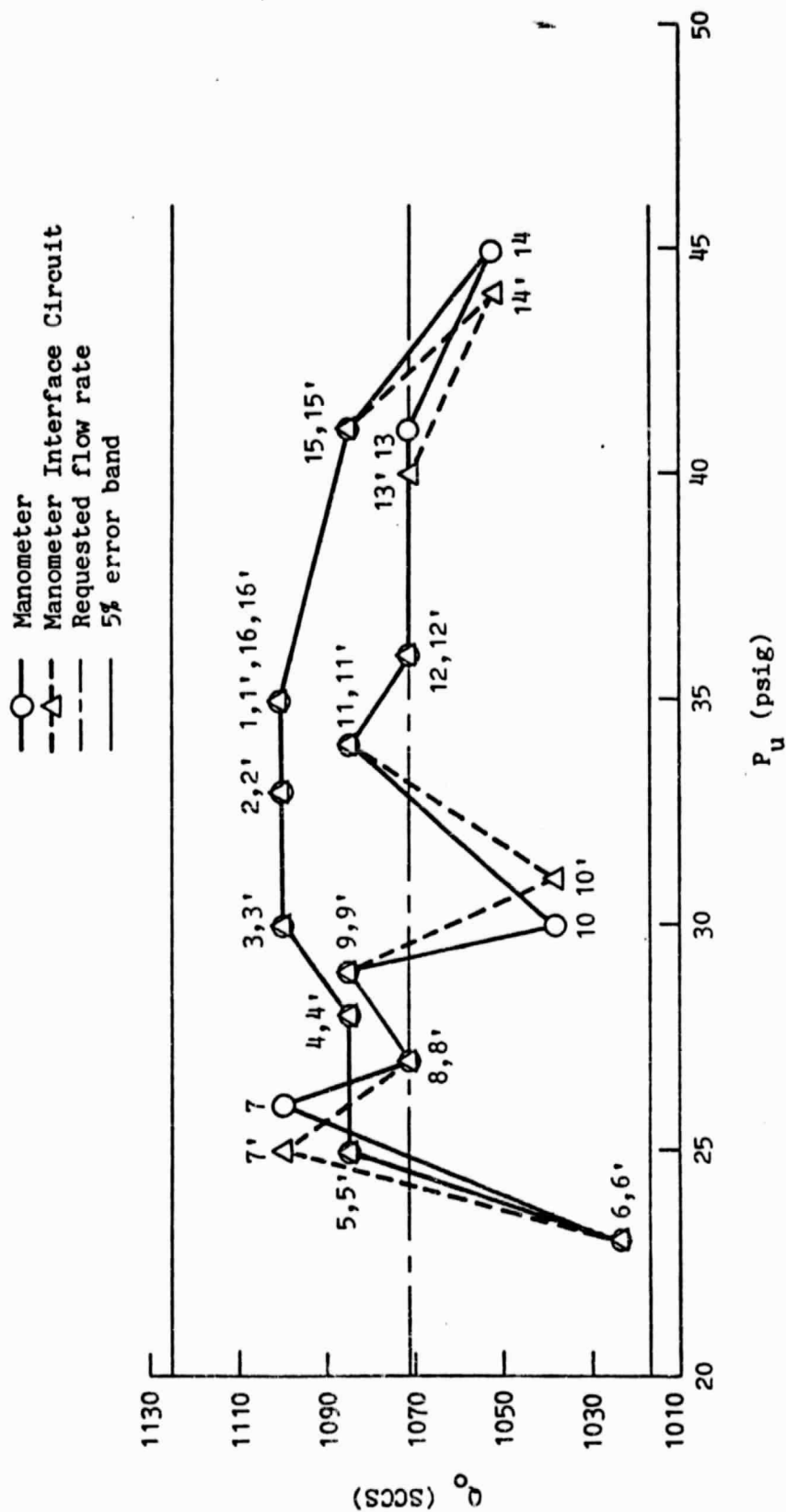


Fig. 21(a) 100 hour test, Part B. Output volumetric flow rate vs. upstream pressure; $Q_r = 1071$ SCCS.

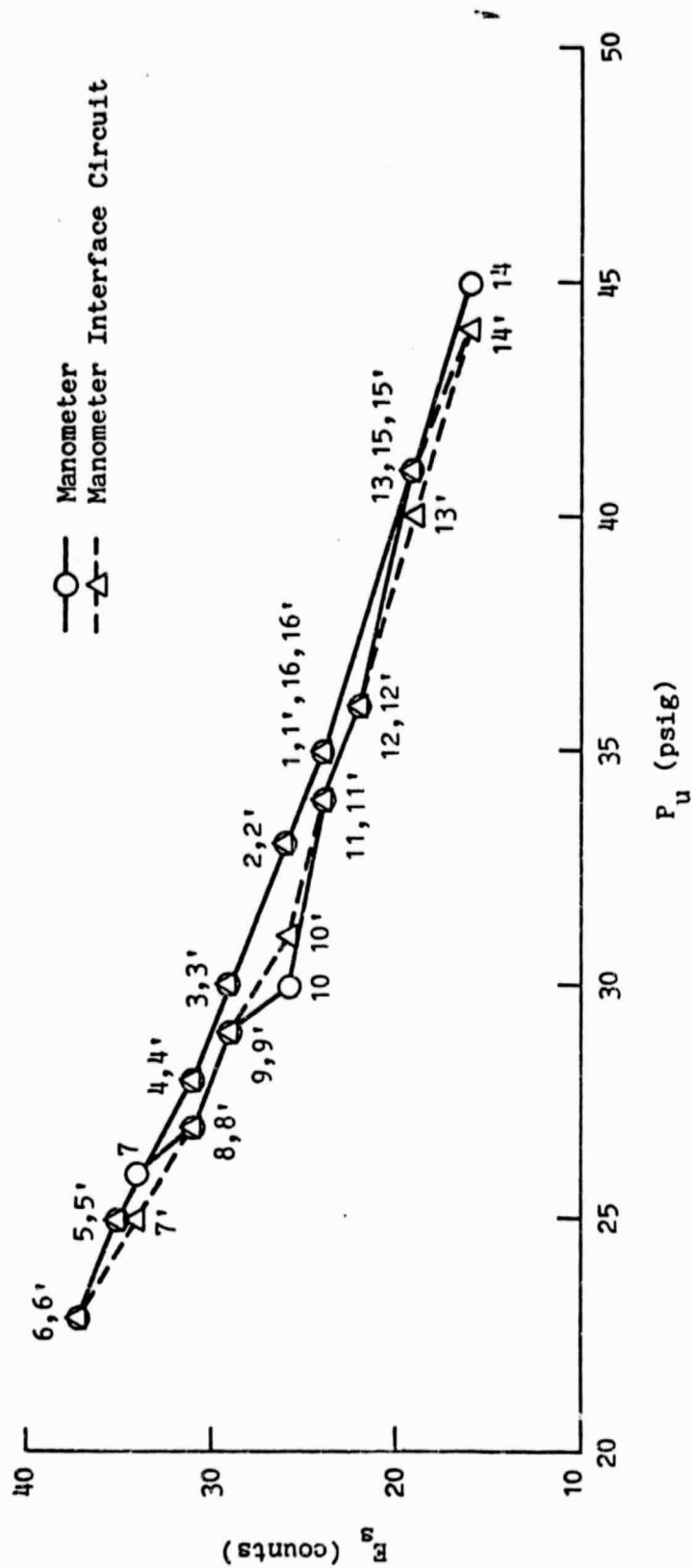


Fig. 21(b) 100 hour test, Part B. IOLE output vs. upstream pressure, $Q_r = 1071$ SCCS.

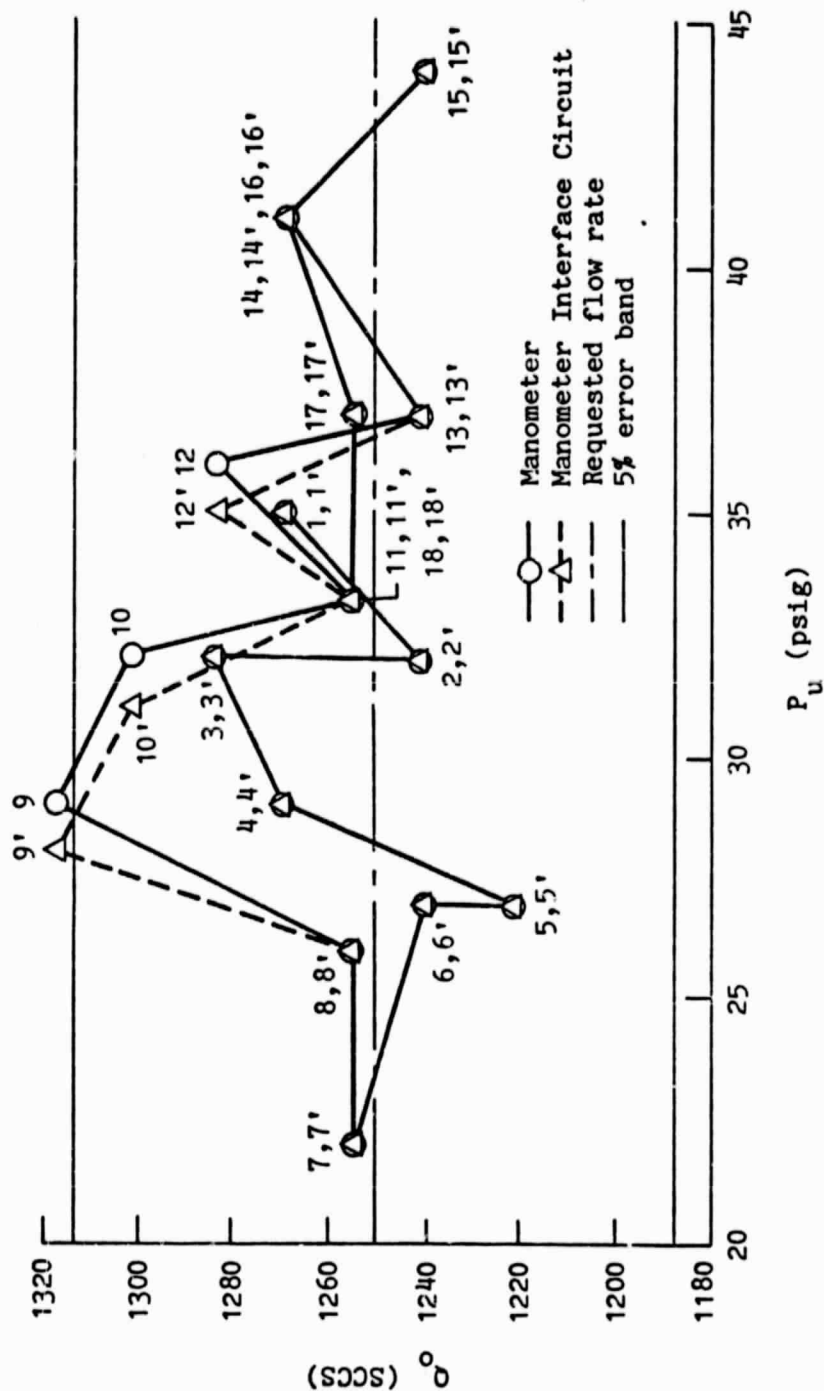


Fig. 22(a) 100 hour test, Part B. Output volumetric flow rate vs. upstream pressure; $Q_r = 1251$ SCCS.

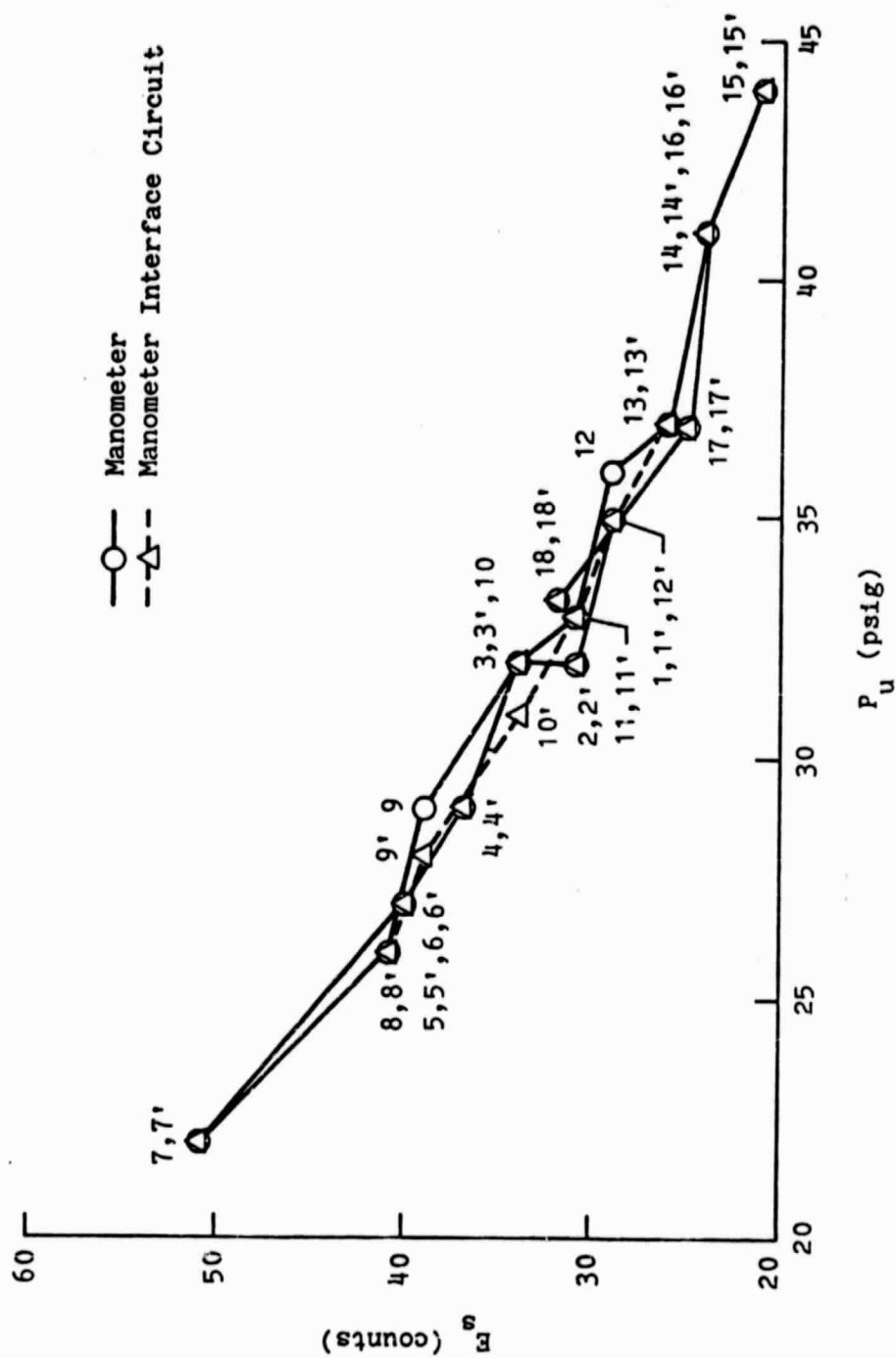


Fig. 22(b) 100 hour test, Part B. IOLE output vs upstream pressure; $Q_r = 1251$ SCCS.

rate, the upstream pressure reading of the circuit did not reflect this (e.g., a pressure between 37.5-38.4 psig appeared as 38 psig on the Manometer Interface Circuit). (2) Also in Fig. 22(a), point nine is outside the 5% error band. This was again due to the Manometer Interface Circuit. The microprocessor system received an upstream pressure reading of 28 psig while it was actually 29 psig. This one psi difference accounts for an error in the microprocessor calculated displacement of the valve stem and hence, the flow rate is 5.43% greater than the requested value. It should be noted that the value of the closest interpolated flow rate for that upstream pressure is 1275 SCCS; already 1.86% over the requested value. (3) In Fig. 22(b), the vertical rise between points two and three was due to the fluctuation of the Manometer Interface Circuit. To illustrate, assume the correct upstream pressure is 32 psig. The circuit would then read between 31-33 psig. But if the correct upstream pressure fluctuates to 32.5 psi, then round-off error by the circuit will cause the reading to be 33±1 psig. This means when the reading fluctuates to 34 psig, the Overall Program would note the two psi difference between sequential readings. It would then compensate for this change by repositioning the valve stem. As the valve stem moves, the pressure drops due to the greater head loss. The resulting pressure could now drop to the original value 32 psig. Hence, a change in E_s without an accompanying change in P_u .

4.3 Compressibility

Because the flowing fluid is a compressible gas, compressibility effects must be considered. This is necessary because of the

dependency of the rotameter on the viscosity of the fluid being studied [6]. An analysis of the velocity of the air through the regulating valve shows the maximum value to be 59.75 m/sec for the valve fully open and $P_u = 47$ psig. This equates to a valve orifice Mach number of 0.2; but according to White [7] the effects of compressibility are negligible for velocities under 0.3 Mach. As this is the greatest velocity through the valve, compressibility effects can be neglected.

4.4 Power Requirement

The two solenoids in the upstream assembly and the linear actuator are each rated at 12 W (12V at 1A). The maximum power consumption of 13.3 W occurred during the valve stem movement phase of the operation, because the three 12 W components were never energized simultaneously. A power consumption of 1.3 W was required for normal operation of the system while the actuator was stationary.

Consequently, a single 24 W (12V at 2A) DC power source would be sufficient to drive the entire system. It should be noted that the circuitry for the operational amplifiers used in the system could be modified to operate with a single-sided 12V supply. Additionally, the 5V supply required to operate the RCA 1802 Evaluation System can be eliminated since a 12V version of the system exists.

Chapter 5

CONCLUSIONS AND RECOMMENDATIONS

While certain components of this assembly performed below expectations (i.e., the Manometer Interface Circuit and the IOLE), the overall development was considered satisfactory for two reasons. First, experimental results demonstrated that an electro-mechanical proportional flow controller was feasible. Of the 88 delivered volumetric flow rates (Q_0) shown in Figs. 15-22, 98.9% of them were within $\pm 5\%$ of the requested values. The assembly was able to reposition for changes in either upstream pressure or the requested flow rate. Second, this development has shown what areas of such a design need further research. These areas are the bellows spring of the regulating valve, the displacement sensor, and a compact volumetric flow rate measurement device. (It should be noted that the components used in this development were neither state-of-the-art, nor designed for this type of application).

5.1 The Bellows Spring

The regulating valve used in the development was rated to a pressure of 6.89 MPa. Because this was much greater than the system maximum operating pressure of 446 kPa, the bellows spring had a greater stiffness than was necessary. This excessive stiffness was the reason for the addition of the upstream assembly and the alterations to the Overall Program. Even when the valve was unpressurized, a force of 53 N was required to close the valve

completely. This meant only 31 N of linear force was available, from the actuator, to move the valve stem when the system was pressurized. With a bellows cross-sectional area of approximately 1.0 cm^2 , the actuator was unable to reposition the valve stem for pressures over 267 kPa gage. Therefore, it is recommended, for future designs, the stiffness of the bellows spring be designed (or selected) to compliment the system maximum operating pressure (i.e., a valve operating at 446 kPa requires a bellows of much less stiffness than a valve operating at 6.89 MPa).

5.2 The Displacement Sensor

The IOLE used in this development was found to be too sensitive to vibrations from the actuator. The accuracy of the displacement sensor in a feedback control system is very important. Because an ADC would already be present to convert the output of the pressure sensor, it is felt that an LVDT, or another displacement sensor with comparable vibration insensitivity, could be used in future designs. Commercially available LVDT's were found that were relatively insensitive to vibration and capable of operating at cryogenic temperatures [3].

5.3 Volumetric Flow Rate Measurement Device

The developed assembly is an example of indirect controls. The delivered volumetric flow rate was monitored as a function of the upstream pressure and the valve stem displacement. This resulted in a delay by the control algorithm, each time a new displacement was calculated. The development of a compact volumetric flow rate measurement device would allow the assembly to be controlled by a

direct feedback signal. This, in turn, would eliminate the need for the complex control algorithm, the displacement sensor, and the pressure sensor. The system would then be able to operate as a quickly responding, continuous system; rather than as a regulator, which has a slower response.

5.4 Other Proportional Flow Controllers

Because the proportional flow controller was shown to be feasible, it is recommended that future research in this area include a mass flow rate controller, and an assembly actuated by a rotary stepping motor. (The concept of rotary motion had been considered in this development, but was not pursued due to constraints on time, manpower, and funds).

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APPENDICES

APPENDIX A

CIRCUIT DIAGRAMS

The figures collected in this appendix are of the interface circuitry, and the control circuitry. The figures listed are:

A.1 IOLE Interface Circuit

A.2 (a) Circuit diagram of Interface Box

(b) Linear actuator power relay

(c) Relay circuit for diverting valve, closing solenoid

(d) Relay circuit for diverting valve, opening solenoid

A.3 Manometer Interface Circuit

A.4 Schematic of the eight switch bit-setter

A.5 Schematic of the 32 switch bit-setter

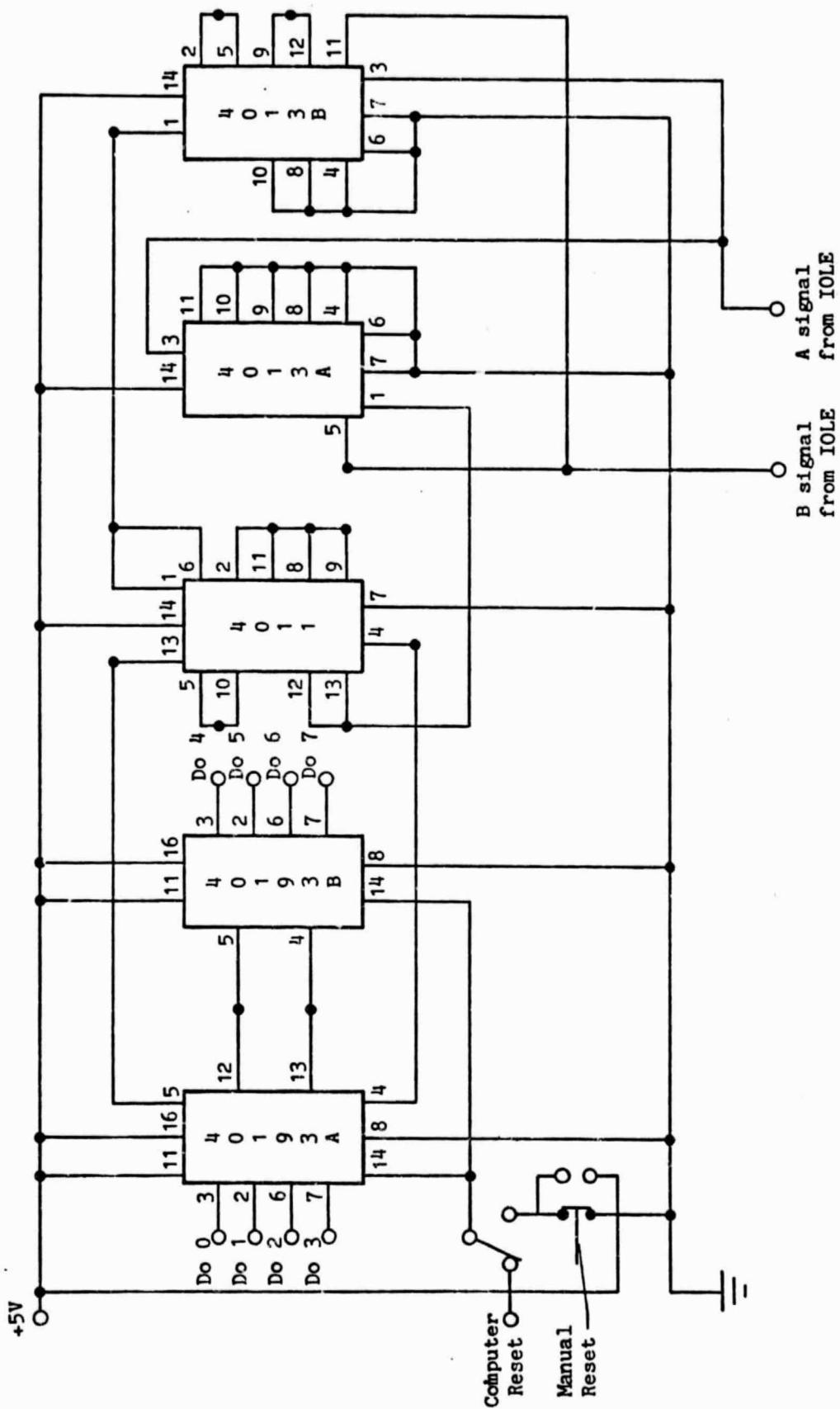


Fig. A.1 IOLE Interface Circuit.

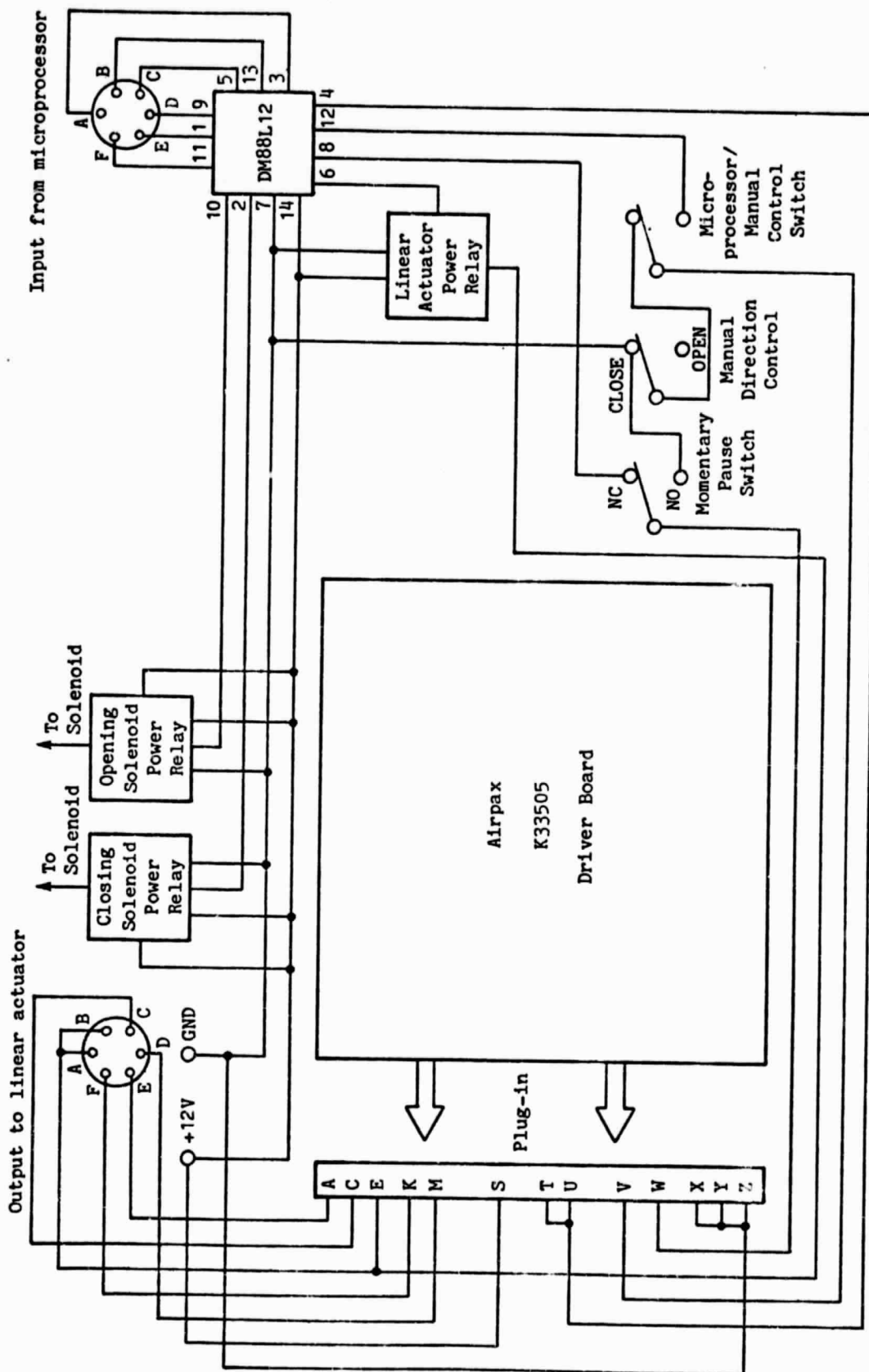


Fig. A.2(a) Circuit diagram of Interface Box.

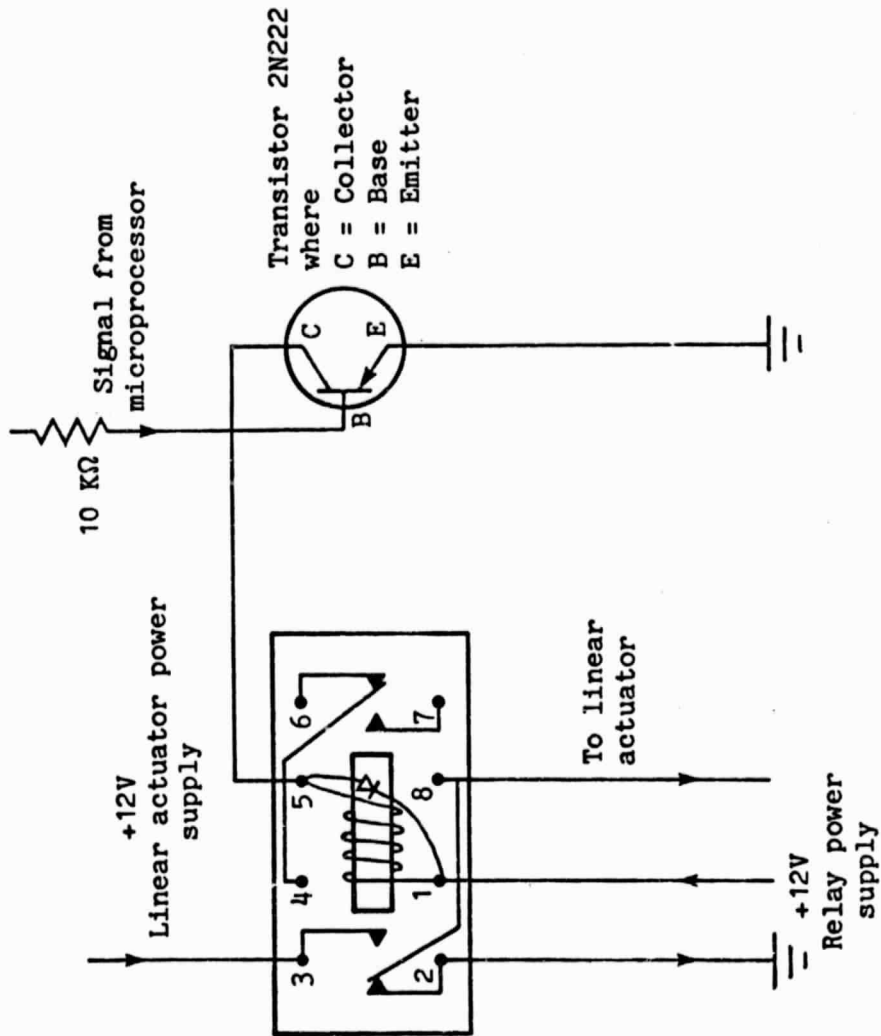


Fig. A.2(b) Linear actuator power relay.

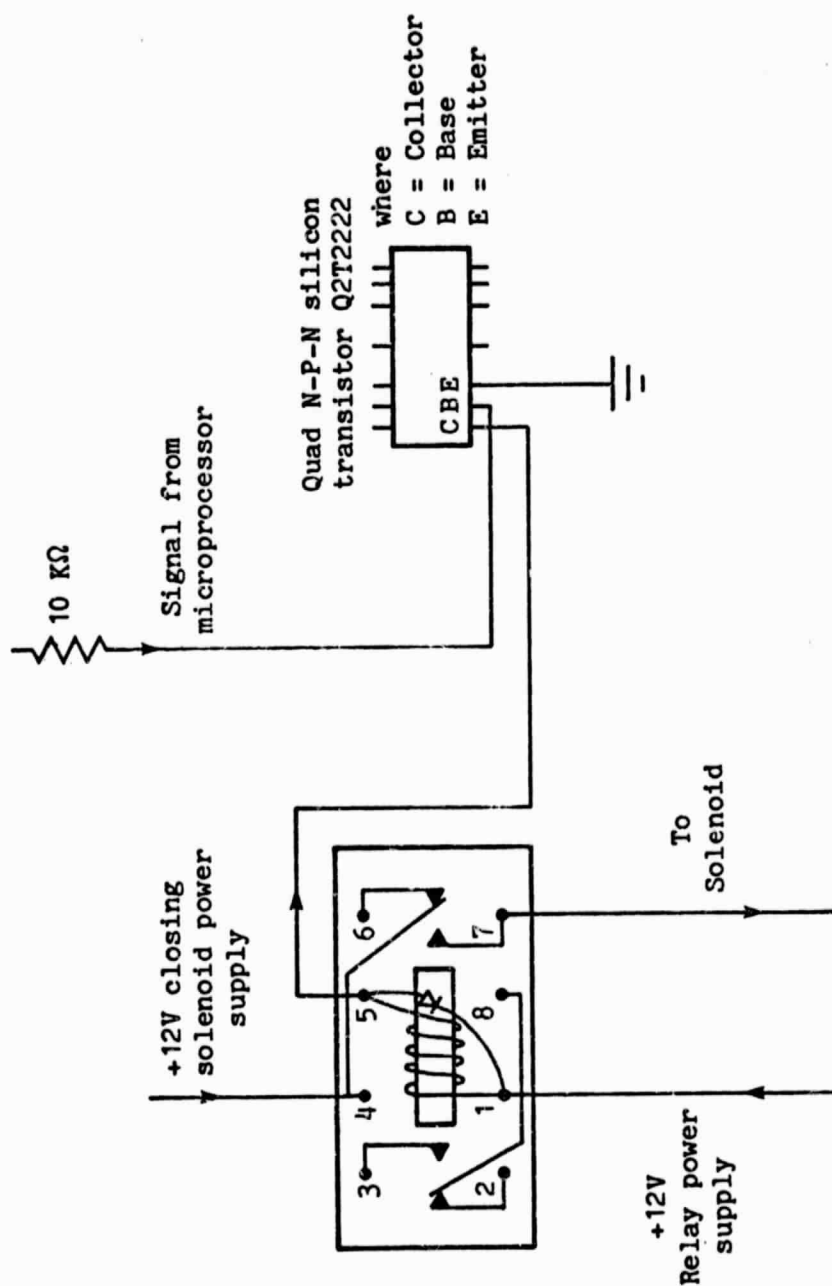


Fig. A.2(c) Relay circuit for diverting valve closing solenoid.

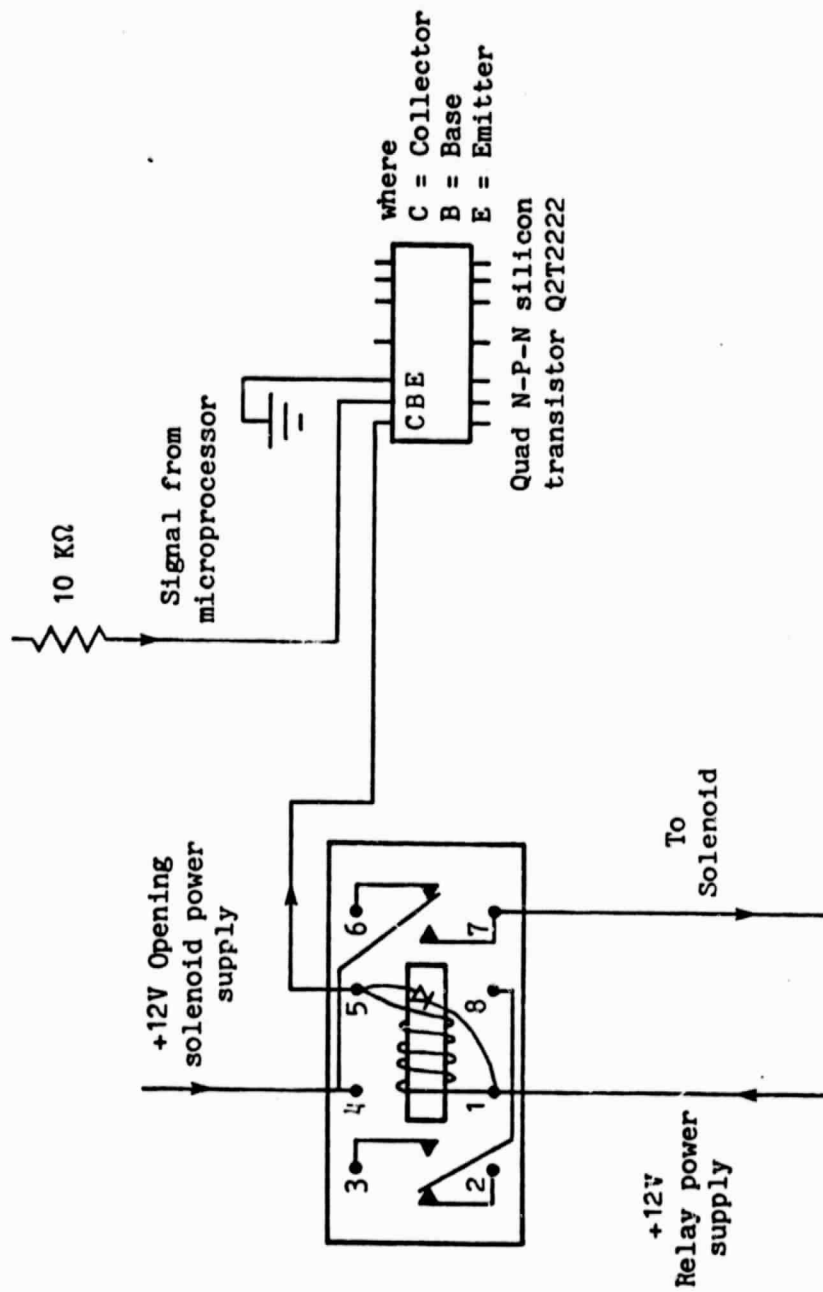
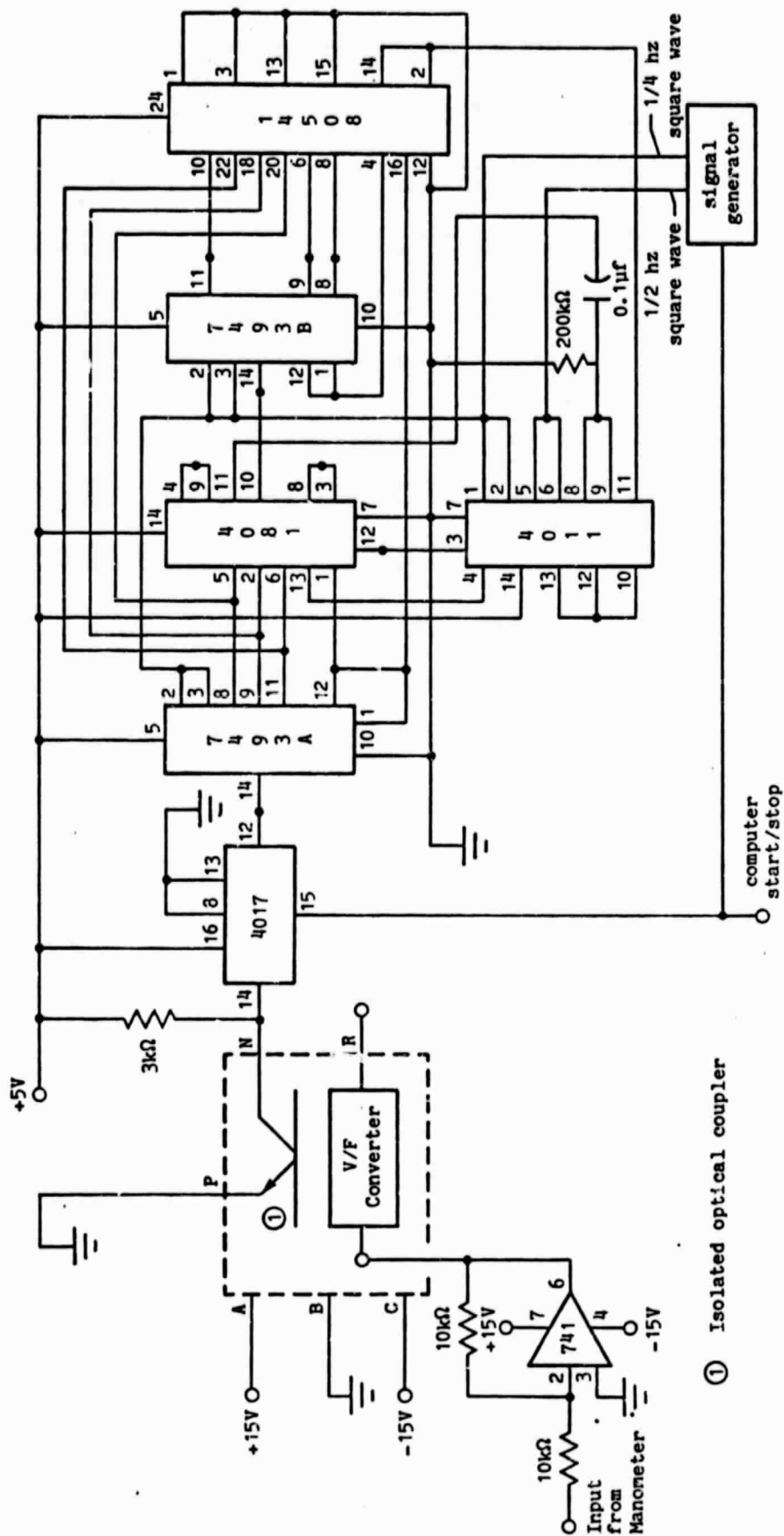


Fig. A.2(d) Relay circuit for diverting valve opening solenoid.



① Isolated optical coupler

Fig. A.3 Manometer Interface Circuit.

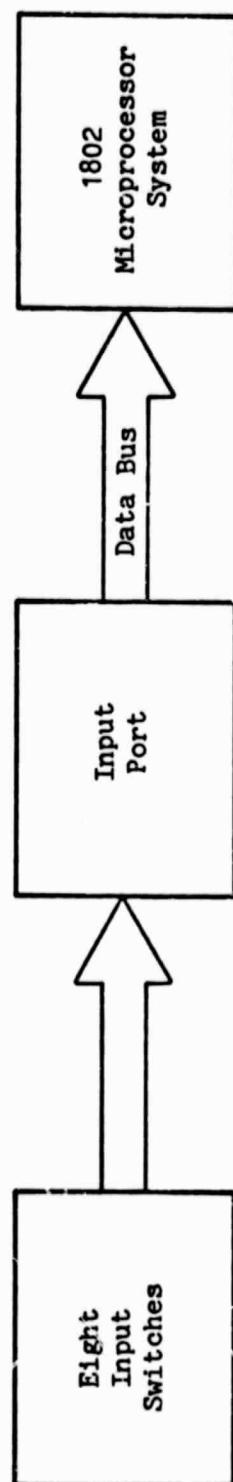


Fig. A.4 Schematic of eight switch bit-setter.

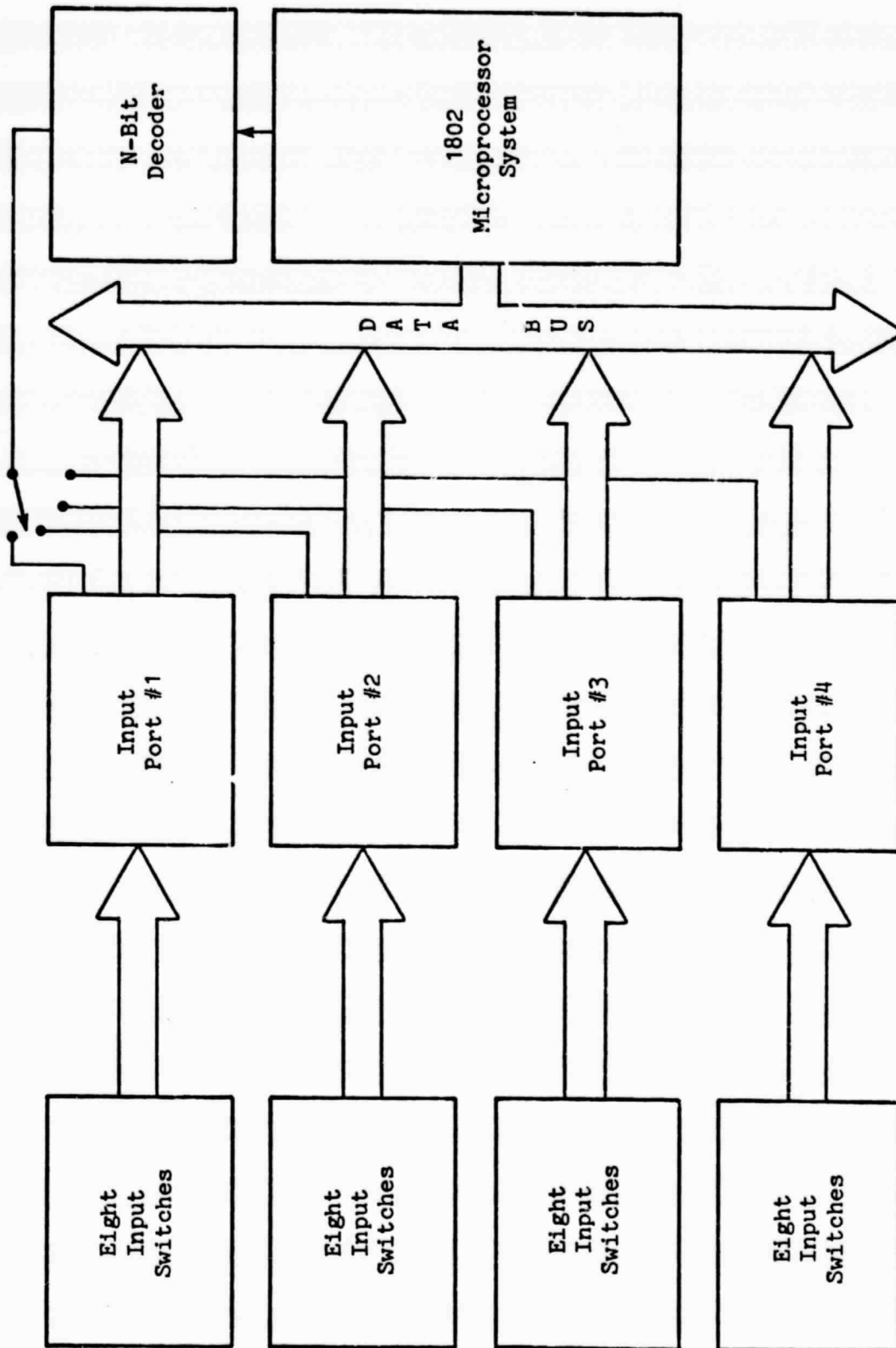


Fig. A.5 Schematic of 32 switch bit-setter.

APPENDIX B

DEVELOPMENT MICROPROCESSOR PROGRAMS

There were four microprocessor programs developed during the course of this development. They were the Cycling Program, the Actuator Movement Program, the Interpolation Program, and the Overall Program. The first three are listed in the following pages along with comments, to explain the programming techniques utilized.

CYCLING PROGRAM

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	DIS	#00	.. Disable interrupts
	LBR	ENTRY	..
	LDI	#00	
	STXD		
SET:	LDI	#00	.. Set RB = #0000
	PHI	RB	..
	PLO	RB	..
LOOP:	INP 6		.. Read input switches
	BZ	LOOP	.. Keep reading until input \neq 0
	PHI	RA	.. Store input into RA hi and lo bytes
	PLO	RA	..
	SHR		.. Divide input by 8 and ..
	SHR		
	SHR		
	SHR		
	PHI	RA	.. Store in RA.1
	GLO	RA	.. Get input from RA.0
	SHL		.. Multiply by 8 and ..
	SHL		
	SHL		
	SHL		
	PLO	RA	.. Store in RA.0
	IRX		
	LDX		
	STXD		
	BNZ	ZY	.. If D \neq 0 jump to ZY
	SEP	R4	.. Output #00 on data bus
	,A(OUTPUT)	;#00	
	SEP	R4	.. Output #01 on data bus
	,A(OUTPUT)	;#01	
	SEP	R4	
	,A(INCR)		.. Increment RB until RB = #0050 ₁₆
	BNF	LOOP	.. If DF = 0, jump to LOOP
	IRX		
	LDI	#01	
	STXD		
	BR	SFT	.. Go to SET
ZY:	SEP	R4	.. Output #02
	,A(OUTPUT)	;#02	..
	SEP	R4	.. Output #03
	,A(OUTPUT)	;#03	..
	SEP	R4	.. Increment RB until RB = #0050 ₁₆
	OBJECT		

<u>LABEL</u>	<u>CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	,A(INCR)		..
	BNF	LOOP	.. If DF = 0, jump to LOOP
	IRX		
	BR	SET	
OUTPUT:	LDA	R6	*** Output Subroutine ***
	STXD		.. Store number to be output
	IRX		
	OUT 5		.. Output number on data bus
	DEC	R2	
	SEP	R4	.. Countdown RA during output for delay
	,A(COUNT)		
	SEP	R5	.. Return to calling program
INCR:	INC	RB	*** Increment RB Subroutine ***
	GLO	RB	
	XRI	#50	.. Compare RB.0 to #50 ₁₆
	BNZ	XX	.. If RB.0 \neq #50 ₁₆ , jump to XX
	GHI	RB	.. Compare RB.1 to #00
	XRI	#00	
	BNZ	XX	.. If RB.0 \neq 00, jump to XX
	LDI	#01	.. Set DF = #01
	SHR		
	SEP	R5	.. Return to calling program
XX:	LDI	#00	.. Set DF = #00
	SHR		
	SEP	R5	.. Return to calling program
COUNT:	GHI	RA	*** Countdown RA Subroutine ***
	STXD		.. Save RA on stack
	GLO	RA	..
	STXD		..
DOWN:	DEC	RA	.. Decrement RA
	GLO	RA	
	BNZ	DOWN	.. If RA.0 \neq 0, jump to DOWN and repeat
	GHI	RA	
	BZ	OVER	.. If RA.1 = 0, jump to OVER
	DEC	RA	.. Decrement RA
	BR	DOWN	.. Jump to DOWN
OVER:	IRX		
	LXDA		.. Restore RA
	PLO	RA	..
	LDX		..
	PHI	RA	..
	SEP	R5	.. Return to calling program
	OBJECT		

<u>LABEL</u>	<u>CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
ENTRY:	LDI	#00	***Standard Call and Return Technique ***
	PHI	R3	
	LDI	#05	.. Load return address into R3
	PLO	R3	
INIT:	LDI	A.1(CALL)	.. Load R4 with call address ..
	PHI	R4	..
	PHI	R5	.. Load R5 with return address
	LDI	A.0(CALL)	..
	PLO	R4	..
	LDI	A.0(RETPGM)	..
	PLO	R5	..
	LDI	#8C	
	PHI	R2	.. Load stack address into R2
	PHI	R7	
	LDI	#1F	
	PLO	R2	
	PLO	RC	
	SEX	R2	.. Set stack pointer as R2
	SEP	R3	.. Return to main program
EXITA:	SEP	R3	.. Return to calling program
CALL:	SEX	R2	.. Point to stack
	GHI	R6	.. Save R6 value
	STXD		..
	GLO	R6	..
	STXD		..
	GHI	R3	.. Put R3 ..
	PHI	R6	.. Into R6
	GLO	R3	.. Both halves
	PLC	R6	
	LDA	R6	.. And put subroutine address ..
	PHI	R3	.. Into R3
	LDA	R6	
	PLO	R3	
	BR	EXITA	.. Go to calling subroutine
EXITR:	SEP	R3	.. Return to calling program
RETPGM:	GHI	R6	.. Put R6 back ..
	PHI	R3	.. Into R3
	GLO	R6	..
	PLO	R3	..
	SEX	R2	.. Point to stack

<u>LABEL</u>	<u>OBJECT</u> <u>CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	INC	R2	.. Recover ..
	LDXA		.. R6
	PLO	R6	.. Lower half
	LDX		
	PHI	R6	.. And upper half
	BR	EXITR	.. Go back to calling program
	END		

ACTUATOR MOVEMENT PROGRAM

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	DIS	#00	.. Disable interrupts
	LBR	ENTRY	.. Set up standard call and return technique
MAIN:	LDI	#00	.. Set up registers (R8,RB,RA)
	PLO	RB	.. R8 = #0200 ₁₆
	PHI	RB	..
	PLO	R8	.. RA = #0050 ₁₆
	PHI	RA	..
	LDI	#02	.. RB = #0000
	PHI	R8	..
	LDI	#50	..
	PLO	RA	..
	BR	START	
PS:	SEP	R4	
	,A(READ)		.. Read in displacement (in steps)
	GHI	RA	
	STXD		
	IRX		
	GLO	RA	
	XOR		.. Compare present input to previous
	BZ	PR	.. If same, cut off actuator power
	SEP	R4	
	,A(CLOSE)		.. Call CLOSE subroutine
	GHI	RA	
	PLO	RA	
	SEP	R4	
	,A(OPEN)		.. Call OPEN subroutine
	BR	PS	
PR:	SEP	R4	
	,A(OUTPUT)	; ₁₆ 04	.. Cut off actuator power when ..
	BR	PS	.. it is not in use
START:	SEP	R4	.. Close valve all the way on ..
	,A(CLOSE)		.. start up to verify position
	LDI	#00	
	PLO	RA	
	BR	PS	.. Return to main program
READ:	LDI	#00	*** Data Read Subroutine ***
	SHR		

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	INP 6		.. Read input displacement off data bus
	PHI	RA	
	SMI	#50	.. Compare input to #50 ₁₆
	BDF	RX	.. If input larger than #50 ₁₆ , jump to RX
	BR	RZ	.. Otherwise, jump to RZ
RX:	LDI	#50	.. Disregard input and store #50 ₁₆
	PHI	RA	
RZ:	SEP	R5	.. Return to main program
OUTPUT:	LDA	R6	*** Data Output Subroutine ***
	STXD		.. Take number to be output
	IRX		
	OUT 5		.. Output it on data bus
	DEC	R2	.. Continue to output number until ..
	GHI	R8	.. R8 is counted down to #0000
	STXD		..
	GLO	R8	..
	STXD		..
OX:	DEC	R8	..
	GLO	R8	..
	BNZ	OX	..
	GHI	R8	..
	BZ	OY	..
	DEC	R8	..
	BR	OX	..
OY:	IRX		.. When R8 is #0000, restore.
	LDXA		.. The original number in R8 (#0200 ₁₆)
	PLO	R8	..
	LDX		..
	PHI	R8	..
	SEP	R5	.. Return to calling program
CLOSE:	GLO	RA	*** Valve Closing Subroutine ***
	ADI	#05	.. Add #05 steps to last displacement ..
	PLO	RA	..
CY:	SEP	R4	.. And close down valve before repositioning
	,A(OUTPUT)	; ₁₆ 03	.. Output #03
	SEP	R4	
	,A(OUTPUT)	; ₁₆ 02	.. Output #02
	SEP	R4	
	,A(INCR)		.. Increment RB to RA
	BNF	CY	

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	GLO	RA	.. Subtract #05 from last displacement
	SMI	#05	
	PLO	RA	
	LDI	#00	.. Set RB = #0000
	PLO	RB	
	PHI	RB	
	SEP	R4	
	,A(OUTPUT)	;\$0A	.. Output #0A ₁₆
	SEP	R5	.. Return to calling program
OPEN:	GLO	RA	*** Valve Opening Subroutine ***
	BZ	PT	
PB:	SEP	R4	
	,A(OUTPUT)	;\$01	.. Output #01
	SEP	R4	
	,A(OUTPUT)	;\$00	.. Output #00
	SEP	R4	
	,A(INCR)		.. Increment RB to RA
	BNF	PB	
	BR	PU	
PT:	SEP	R4	
	,A(OUTPUT)	;\$04	.. Output #04 to cut off actuator power
PU:	LDI	#00	
	PLO	RB	.. Set RB = #0000
	PHI	RB	
	SEP	R5	.. Return to calling program
INCR:	INC	RB	*** Incrementing RB to RA Subroutine ***
	GLO	RA	
	STXD		
	IRX		
	GLO	RB	
	XOR		.. Compare RB to RA
	BNZ	PJ	.. If not equal, jump to PJ
	LDI	#01	.. Set DF = #01
	SHR		..
	SEP	R5	.. Return to calling program
PJ:	LDI	#00	.. Set DF = #00
	SHR		..
	SEP	R5	.. Return to calling program
ENTRY:	LDI	#00	*** Standard Call and Return Technique ***
	PHI	R3	
	LDI	#05	.. Load return address into R3
	PLO	R3	

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
INIT:	LDI	A.1(CALL)	.. Load R4 with call address
	PHI	R4	..
	PHI	R5	.. Load R5 with return address
	LDI	A.0(CALL)	..
	PLO	R4	..
	LDI	A.0(RETPGM)	..
	PLO	R5	..
	LDI	#8C	
	PHI	R2	.. Load Stack address into R2
	PHI	R7	
	LDI	#1F	
	PLO	R2	
	PLO	RC	
	SEX	R2	.. Set stack pointer as R2
	SEP	R3	.. Return to main program
EXITA:	SEP	R3	.. Return to calling program
CALL:	SEX	R2	.. Point to stack
	GHI	R6	.. Save R6 value
	STXD		..
	GLO	R6	..
	STXD		..
	GHI	R3	.. Put R3 ..
	PHI	R6	.. Into R6
	GLO	R3	.. Both halves
	PLO	R6	
	LDA	R6	.. And put subroutine address ..
	PHI	R3	.. Into R3
	LDA	R6	
	PLO	R3	
	BR	EXITA	.. Go to calling subroutine
EXITR:	SEP	R3	.. Return to calling program
RETPGM:	GHI	R6	.. Put R6 back ..
	PHI	R3	.. Into R3
	GLO	R6	..
	PLO	R3	..
	SEX	R2	.. Point to stack
	INC	R2	.. Recover ..
	LDXA		.. R6
	PLO	R6	.. Lower half
	LDX		
	PHI	R6	.. And upper half
	BR	EXITR	.. Go back to calling program END

INTERPOLATION PROGRAM

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	DIS	#00	.. Disable interrupts
	LBR	ENTRY	.. Set up standard call and return technique
	LDI	#00	.. Set registers R7, RA, RB = #0000
	PLO	R9	..
	PLO	R7	..
	PHI	R7	..
	PLO	RA	..
	PHI	RA	..
	PLO	RB	..
	PHI	RB	..
	LDI	#08	.. Set R9 = #0800 ₁₆
	PHI	R9	
	LDI	XX	.. Load requested lo byte (XX) ..
	PLO	R8	.. into R8.0
	LDI	YY	.. Load requested hi byte (YY) (YY) ..
	PHI	R8	.. into R8.1
	LDI	ZZ	.. Load system pressure (ZZ) ..
	PHI	R8	.. into RD.0
	GLO	R8	
	STR	R2	.. Put requested flow ..
	GHI	R8	.. rate on stack
	OR		.. Compare hi and lo bytes
	LBZ	VV	.. If both = 0, jump to VV
	LDI	#01	.. Set DF = #01
	SHR		
	GLO	RD	.. Subtract #1E ₁₆ from ..
	SMI	#1E	.. system pressure
	BDF	LP	.. If result > or = 0, jump to LP
	LDI	#FF	.. Set RA, RB = #FFFF ₁₆
	PLO	RA	.. Branch to VV as ..
	PHI	RA	.. default because system ..
	PLO	RB	.. pressure below table ..
	PHI	RB	.. range
	LBR	VV	..
LP:	GLO	RD	.. Subtract #46 ₁₆ from ..
	SDI	#46	.. System pressure
	EDF	MM	.. If result > or = 0, jump to MM
	LDI	#FF	.. Set RA, RB = #FFFF ₁₆
	PLO	RA	.. Branch to VV as ..
	PHI	RA	.. default because system ..
	PLO	RB	.. pressure above table ..

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	PHI	RR	.. range
	LBR	VV	..
MM:	GLO	RD	
	STR	R2	
	LDI	#01	.. Set DF = #01
	SHR		
	LDN	R9	.. Load first table pressure
	SD		.. Subtract from system pressure
	BNF	JJ	.. If result < = 0, jump to JJ
	RZ	JJ	..
	GLO	R9	
	STR	R2	.. Otherwise, increment table ..
	LDI	#2B	.. pointer to next pressure line
	ADD		..
	PLO	R9	..
	BR	MM	.. Jump to MM
JJ:	LDA	R9	.. Store upper pressure
	STXD		..
KK:	GHI	R8	.. Compare hi byte ..
	STR	R2	.. requested flow rate to
	LDN	R9	.. hi byte table flow rates
	SD		..
	BNF	RR	.. If result < or =0, jump to PR
	BZ	RR	..
	GLO	R7	
	STR	R2	.. Increment R7.0 by #04
	LDI	#04	..
	ADD		..
	PLO	#R7	..
	INC	R9	.. Move table pointer to ..
	INC	R9	.. next 2 bytes of data
	BR	KK	.. Jump to KK
RR:	GLO	R8	.. Now search for ..
	STR	R2	.. 2 byte table value ..
	INC	R9	.. > requested value
	LDI	#01	.. Set DF = #01
	SHR		..
	LDN	R9	
	SD		
	STXD		
	GHI	R8	.. Compare 2 byte ..
	STR	R2	.. requested flow rate to ..
	DEC	R9	.. 2 byte table flow rate ..
	LDN	R9	..
	SDB		..
	IRX		..
	BNF	RZ	.. If result < 0, jump to RZ
	BNZ	TT	.. If result = 0 jump to TT

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	LDX		
	BNZ	TT	
	BR	RZ	.. Otherwise, jump to RZ
TT:	GLO	R7	
	STR	R2	.. Increment R7.0 by #04
	LDI	#04	..
	ADD		..
	PLO	R7	..
	INC	R9	.. Move table pointer
	INC	R9	.. to next 2 bytes
	BR	RR	
RZ:	LDA	R9	
	STXD		.. Store 2 byte flow ..
	LDN	R9	.. rate as upper limit
	STXD		..
	GLO	R9	..
	STR	R2	..
	GLO	RD	.. Test to see if at ..
	SMI	#1E	.. lowest table pressure
	BZ	RY	.. If so, jump to RY
	LDI	#2B	.. Otherwise, decrement table ..
	SD		.. pointer to next lowest ..
	PLO	R9	.. pressure line
RY:	DEC	R9	
	LDA	R9	.. Begin data collection ..
	STXD		.. at same displacement
	LDN	R9	.. Store flow rate ..
	STXD		.. data of lower limit
	GLO	R7	..
	SHR		..
	STR	R2	..
	LDI	#01	.. Set DF = #01
	SHR		
	GLO	R9	
	SM		
	PLO	R9	
	DEC	R9	
	DEC	R9	
	LDN	R9	.. Store lower pressure
	STXD		..
	LDN	R9	
	ADI	#05	.. Add #05 to lower ..
	STR	R2	.. Pressure
	GLO	RD	
	SD		.. Subtract system pressure
	BDF	TL	.. If result < 0, jump to TL
	SEP	R4	

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	,A(HIPR)		.. Call high pressure interpolation subroutine
TL:	BR	QQ	
	SEP	R4	.. Call low pressure interpolation subroutine ..
QQ:	,A(LOPR)		..
	SEP	R4	
	,A(DISP)		.. Display resulting interpolated ..
	BR	QQ	.. flow rate at system pressure
	IRX		.. Go to bottom of stack
	IRX		..
	IRX		..
	IRX		..
	IRX		..
	IRX		..
	GLO	R8	
	STR	R2	
	LDI	#01	.. Set DF = #01
	SHR		
	GLO	RA	.. Subtract interpolated flow ..
	SD		.. rate from requested one
	STXD		..
	GHI	R8	..
	STR	R2	..
	GHI	RA	..
	SDB		..
	IRX		
	BNF	XZ	.. If result < 0, jump to XZ
	BNZ	ZG	.. If result > 0, jump to ZG
	LDX		..
	BNZ	ZG	..
	BR	XZ	.. Jump to XZ
ZG:	GLO	R9	
	STR	R2	
	GLO	RD	
	SMI	#1E	
	BZ	ZR	
	LDI	#2B	.. Move table pointer
	ADD		.. to upper pressure line
	PLO	R9	..
ZR:	LDA	R9	
	STXD		
	GLO	R7	.. Find location of next ..
	STR	R2	.. 2 byte flow rate ..
	LDI	#04	.. from R7.0 accumulator
	ADD		..
	PLO	R7	..

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	GLO	R7	
	SHR		
	STR	R2	
	GLO	R9	.. Move table pointer to ..
	ADD		.. next 2 byte flow rate
	PLO	R9	..
	BR	RZ	.. Jump to RZ
XZ:	GLO	R7	
	STR	R2	.. Have upper interpolated ..
	LDI	#04	.. flow rate, now move ..
	SD		.. back 2 data bytes to ..
	PLO	R7	.. get lower limit
	GLO	RD	..
	SMI	#1E	..
	LBZ	HH	..
	GLO	R9	..
	STR	R2	
	LDI	#28	.. Move table pointer ..
	ADD		.. to upper pressure
	PLO	R9	..
HH:	GHI	RA	
	PHI	RF	.. Store old interpolated ..
	GLO	RA	.. flow rate in RF
	PLO	RF	..
	LDA	R9	.. Begin process again ..
	STXD		.. at next lowest displacement
	GLO	R9	..
	STR	R2	
	GLO	R7	
	SHR		
	ADD		
	PLO	R9	
	LDA	R9	
	STXD		
	LDN	R9	
	STXD		
	GLO	RD	
	SMI	#1E	
	BZ	TR	
	GLO	R9	
	STR	R2	
	LDI	#2B	
	SD		
	PLO	R9	
TR:	DEC	R9	
	LDA	R9	
	STXD		
	LDN	R9	

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	STXD		
	GLO	R7	
	SHR		
	STR	R2	
	LDI	#01	.. Set DF = #01
	SHR		
	GLO	R9	
	SM		
	PLO	R9	
	DEC	R9	
	DEC	R9	
	LDN	R9	
	STXD		
	LDN	R9	
	ADI	#05	
	STR	R2	
	GLO	RD	
	SD		
	BDF	JN	
	SEP	R4	
	,A(HIPR)		.. Call hi pressure interpolation subroutine
JN:	BR	JP	
	SEP	R4	
	,A(LOPR)		.. Call lo pressure interpolation subroutine
JP:	IRX		.. Go to bottom of stack
	IRX		..
	IRX		..
	IRX		..
	IRX		..
	IRX		..
	GHI	RF	.. Store upper flow rate limit
	STXD		..
	GLO	RF	
	STXD		
	GLO	R7	.. Store upper displacement
	STR	R2	..
	LDI	#04	
	ADD		
	STXD		.. Store lower displacement
	GLO	R7	
	STXD		
	GHI	RA	.. Store lower flow rate limit
	STXD		..
	GLO	RA	..
	STXD		..
	SEP	R4	

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	GLO	RB	.. Test RB.0 to see if it is ..
	SHR		.. an odd or even number
	BNF	RJ	.. If even, jump to RJ
	GLO	RB	..
	STR	R2	.. If odd, add #01 to lo byte
	LDI	#01	..
	ADD		..
	PLO	RB	..
	GHI	RB	.. Add #00, with carry, to ..
	STR	R2	.. hi byte in case of ..
	LDI	#00	.. overflow
	ADC		..
	PHI	RB	..
RJ:	GHI	RB	.. Divide RB by 2
	SHR		..
	PHI	RB	..
	GLO	RB	..
	SHRC		..
	PLO	RB	..
	GLO	RB	.. Test RB.0 to see if it is ..
	SHR		.. an odd or even number
	BNF	RK	.. If even, jump to RK
	GLO	RB	..
	STR	R2	.. If odd, subtract #01 from ..
	LDI	#01	.. lo byte
	SD		..
	PLO	RB	..
	GHI	RB	.. Subtract #00, with borrow, ..
	STR	R2	.. from hi byte in case of ..
	LDI	#00	.. underflow
	SDB		..
	PHI	RB	..
RK:	GHI	RB	.. Divide RB by 2
	SHR		..
	PHI	RB	..
	GLO	RB	..
	SHRC		..
	PLO	RB	..
	GLO	RB	.. Test RB.0 to see if it is ..
	SHR		.. an odd or even number
	BNF	RL	.. If even, jump to RL
	GLO	RB	..
	STR	R2	.. If odd, add #01 to lo byte
	LDI	#01	..
	ADD		..
	PLO	RB	..
	GHI	RB	.. Add #00, with carry, to ..
	STR	R2	.. hi byte in case of ..

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	LDI	#00	.. overflow
	ADC		..
	PHI	RB	..
	GHI	RB	.. Divide RB by 2
	SHR		..
	PHI	RB	..
	GLO	RB	..
	SHRC		..
	PLO	RB	..
	GLO	RD	
	STR	R2	.. Subtract lower table pressure ..
	GLO	RE	.. from system pressure
	SD		..
	PLO	RC	.. Store result in RC.0
	LDI	#01	.. Set DF = #01
	SHR		..
RP:	GLO	RE	.. Get lower table pressure
	STR	R2	..
	GHI	RC	.. Add pressure increment
	ADD		..
	PLO	RE	..
	GLO	RA	.. Get lower flow rate
	STR	R2	..
	GLO	RB	.. Add flow rate increment
	ADD		..
	PLO	RA	..
	GHI	RA	..
	STR	R2	..
	GHI	RB	..
	ADC		..
	PHI	RA	..
	GLO	RD	
	STR	R2	.. Subtract modified lower table ..
	GLO	RE	.. pressure from system pressure
	SD		..
	PHI	R9	.. Store result in R9.1
	LDI	#01	.. Set DF = #01
	SHR		..
	GLO	RC	
	STR	R2	.. Subtract new pressure difference ..
	GHI	R9	.. from old pressure difference in RC.0
	SD		..
	BNF	ZZ	.. If result < 0, jump to ZZ
	BNF	ZZ	.. If result = 0, jump to ZZ

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	GHI	R9	.. Store new pressure difference ..
	PLO	RC	.. in RC.0
	BR	RP	.. Jump to RP
ZZ:	GLO	RA	
	STR	R2	.. Subtract flow rate increment increment ..
	GLO	RB	.. from RA and store result ..
	SD		.. back in RA
	PLO	RA	..
	GHI	RA	..
	STR	R2	..
	GHI	RB	..
	SDB		..
	PHI	RA	..
	LDI	#00	.. Set RB = #0000
	PLO	RB	..
	PHI	RB	..
	LDI	#08	.. Set R9.1 = #08
	PHI	R9	..
	SEP	R5	.. Return to calling program
HIPR:	IRX		*** High Pressure Subroutine ***
	IRX		
	IRX		
	LDXA		.. Load registers
	PLO	RE	.. RE.0 = lo pressure
	LDXA		
	PLO	RA	
	LDXA		
	PHI	RA	.. RA = lo flow rate
	LDXA		
	PLO	RB	
	LDXA		
	PHI	RB	.. RB = hi flow rate
	LDX		
	PHI	RE	.. RE.1 = hi pressure
	GLO	RE	.. Subtract lo pressure from ..
	SD		.. hi pressure
	SHR		.. Divide result by 8
	SHR		..
	SHR		..
	PHI	RC	.. Store pressure increment in RC.1
	LDI	#01	.. Set DF = #01
	SHR		..
	DEC	R2	.. Move stack to hi flow ..
	DEC	R2	.. rate, lo byte

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	GLO	RA	.. Subtract lo flow rate, ..
	SD		.. lo byte
	PLO	RA	.. Store result in RA.0
	IRX		.. Move stack to hi flow rate, ..
	GHI	RA	.. hi byte
	SDB		.. Subtract lo flow rate, lo byte
	PHI	RA	.. Store result in RA.1
	DEC	R2	.. Go to top of stack
	DEC	R2	..
	DEC	R2	..
	DEC	R2	..
	DEC	R2	..
	DEC	R2	..
	GLO	RA	.. Test RA.0 to see if it is ..
	SHR		.. an odd or even number
	BNF	RM	.. If even, jump to RM
	GLO	RA	
	STR	R2	.. If odd, add #01 to lo byte ..
	LDI	#01	..
	ADD		..
	PLO	RA	..
	GHI	RA	.. Add #00, with carry, to ..
	STR	R2	.. hi byte in case of
	LDI	#00	.. overflow
	ADC		..
	PHI	RA	..
RM:	GHI	RA	.. Divide RA by 2
	SHR		..
	PHI	RA	..
	GLO	RA	..
	SHRC		..
	PLO	RA	..
	GLO	RA	.. Test RA.0 to see if it is ..
	SHR		.. an odd or even number
	BNF	RN	.. If even, jump to RN
	GLO	RA	
	STR	R2	.. If odd, subtract #01 from ..
	LDI	#01	.. lo byte
	SD		..
	PLO	RA	..
	GHI	RA	..
	STR	R2	.. Subtract #00, with borrow, ..
	LDI	#00	.. from hi byte in case of ..
	SDB		.. underflow
	PHI	RA	..
RN:	GHI	RA	.. Divide RA by 2
	SHR		..

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	PHI	RA	..
	GLO	RA	..
	SHRC		..
	PLO	RA	..
	GLO	RA	.. Test RA.0 to see if it is ..
	SHR		.. an odd or even number
	BNF	RQ	.. If even, jump to RQ
	GLO	RA	..
	STR	R2	.. If odd, add #01 to lo byte
	LDI	#01	..
	ADD		..
	PLO	RA	..
	GHI	RA	.. Add #00, with cary, to ..
	STR	R2	.. hi byte in case of ..
	LDI	#00	.. overflow
	ADC		..
	PHI	RA	..
RQ:	GHI	RA	.. Divide RA by 2
	SHR		..
	PHI	RA	..
	GLO	RA	..
	SHRC		..
	PLO	RA	..
	GHI	RE	.. Get hi pressure
	STR	R2	..
	GLO	RD	.. Subtract system pressure ..
	SD		.. from hi table pressure
	PLO	RC	.. Store result in RC.0
	LDI	#01	.. Set DF = #01
	SHR		..
RG:	GHI	RE	.. Subtract pressure increment ..
	STR	R2	.. from hi pressure
	GHI	RC	..
	SD		..
	PHI	RE	.. Store result in RE.1
	GLO	RB	.. Subtract flow rate increment
	STR	R2	.. from hi flow rate
	GLO	RA	..
	SD		..
	PLO	RB	.. Store result in RB
	GHI	RB	..
	STR	R2	..
	GHI	RA	..
	SDB		..
	PHI	RB	..
	GHI	RE	..
	STR	R2	.. Subtract system pressure ..
	GLO	RD	.. from modified hi pressure

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	SD		
	PHI	R9	.. Store result in R9.1
	LDI	#01	.. Set DF = #01
	SHR		
	GLO	RC	.. Subtract old pressure difference in RC.0 ..
	STR	R2	.. from new pressure difference
	GHI	R9	..
	SD		..
	BNF	ZX	.. If result < 0, jump to ZX
	BZ	ZX	.. If result = 0, jump to ZX
	GHI	R9	.. Store new pressure ..
	PLO	RC	.. difference in RC.0
	BR	RG	.. Jump to RG
ZX:	GLO	RB	.. Add flow rate increment to ..
	STR	R2	.. RB
	GLO	RA	..
	ADD		..
	PLO	RA	..
	GHI	RB	..
	STR	R2	..
	GHI	RA	..
	ADC		..
	PHI	RA	..
	LDI	#00	.. Set RB = #0000
	PLO	RB	..
	PHI	RB	..
	LDI	#08	.. Set R9.1 = #08
	PHI	R9	
	SEP	R5	.. Return to calling program
VSD:	IRX		*** VALVE DISPLACEMENT SUBROUTINE ***
	IRX		
	IRX		
	LDXA		.. Load registers
	PLO	RA	.. RA = Lower flow rate
	LDXA		
	PHI	RA	
	LDXA		
	PLO	RE	.. RE.0 = lower valve stem displacement (VSD)
	LDXA		
	PHI	RE	.. RE.0 = Upper valve stem displacement
	LDXA		..
	PLO	RB	.. RB = Upper flow rate
	LDX		

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	PHI	RB	
	DEC	R2	.. Move stack to upper flow rate, lo byte
	LDI	#01	.. Set DF = #01
	SHR		
	GLO	RA	.. Subtract lower flow rate, lo byte
	SD		..
	PLO	RB	.. Store result in RB.0
	IRX		.. Move stack to upper flow rate, hi byte
	GHI	RA	.. Subtract lower flow rate, hi byte
	SDB		.. with borrow
	PHI	RB	.. Store result in RB.1
	GHI	RB	.. Divide by 2
	SHR		..
	PHI	RB	..
	GLO	RB	..
	SHRC		..
	PLO	RB	..
	GHI	RB	.. Divide RB by 2
	SHR		..
	PHI	RB	..
	GLO	RB	..
	SHRC		..
	PLO	RB	.. RB contains flow rate increment
	DEC	R2	.. Move stack to upper VSD
	DEC	R2	..
	GLO	RE	.. Subtract lower VSD
	SD		..
	SHR		.. Divide result by 4
	SHR		..
	PHI	RE	.. Store final result in RE.0
	DEC	R2	.. Go to top of stack
	DEC	R2	..
	DEC	R2	..
	DEC	R2	..
	DEC	R2	..
	DEC	R2	..
	GLO	R8	.. Get requested flow rate, lo byte
	STR	R2	
	GLO	RA	.. Subtract lower flow rate, lo byte
	SD		..
	PLO	RC	.. Store result in RC.0

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	GHI	R8	.. Get requested flow rate, hi byte
	STR	R2	..
	GHI	RA	.. Subtract, with borrow, lower flow rate, ..
	SDB		.. hi byte
	PHI	RC	.. Store result in RC.1
	LDI	#01	.. Set DF = #01
	SHR		
PP:	GLO	RA	..
	STR	R2	.. Add flow rate increment to ..
	GLO	RB	.. lower flow rate
	ADD		..
	PLO	RA	.. Store result in RA
	GHI	RA	..
	STR	R2	..
	GHI	RB	..
	ADC		..
	PHI	RA	..
	GLO	RE	.. Add VSD increment to ..
	STR	R2	.. lower VSD
	GHI	RE	..
	ADD		..
	PLO	RE	..
	GLO	R8	.. Subtract modified lower flow rate ..
	STR	R2	.. from requested flow rate
	GLO	RA	..
	SD		..
	PLO	R7	.. Store result in R7
	GHI	R8	..
	STR	R2	..
	GHI	RA	..
	SDB		..
	PHI	R7	..
	BDF	XH	.. If result > 0, jump to XH
	GLO	R7	.. Otherwise, take 2's compliment ..
	XRI	#FF	.. of result and store in R7
	ADI	#01	..
	PLO	R7	..
	GHI	R7	..
	XRI	#FF	..
	ADCI	#00	..
	PHI	R7	..
XH:	LDI	#01	.. Set DF = #01
	SHR		..
	GLO	RC	

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	STR	R2	.. Subtract new flow rate difference ..
	GLO	R7	.. from old flow rate difference
	SD		..
	STXD		..
	GHI	RC	..
	STR	R2	..
	GHI	R7	..
	SDB		..
	IRX		..
	BNF	GG	.. If result < 0, jump to GG
	BNZ	LJ	.. If result > 0, jump to LJ
	LDX		.. Move stack to 10 byte of result
	BNZ	LJ	.. If 10 byte > 0, jump to LJ
	BR	GG	.. Jump to GG
LJ:	GLO	R7	.. Store R7 value in RC
	PLO	RC	..
	GHI	R7	..
	PHI	RC	..
	BR	PP	.. Jump to PP
GG:	GLO	RE	.. Subtract VSD increment
	STR	R2	.. from modified VSD
	GHI	RE	..
	SD		..
	PLO	RA	.. Store result in RA
	LDI	#00	.. Set RB = #0000
	PLO	RB	..
	PHI	RB	..
	PHI	RA	.. Set RA.1 = #00
	LDI	#08	.. Set R9.1 = #08
	PHI	R9	..
	SEP	R5	.. Return to calling program
DISP:	LDI	A.1(DIGITS)	*** DISPLAY SUBROUTINE ***
	PHI	RF	.. Displays HEX. content of RA ..
	LDI	A.0(DIGITS)	.. and RB in LED display
	PLO	RF	.. Setup pointer
	GHI	RA	.. Digit 1
	SHR		
	SHR		
	SHR		
	SHR		
	STR	RF	
	INC	RF	.. Digit 2
	GHI	RA	
	ANI	#0F	
	STR	RF	

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	INC	RF	.. Digit 3
	GLO	RA	
	SHR		
	SHR		
	SHR		
	SHR		
	STR	RF	.. Digits 4
	INC	RF	
	GLO	RA	
	ANI	#OF	
	STR	RF	
	INC	RF	.. Digit 5
	GHI	RB	
	SHR		
	SHR		
	SHR		
	SHR		
	STR	RF	
	INC	RF	.. Digit 6
	GHI	RB	
	ANI	#OF	
	STR	RF	
	INC	RF	.. Digit 7
	GLO	RB	
	SHR		
	SHR		
	SHR		
	SHR		
	STR	RF	
	INC	RF	.. Digit 8
	GLO	RB	
	ANI	#OF	
	STR	RF	
	LDI	A.0(DIGITS)	
	PLO	RF	
	SEP	R4	
	,A(LEDD)		
	SEP	R5	
	DIGITS :	#8C00	.. Dummy save area
LEDD:	LDI	#80	*** LEDD SUBROUTINE ***
	PLO	RE	.. Refresh display
	LDI	A.1(DTAB)	.. Start with leftmost digit
	PHI	RD	.. Segment table pointer
LOOPD:	LDN	RF	.. Turn off?
	ANI	#80	
	BNZ	SKIPD	.. Yes, skip digit
	DEC	R2	.. Get 2 free RAM bytes

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	DEC	R2	
	GLO	RE	.. Ready for digit out
	STR	R2	
	LDN	RF	.. Tab
	ANI	#OF	
	ADI	A.0(DTAB)	
	PLO	RD	
	LDA	RD	.. Fetch segments
	PHI	RE	
	LDN	RF	
	ANI	#10	.. Decimal ?
	BZ	DISPD	.. No, show digit only
	GHI	RE	
	ANI	#BF	
	PHI	RE	.. Else add point
DISPD:	OUT4		.. Select digit
	GHI	RE	.. Put in segments
	STR	R2	
	OUT 3		.. Turn on
	SEP	R4	.. Delay
	A,(DELSUB)		
	SEX	R3	
	OUT 3	,#FF	.. Turn off segment
	SEX	R2	
SKIPD:	INC	RF	.. Bump pointer
	GLO	RE	.. Shift
	SHR		
	PLO	RE	
	BNZ	LOOPD	.. Continue
	DEC	RF	.. Else
	DEC	RF	
	DEC	RF	
	DEC	RF	
	DEC	RF	.. Fix pointer
	DEC	RF	
	DEC	RF	
	SEP	R5	.. And return
DELSUB:	LDI	#20	*** DELAY SUBROUTINE ***
	PLO	RD	
DL:	DEC	RD	.. Count done?
	GLO	RD	
	BNZ	DL	.. No, keep going
	SEP	R5	.. Return
ENTRY:	LDI	#00	*** Standard Call and Return Technique ***

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	PHI	R3	
	LDI	#05	.. Load return address into R3
	PLO	R3	
INIT:	LDI	A.1(CALL)	.. Load R4 with call address
	PHI	R4	..
	PHI	R5	.. Load R5 with return address
	LDI	A.0(CALL)	..
	PLO	R4	..
	LDI	A.0(RETSGM)	..
	PLO	R5	..
	LDI	#8C	
	PHI	R2	.. Load Stack address into R2
	PHI	R7	
	LDI	#1F	
	PLO	R2	
	PLO	RC	
	SEX	R2	.. Set stack pointer as R2
	SEP	R3	.. Return to main program
EXITA:	SEP	R3	.. Return to calling program
CALL:	SEX	R2	.. Point to stack
	GHI	R6	.. Save R6 value
	STXD		..
	GLO	R6	..
	STXD		..
	GHI	R3	.. Put R3 ..
	PHI	R6	.. Into R6
	GLO	R3	.. Both halves
	PLO	R6	
	LDA	R6	.. And put subroutine address ..
	PHI	R3	.. Into R3
	LDA	R6	
	PLO	R3	
	BR	EXITA	.. Go to calling subroutine
EXITR:	SEP	R3	.. Return to calling program
RETSGM:	GHI	R6	.. Put R6 back ..
	PHI	R3	.. Into R3
	GLO	R6	..
	PLO	R3	..

<u>LABEL</u>	<u>OBJECT CODE</u>	<u>OPERAND</u>	<u>COMMENTS</u>
	SEX	R2	.. Point to stack
	INC	R2	.. Recover ..
	LDXA		.. R6
	PLO	R6	.. Lower half
	LDX		
	PHI	R6	.. And upper half
	BR	EXITR	.. Go back to calling program
	END		

APPENDIX C

OVERALL PROGRAM

The Overall Program was the controlling algorithm for the developed assembly. The program listing is from the 1802 assembler at NASA Langley Research Center. To simulate the Standard Call and Return Technique (SCRT) of the actual 1802 microprocessor system, a separate program, entitled ML18, was supplied by Mr. Dave Akey of the Sperry Support Systems. This program was assembled with the Overall Program and is listed as lines 2-232 in the following pages.

ER LINE ADDR B1 02 03 04

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```

2
3
4
5
6
7
8 0000
9 0001
10 0002
11 0003
12 0004
13 0005
14 0007
15 0008
16 0009
17 000A
18 000B
19 000C
20 000D
21 000E
22 0001
23 0002
24 0004
25 0008
26 0010
27 0020
28 0040
29 0080
30 0100
31 0200
32 0400
33 0800
34 1000
35 2000
36 4000
37 8000

      NLIST M
      LIST X,A
      ..MACRO LIBRARY
      ..
      .. REGISTER NAME DEFINITIONS
      ..
      DMA=R0      ..DMA POINTER REGISTER
      INT=R1      ..INTERRUPT HANDLER PC
      SP=R2      ..STACK POINTER
      PC=R3      ..PROGRAM COUNTER
      CC=R4      ..CALL CODE PC
      RT=R5      ..RETURN CODE PC
      SI=R7      ..SCRATCH PAD POINTER
      AO=R8      ..ACCU 0 (MUST BE SAVED BY CALLED ROUTINE)
      AI=R9      ..ACCU 1
      A2=RA      ..ACCU 2
      TO=RA      ..TEMP 0(MUST BE SAVED BY CALLING ROUTINE)
      T1=RC      ..TEMP 1
      T2=RD      ..TEMP 2
      BP=RE      ..BREAK POINT REGISTER
      BIT0=1
      BIT1=2
      BIT2=4
      BIT3=8
      BIT4=16
      BIT5=32
      BIT6=64
      BIT7=128
      BIT8=256
      BIT9=512
      BIT10=1024
      BIT11=2048
      BIT12=4096
      BIT13=8192
      BIT14=16384
      BIT15=32768

```



```

39  ..
40  ..
41  ..
42  ..CALL SUBROUTINE CODE
43  ..THE LABEL CALLA--CALL ADDRESS MUST BE LOADED INTO
44  ..R4 AT SYSTEM INITIALIZATION TIME. THE OPCODE CALLC
45  ..MUST BE PLACED SOMEWHERE IN THE PROGRAM SPACE BUT
46  ..NOT WITHIN 15 WORDS OF THE END OF A PAGE.
47  ..
48  ..CALLC: MACRO
49  ..    LOCAL EXITA
50  ..    EXITA: SEP R3  ..R(3) IS POINTING TO THE FIRST INSTRUCTION IN SUBR,
51  ..    CALLA: SEX R2  ..POINT TO STACK
52  ..    GHI R6  ..PUSH R(6) ONTO STACK
53  ..    STXD   TO PREPARE IT FOR POINTING TO ARGUMENTS, AND
54  ..    GLO R6  ..DECREMENT TO FREE LOCATION.
55  ..    STXD
56  ..    GHI R3  ..COPY R(3) INTO R(6) TO SAVE THE RETURN ADDRESS
57  ..    PHI R6
58  ..    GLO R3
59  ..    PLO R6
60  ..    LDA R6  ..LOAD THE ADDRESS OF THE SUBROUTINE INTO R(3)
61  ..    PHI R3
62  ..    LDA R6
63  ..    PLO R3
64  BR EXITA..BRANCH TO ENTRY POINT OF "CALL" MINUS ONE BYTE
65  ..THIS LEAVES R(4) POINTING TO "CALL" ALLOWING FOLLOWING
66  ..REPEATED CALLS.
67  ENDM

```

EP LINE ADDR 01 02 03 04

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```

69      ..RETURN SUBROUTINE CODE
70      ..THE LABEL RETA--RETURN ADDRESS MUST BE LOADED INTO R5 AT
71      ..SYSTEM INITIALIZATION TIME.
72      ..THE OP CODE RETCOD MUST BE PLACED SOMEWHERE IN THE PROGRAM SPACE
73      ..BUT NOT WITHIN 12 WORDS OF THE END OF A PAGE.
74      ..
75      RETCOD:MACRO
76      LOCAL EXTR
77      EXTR: SEP R3 ..RETURN TO MAIN PROGRAM
78      RETA:  GHI R6 ..COPY R(6) INTO R(3)
79      PHI R3  ..R(3) CONTAINS THE RETURN ADDRESS
80      GLO R6
81      PLO R3
82      SEX R2
83      TNC R2
84      LDXA
85      PLO R6
86      LDX
87      PHI R6
88      BR EXTR..BRANCH TO ENTRY POINT OF "RETA" MINUS ONE BYTE
89      ..THIS LEAVES R(5) POINTING TO "RETA" FOR FOLLOWING REPEAT
90      ..CALLS.
91      ..
92      ENDM
93

```

ER LINE ADDR B1 B2 B3 B4

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```

95          PUSH:  MACRO P1,P2
96          SEX R2
97          GHI P1  ..PUSH P1 ON STACK AND SELECT P2 AS RX
98          STXD
99          GLO P1
100         STXD
101         IF P2+0
102         SEX P2
103         ENDF
104         ENDM
105
106         ..POP
107         POP:
108         MACRO P1,P2
109         SEX R2  ..POP P1 OFF STACK AND SELECT P2 AS RX
110         IRX
111         LDXA
112         PLO P1
113         LDX
114         PHI P1
115         IF P2+0
116         SEX P2
117         ENDF
118         ENDM
119
120         ..
121         ..
122         ..RETURN FROM SUBROUTINE MACRO
123         RETURN:MACRO
124         SEP R5
125         ENDM
126
127         ..
128         ..LOAD REGISTER IMMEDIATE MACRO
129         ..LOADS RN WITH VALUE P1
130         LDRI:  MACRO RN,P1
131         LDI A.1(P1)
132         PHI RN
133         LDI A.0(P1)
134         PLO RN
135         ENDM
136
137         ..
138         ..LOAD REGISTER VIA X AND ADVANCE X

```

EP LINE ADDR B1 B2 B3 B4

139
140
141
142
143
144
145
146LDRVXA MACRO RN
LDXA
PHI RN
LDXA
PLO RN
ENDM
.
.
.

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ER LINE ADDR B1 B2 B3 B4

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148
149
150
151
152
153

..
..
CALL1

THE CALL MACRO AUTOMATICALLY GENERATES THE CODE NECESSARY FOR
STANDARD CALL AND RETURN TECHNIQUE.

MACRO SUBR

SEP R4

ACON SUBR

ENDM

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ER LINE ADDR 01 02 03 04

```

155
156
157
158
159
160
161
162
163
164
165
166
167
168
169
170
171
172
173
174
175

..ONE PARAMETER CALL
CALL1: MACRO SUBR,P1
SEP R4
ACON SUBR
DC P1+0
ENDM

..TWO
CALL2: MACRO SUBR,P1,P2
CALL1 SUBR,P1
ACON P2+0
ENDM

..THREE
CALL3: MACRO SUBR,P1,P2,P3
CALL2 SUBR,P1,P2
ACON P3+0
ENDM

..FOUR
CALL4: MACRO SUBR,P1,P2,P3,P4
CALL3 SUBR,P1,P2,P3
ACON P4+0
ENDM

```

ER LINE ADDR 01 02 03 04

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```

177      LRVPI:  MACRO REG
178      LDA R6  ..LOAD REGISTER REG WITH
179      PHI REG ..PARAMETER SUPPLIED BY
180      LDA R6  ..THE CALLING ROUTINE
181      PLO REG
182      ENDM
183
184      ..
185      ..
186      LRVPI:  MACRO REG,REG2
187      LRVPI:  MACRO REG
188      LDA REG
189      PHI REG2
190      LDA REG ..BY THE CALLING ROUTINE SUPPLIED PARAMETER
191      PLO REG2..REG IS USED AS AN INTERMEDIATE REG.
192      ENDM

```

FR LINE ADDR 81 82 83 84

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```

194      ..STORE REGISTER VIA REGISTER AND DECREMENT
195      ..
196      ..RS.0 -> M(RD) RS.1 -> M(RD+1) RD -> RD+1
197      ..
198      SRVRD: MACRO RS,RD
199      GLO RS
200      STR RD
201      DEC RD
202      GHI RS
203      STR RD
204      ENDM
205
206      ..STORE REGISTER VIA REGISTER AND INCREMENT
207      ..
208      ..RS.1 -> M(RD) RS.0 -> M(RD-1) RD -> RD-1
209      ..
210      SRVRA: MACRO RS,RD
211      GHI RS
212      STR RD
213      INC RD
214      GLO RS
215      STR RD
216      ENDM
217
218      ..STORE REG VIA REG AND INCREMENT PAST
219      SRVRAA: MACRO RS,RD
220      SRVRA RS,RD
221      INC RD
222      ENDM
223
224      ..LOAD REG VIA REG AND DECREMENT PAST
225      LRVRD: MACRO RS,RD
226      LDH RD
227      PLO RS
228      DEC RD
229      LDH RD
230      PHI RS
231      DEC RD
232      ENDM
233      LIST S
234      DIS
235      ,00H
236      LDRI R2,RANTOP
237      LDRI R3,MAIN
238      LDRI R4,CALLA
239
240      ..SETUP STACK
241      ..SETUP PROGRAM START ADDRESS
242      ..SCRT CALL ADDRESS

```

```

243      0000 71
244      0001 00
245      0002
246      0008
247      000F

```


ER LINE ADDR B1 B2 B3 B4

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```

35R 0082 32 8C      BZ DP
359 0084 72      LDXA
360 0095 AC      PLO RC
361 0086 F0      LDX RC
362 0097 BC      PHI RC
363 0098 FA 01    LDI 01H
364 009A AB      PLO RB
365 0088 38      SKP
366 008C 60      IRX
367 009D 9C      GHI RC
368 009E F3      XOR
369 009F 32 9A    BZ GQ
370 0091 F0      LDX
371 0092 AC      PHI RC
372 0093 22      DEC R2
373 0094 F0      LDX
374 0095 AC      PLO RC
375 0096 FA 01    LDI 01H
376 0098 AB      PLO RB
377 0099 60      IRX
378 009A AB      INP 3
379 009B BD      PHI RD
380 009C FA 10    LDI 010H
381 009E AF      PHI RF
382 009F          CALL1 OUTPUT,04H
386 00A3 9D      GHI RD
387 00A4 FF 0F    SMI 0FH
388 00A6 23 AB    RDF DD
389 00A8 FF 0F    LDI 0FH
390 00AA DD      PHI RD
391 00AB 9D      GHI RD
392 00AC 52      STR R2
393 00AD FD 2F    SOI 02FH
394 00AF 33 B5    BDF EE
395 00B1 FA 2F    LDI 02FH
396 00B3 DD      PHI RD
397 00B4 52      STR R2
398 00B5 DD      GLO RD
399 00B6 F3      XOR
400 00B7 32 CC    RZ PZ
401 00B9 AD      GLO RD
402 00BA F5      SD
403 00BB 33 C1    BDF RW
404 00BD FB FF    XRI 0FFH
405 00BF FC 01    ADI 01H

DP:
GQ:
DD:
EE:

```

..FOR CHANGE. IF SAME, JUMP TO CHECK HI BYTES
 ..IF DIFFERENT, LOAD NEW VALUES INTO RC
 ..VALUES IN RC HOLDING AREA
 ..SET RB.0 = #01 (SET FLAG)
 ..COMPARE HI BYTE TO PREVIOUS HI BYTE TO CHECK..
 ..FOR CHANGE
 ..IF SAME, JUMP TO GQ
 ..IF DIFFERENT, LOAD NEW VALUES INTO RC
 ..SET RB.0 = #01 (SET FLAG)
 ..INPUT UPSTREAM PRESSURE
 ..STORE IN RD.1
 ..STOP MANOMETER INTERFACE CIRCUIT
 ..COMPARE INPUT PRESSURE TO TABLE LD VALUE
 ..COMPARE INPUT PRESSURE TO TABLE HI VALUE
 ..COMPARE PRESS. TO PREVIOUS PRESS. READING
 ..IF SAME, JUMP TO PZ
 ..GET OLD SYSTEM PRESSURE
 ..SUBT. OLD SYSTEM PRESS. FROM CURRENT READING

ER LINE ADDR B1 B2 B3 B4

1802 ASSEMBLER VER 1.0MR

```

406 00C1 52          STR R2
407 00C2 F8 01      LDI 01H
408 00C4 F5          SD
409 00C5 38 CC      BNF PZ
410 00C7 9D          GHI RD
411 00CF AD          PLO RD
412 00C9 F8 01      LDI 01H
413 00C8 B8          PHI R8
414 00CC 8R          GLO RR
415 00CD 52          STR R2
416 00CE 9R          GHI R8
417 00CF F1          OR
418 00D0 3A D7      RNZ PY
419 00D2          CALL LVDT
422 00D5 30 6D      RR RX
423 00D7 RA          GLO RA
424 00D8 73          STXD
425 00D9 9C          GHI RC
426 00DA 73          STXD
427 00DB 8C          GLO RC
428 00DC 73          STXD
429 00DD AD          GLO RD
430 00DF 73          STXD
431 00FF FF          GLO RE
432 00E0 73          STXD
433 00E1          CALL TABRD
436 00F4 F8 10      LDI 010H
437 00F6 BF          PHI RF
438 00E7 F8 00      LDI 00H
439 00F9 B9          PHI R9
440 00EA AF          PLO RF
441 00EB 60          IRY
442 00EC 72          LDXA
443 00ED AE          PLO RE
444 00EE 72          LDXA
445 00EF AD          PLO RD
446 00F0 72          LDXA
447 00F1 AC          PLO RC
448 00F2 72          LDXA
449 00F3 BC          PHI RC
450 00F4 F0          LDX
451 00F5 BA          PHI RA
452 00F6 9A          GHI RA
453 00F7 52          STR R2
454 00F8 8A          GLO RA

RV:
406 00C1 52          STR R2
407 00C2 F8 01      LDI 01H
408 00C4 F5          SD
409 00C5 38 CC      BNF PZ
410 00C7 9D          GHI RD
411 00CF AD          PLO RD
412 00C9 F8 01      LDI 01H
413 00C8 B8          PHI R8
414 00CC 8R          GLO RR
415 00CD 52          STR R2
416 00CE 9R          GHI R8
417 00CF F1          OR
418 00D0 3A D7      RNZ PY
419 00D2          CALL LVDT
422 00D5 30 6D      RR RX
423 00D7 RA          GLO RA
424 00D8 73          STXD
425 00D9 9C          GHI RC
426 00DA 73          STXD
427 00DB 8C          GLO RC
428 00DC 73          STXD
429 00DD AD          GLO RD
430 00DF 73          STXD
431 00FF FF          GLO RE
432 00E0 73          STXD
433 00E1          CALL TABRD
436 00F4 F8 10      LDI 010H
437 00F6 BF          PHI RF
438 00E7 F8 00      LDI 00H
439 00F9 B9          PHI R9
440 00EA AF          PLO RF
441 00EB 60          IRY
442 00EC 72          LDXA
443 00ED AE          PLO RE
444 00EE 72          LDXA
445 00EF AD          PLO RD
446 00F0 72          LDXA
447 00F1 AC          PLO RC
448 00F2 72          LDXA
449 00F3 BC          PHI RC
450 00F4 F0          LDX
451 00F5 BA          PHI RA
452 00F6 9A          GHI RA
453 00F7 52          STR R2
454 00F8 8A          GLO RA

PY:
423 00D7 RA          GLO RA
424 00D8 73          STXD
425 00D9 9C          GHI RC
426 00DA 73          STXD
427 00DB 8C          GLO RC
428 00DC 73          STXD
429 00DD AD          GLO RD
430 00DF 73          STXD
431 00FF FF          GLO RE
432 00E0 73          STXD
433 00E1          CALL TABRD
436 00F4 F8 10      LDI 010H
437 00F6 BF          PHI RF
438 00E7 F8 00      LDI 00H
439 00F9 B9          PHI R9
440 00EA AF          PLO RF
441 00EB 60          IRY
442 00EC 72          LDXA
443 00ED AE          PLO RE
444 00EE 72          LDXA
445 00EF AD          PLO RD
446 00F0 72          LDXA
447 00F1 AC          PLO RC
448 00F2 72          LDXA
449 00F3 BC          PHI RC
450 00F4 F0          LDX
451 00F5 BA          PHI RA
452 00F6 9A          GHI RA
453 00F7 52          STR R2
454 00F8 8A          GLO RA

PZ:
412 00C9 F8 01      LDI 01H
413 00C8 B8          PHI R8
414 00CC 8R          GLO RR
415 00CD 52          STR R2
416 00CE 9R          GHI R8
417 00CF F1          OR
418 00D0 3A D7      RNZ PY
419 00D2          CALL LVDT
422 00D5 30 6D      RR RX
423 00D7 RA          GLO RA
424 00D8 73          STXD
425 00D9 9C          GHI RC
426 00DA 73          STXD
427 00DB 8C          GLO RC
428 00DC 73          STXD
429 00DD AD          GLO RD
430 00DF 73          STXD
431 00FF FF          GLO RE
432 00E0 73          STXD
433 00E1          CALL TABRD
436 00F4 F8 10      LDI 010H
437 00F6 BF          PHI RF
438 00E7 F8 00      LDI 00H
439 00F9 B9          PHI R9
440 00EA AF          PLO RF
441 00EB 60          IRY
442 00EC 72          LDXA
443 00ED AE          PLO RE
444 00EE 72          LDXA
445 00EF AD          PLO RD
446 00F0 72          LDXA
447 00F1 AC          PLO RC
448 00F2 72          LDXA
449 00F3 BC          PHI RC
450 00F4 F0          LDX
451 00F5 BA          PHI RA
452 00F6 9A          GHI RA
453 00F7 52          STR R2
454 00F8 8A          GLO RA

..STORE DIFFERENCE ON STACK
..SUBTRACT #01 FROM DIFFERENCE
..GET CURRENT SYSTEM PRESSURE READING
..LOAD NEW PRESS. INTO R0.0 IF DIFFERENT
..SET R8.1 = #01 (SET FLAG)
..CHECK FOR FLAGS IN R8

..IF A FLAG IS SET, JUMP TO START VALVE MOVEMENT
..OTHER WISE, CHECK VALVE FOR CREEP OR SLIPPAGE
..LOOP BACK TO RX FOR DATA INPUT
..STORE INPUT VALUES ON STACK
..RA.0= CALCULATED VSD IN STEPS
..RC.1= HI BYTE OF REQUESTED FLOW RATE
..
..RC.0= LO BYTE OF REQUESTED FLOW RATE
..
..RD.0= UPSTREAM PRESSURE
..
..RE.0= READING FROM VSD SENSOR
..
..CALL TABLE READ SUBROUTINE TO CALCULATE..
..NECESSARY DISPLACEMENT
..
..PUT #00 INTO R9.1,R0.0

..INCREMENT STACK BACK ONE POSITION
..RELOAD STORED VALUES ON STACK BACK INTO..
..THEIR RESPECTIVE REGISTERS

..BUT THIS TIME, PUT STORED OLD VSD IN RA.1

..PUT OLD VSD ON STACK

```

ER LINE ADDR B1 B2 B3 B4

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```

455 00F9 F5
456 00FA C2 01 27
457 00FD
461 0101
465 0105 9A
466 0106 A9
467 0107 32 0C
468 0109
471 010C F8 18
472 010E RF
473 010F
477 0113
481 0117 RA
482 0118 40
483 0119 FR 10
484 011B RF
485 011C
488 011F
492 0123
496 0127
499 0128 FR 00
500 012C AH
501 012D B8
502 012F C0 00 6D
503 0200
504 0200
506 0201
511 0207 FR 00
512 0209 A7
513 020A B7
514 020B AA
515 020C BA
516 020D AB
517 020E RR
518 020F 8F
519 0210 52
520 0211 98
521 0212 F1
522 0213 C2 03 38
523 0216 8D
524 0217 52
525 021A FR 01
526 021A F6
527 021B 09
528 021C F5

SD
LBZ JB
CALL1 OUTPUT,044H
CALL1 OUTPUT,04H
GHI RA
PLO R9
RZ JA
CALL CLOSE
LDI 18H
PHI RF
CALL1 OUTPUT,014H
CALL1 OUTPUT,04H
GLN RA
PLJ R9
LDI 010H
PHI RF
CALL OPEN
CALL1 OUTPUT,004H
CALL1 OUTPUT,04H
CALL LVDT
LDI 004
PLN R8
PHI R8
LRR RX
PAGE
RET8: RETURN
TARRD: LDRI R9,FLODAT
LDI 00H
FLN R7
PHI R7
PLN RA
PHI RA
PLO R8
PHI R8
GLO R8
STR R2
GHI R8
OR
LBZ YY
GLO RD
STR R2
LDI 01H
SHR
LDH R9
SD

JAI:
SD
LBZ JB
CALL1 OUTPUT,044H
CALL1 OUTPUT,04H
GHI RA
PLO R9
RZ JA
CALL CLOSE
LDI 18H
PHI RF
CALL1 OUTPUT,014H
CALL1 OUTPUT,04H
GLN RA
PLJ R9
LDI 010H
PHI RF
CALL OPEN
CALL1 OUTPUT,004H
CALL1 OUTPUT,04H
CALL LVDT
LDI 004
PLN R8
PHI R8
LRR RX
PAGE
RET8: RETURN
TARRD: LDRI R9,FLODAT
LDI 00H
FLN R7
PHI R7
PLN RA
PHI RA
PLO R8
PHI R8
GLO R8
STR R2
GHI R8
OR
LBZ YY
GLO RD
STR R2
LDI 01H
SHR
LDH R9
SD

JBI:
SD
LBZ JB
CALL1 OUTPUT,044H
CALL1 OUTPUT,04H
GHI RA
PLO R9
RZ JA
CALL CLOSE
LDI 18H
PHI RF
CALL1 OUTPUT,014H
CALL1 OUTPUT,04H
GLN RA
PLJ R9
LDI 010H
PHI RF
CALL OPEN
CALL1 OUTPUT,004H
CALL1 OUTPUT,04H
CALL LVDT
LDI 004
PLN R8
PHI R8
LRR RX
PAGE
RET8: RETURN
TARRD: LDRI R9,FLODAT
LDI 00H
FLN R7
PHI R7
PLN RA
PHI RA
PLO R8
PHI R8
GLO R8
STR R2
GHI R8
OR
LBZ YY
GLO RD
STR R2
LDI 01H
SHR
LDH R9
SD

MM:
SD
LBZ JB
CALL1 OUTPUT,044H
CALL1 OUTPUT,04H
GHI RA
PLO R9
RZ JA
CALL CLOSE
LDI 18H
PHI RF
CALL1 OUTPUT,014H
CALL1 OUTPUT,04H
GLN RA
PLJ R9
LDI 010H
PHI RF
CALL OPEN
CALL1 OUTPUT,004H
CALL1 OUTPUT,04H
CALL LVDT
LDI 004
PLN R8
PHI R8
LRR RX
PAGE
RET8: RETURN
TARRD: LDRI R9,FLODAT
LDI 00H
FLN R7
PHI R7
PLN RA
PHI RA
PLO R8
PHI R8
GLO R8
STR R2
GHI R8
OR
LBZ YY
GLO RD
STR R2
LDI 01H
SHR
LDH R9
SD

***TABLE READ SUBROUTINE***
..THIS COLLECTS THE DATA FOR INTERPOLATION SUBR.
..LOAD R9 WITH LOCATION OF FLOW RATE TABLE
..PUT #00 INTO R7,RA,RB
..
..
..
..
..PUT REQUESTED FLOW RATE ON STACK
..COMPARE HI BYTE TO LO BYTE TO SEE IF INPUT..
..REQUESTED = #00. IF SO, JUMP TO YY
..
..PUT UPSTREAM PRESSURE ON STACK
..SET DF= #01
..
..GET PRESS. VALUE OF FIRST DATA LINE IN TABLE
..SUBTRACT TABLE VALUE FROM UPSTREAM READING

```

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FR LINE ADDR B1 B2 B3 B4

```

529 021D 3B 29      BNF JJ
530 021F 32 29      RZ JJ
531 0221 A9          GLO R9
532 0222 52         STR R2
533 0223 F8 2B      LOI 02BH
534 0225 F4         ADD
535 0226 A9          PLO R9
536 0227 30 16      BR HM
537 0229 49          LNA R9
538 022A 73         STXD
539 022B 98         GHI R8
540 022C 52         STR R2
541 022D 09         LDN R9
542 022E F3         SD
543 022F 3B 49      BNF RR
544 0231 32 49      BZ RR
545 0233 87         GLC R7
546 0234 52         STR R2
547 0235 F8 04      LDY 04H
548 0237 F4         ADD
549 0238 A7         PLO R7
550 0239 87         GLO R7
551 023A 52         STR R2
552 023B F8 50      YRI 050H
553 023D 3A 45      RHZ KL
554 023F F8 50      LDY 050H
555 0241 AA         PLO RA
556 0242 60         IRX
557 0243 30 00      RR PET8
558 0245 19         INC R9
559 0246 19         INC R9
560 0247 30 2B      BR KK
561 0249 88         GLO R8
562 024A 52         STR R2
563 024B 19         INC R9
564 024C F8 01      LOI 01H
565 024E F6         SHR
566 024F 09         LDN R9
567 0250 F5         SD
568 0251 73         STXD
569 0252 98         GHI R8
570 0253 52         STR R2
571 0254 29         DEC R9
572 0255 09         LDN R9
573 0256 75         SDB

      JJ:
      KK:

      KL:
      RR:

      ..IF DIFFERENCE < 0, JUMP TO JJ
      ..IF DIFFERENCE = 0, JUMP TO JJ

      ..IF DIFFERENCE > 0, GET R9.0(MEMORY ADDRESS)..
      ..AND ADD #28HEX. TO IT TO JUMP THE POINTER..
      ..TO THE NEXT PRESSURE LINE IN THE TABLE

      ..PUT TABLE PRESS. ON STACK FOR C.L.C. PURPOSES
      ..THIS IS HIGH PRESS. IN INTERPOLATION ROUTINE

      ..PUT HI BYTE OF REQUESTED FLOW RATE ON STACK
      ..GET DATA IN MEMORY SPACE DEFINED BY R9
      ..SUBTRACT STORED DATA FROM HI BYTE ON STACK
      ..IF DIFFERENCE < 0, JUMP TO RR
      ..IF DIFFERENCE = 0, JUMP TO RR

      ..PUT R7.0 VALUE ON STACK

      ..ADD #04 TO R7.0 VALUE..
      ..AND PUT NEW VALUE BACK IN R7.0
      ..R7 IS USED AS AN ACCUMULATOR TO CALC..
      ..# STEPS THE VALVE STEM NEEDS TO BE DISPL.
      ..SEE IF THE # OF STEPS IN R7 IS = TO MAXIMUM..
      ..THE STEM CAN TRAVEL. IF NOT, THEN JUMP TO KL
      ..IF R7 DOES = #50HEX., THEN PUT #50HEX. INTO..
      ..RA.0

      ..RETURN TO MAIN PROGRAM AS A DEFAULT
      ..IF R7 NOT AT MAX.,GO TO NEXT 2 MEMORY BYTES..
      ..POINTED TO BY R9
      ..JUMP TO KK AND START OVER AGAIN

      ..PUT LO BYTE OF REQUESTED FLOW RATE ON STACK
      ..INCREMENT TABLE POINTER TO TABLE LO BYTE
      ..SET DF = #01

      ..GET TABLE LO BYTE
      ..SUBTRACT TABLE LO BYTE FROM REQUESTED LO BYTE
      ..PUT DIFFERENCE ON STACK AND DECREMENT STACK

      ..COMPARE TABLE HI BYTE TO REQUESTED HI BYTE
      ..
      ..SUBTRACT TABLE HI BYTE FROM REQUESTED HI BYTE

```

1002 ASSEMBLER VER 1.0MR

EP LINE ADDR R1 R2 R3 R4

```

574 0257 60      IRX
575 0258 38 72    BNF RZ
576 025A 3A 5F    BNZ TT
577 025C F0      LDX
578 025D 32 72    BZ RZ
579 025F 87      GLO R7
580 0260 FC 04    ADI 04H
581 0262 A7      PLO R7
582 0263 52      STR R2
583 0264 FR 50    XPI 050H
584 0266 3A 6E    RNZ KZ
585 0268 FR 50    LDI 050H
586 026A AA      PLO RA
587 026B 60      IRX
588 026C 30 00    RP RET8
589 026F 19      INC R9
590 026F 19      INC R9
591 0270 30 49    BR RR
592 0272 49      LDA R9
593 0273 73      STXD
594 0274 09      LDN R9
595 0275 73      STXD
596 0276 A9      GLO R9
597 0277 52      STR R2
598 0278 AD      GLO RD
599 0279 FF OF    SHI OFH
600 027B 32 R1    BZ RY
601 027D FR 28    LDI 02BH
602 027F F5      SN
603 0280 A9      PLO R9
604 0281 29      DEC R9
605 0282 49      LDA R9
606 0283 73      STXD
607 0284 09      LDN R9
608 0285 73      STXD
609 0286 87      GLO R7
610 0287 F6      SHR
611 028A 52      STR R2
612 028Q FR 01    LDI 01H
613 028B F6      SHR
614 028C 89      GLO R9
615 028D F7      SN
616 028E A9      PLO R9
617 028F 29      DEC R9
618 0290 29      DEC R9

574 0257 60      IRX
575 0258 38 72    BNF RZ
576 025A 3A 5F    BNZ TT
577 025C F0      LDX
578 025D 32 72    BZ RZ
579 025F 87      GLO R7
580 0260 FC 04    ADI 04H
581 0262 A7      PLO R7
582 0263 52      STR R2
583 0264 FR 50    XPI 050H
584 0266 3A 6E    RNZ KZ
585 0268 FR 50    LDI 050H
586 026A AA      PLO RA
587 026B 60      IRX
588 026C 30 00    RP RET8
589 026F 19      INC R9
590 026F 19      INC R9
591 0270 30 49    BR RR
592 0272 49      LDA R9
593 0273 73      STXD
594 0274 09      LDN R9
595 0275 73      STXD
596 0276 A9      GLO R9
597 0277 52      STR R2
598 0278 AD      GLO RD
599 0279 FF OF    SHI OFH
600 027B 32 R1    BZ RY
601 027D FR 28    LDI 02BH
602 027F F5      SN
603 0280 A9      PLO R9
604 0281 29      DEC R9
605 0282 49      LDA R9
606 0283 73      STXD
607 0284 09      LDN R9
608 0285 73      STXD
609 0286 87      GLO R7
610 0287 F6      SHR
611 028A 52      STR R2
612 028Q FR 01    LDI 01H
613 028B F6      SHR
614 028C 89      GLO R9
615 028D F7      SN
616 028E A9      PLO R9
617 028F 29      DEC R9
618 0290 29      DEC R9

TT:
KZ:
RZ:
RY:

```

```

..IF DIFFERENCE < 0, JUMP TO RZ
..IF DIFFERENCE > 0, JUMP TO TT

```

```

..ADD #04 TO R7.0
..PUT NEW VALUE BACK IN R7.0

```

```

..SEE IF THE # OF STEPS IN R7 IS = TO MAXIMUM..
..THE STEM CAN TRAVEL. IF NOT, THEN JUMP TO KZ
..IF R7 DOES = #50HEX., THEN PUT #50HEX. INTO..
..RA.0

```

```

..RETURN TO MAIN PROGRAM AS A DEFAULT
..GO TO THE NEXT 2 MEMORY BYTES DEFINED BY R9

```

```

..IF NUMBERS IN MEMORY MEET ABOVE TESTS, STORE..
..VALUES ON STACK AND DECREMENT STACK

```

```

..SUBTRACT #0FHEX. FROM SYSTEM PRESS. TO SEE..
..IF THEY EQUAL. THIS IS TO PREVENT THE TABLE..
..FROM UNDERFLOWING
..SUBTRACT #2BHEX. FROM R9.0
..THIS WILL JUMP THE TABLE POINTER DOWN..
..TO THE NEXT LOWER PRESSURE RANGE
..STORE FLOW RATE DATA OF LOWER PRESS.
..
..
..
..DIVIDE R7.0 BY 2 TO FIND TRUE VSD
..

```

```

..SET DF = #01

```

```

..SUBTRACT VSD FROM R9.0 TO GET TO LOWER..
..PRESSURE VALUE
..
..

```

ER LINE ADDR R1 R2 R3 R4

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```

619 0291 09 LDN R9
620 0292 73 STXD
621 0293 CALL INTER
624 0296 60 IRX
625 0297 60 IRX
626 0298 60 IRX
627 0299 60 IRX
628 029A 60 IRX
629 029B 60 IRX
630 029C 8F GLN R8
631 029D 52 STR R2
632 029F F8 01 LDI 01H
633 02A0 F6 SHR
634 02A1 AA GLD RA
635 02A2 F5 SD
636 02A3 73 STXD
637 02A4 98 GHI R8
638 02A5 F2 STR R2
639 02A6 9A GHI RA
640 02A7 75 SOB
641 02A8 60 IRX
642 02A9 3R D4 RNF XZ
643 02AB 3A R0 BNZ ZG
644 02AD F0 LDY
645 02AF 32 D4 RZ XZ
646 02B0 89 GLN R9
647 02B1 52 <TR R2
648 02B2 RD GLN R0
649 02B3 FF 0F SHI 0FH
650 02B5 32 8B RZ ZR
651 02B7 F2 2H LDI 02BH
652 02B9 F4 ADD
653 02BA A9 PLJ R9
654 02BB 49 LDA R9
655 02BC 73 STXD
656 02BD R7 GLD R7
657 02BE FC 04 ADI 04H
658 02C0 A7 PLD R7
659 02C1 F2 STR R2
660 02C2 FB 50 XPI 050H
661 02C4 3A CC BNZ KX
662 02C6 FA 50 LDI 05CH
663 02C8 AA PLD RA
664 02C9 60 IRX
665 02CA 30 00 BR RET0

                                ZG:
645 02AF 32 D4
647 02B1 52
648 02B2 RD
649 02B3 FF 0F
650 02B5 32 8B
651 02B7 F2 2H
652 02B9 F4
653 02BA A9
654 02BB 49
655 02BC 73
656 02BD R7
657 02BE FC 04
658 02C0 A7
659 02C1 F2
660 02C2 FB 50
661 02C4 3A CC
662 02C6 FA 50
663 02C8 AA
664 02C9 60
665 02CA 30 00

                                ZR:
653 02BA A9
654 02BB 49
655 02BC 73
656 02BD R7
657 02BE FC 04
658 02C0 A7
659 02C1 F2
660 02C2 FB 50
661 02C4 3A CC
662 02C6 FA 50
663 02C8 AA
664 02C9 60
665 02CA 30 00

..STORE LOWER PRESSURE
..
..CALL INTERPOLATION SUBROUTINE
..INCREMENT TO BOTTOM OF STACK
..
..
..
..
..RA HOLDS INTERPOLATED FLOW RATE
..GET LO BYTE OF REQUESTED FLOW RATE
..SET DF = #01
..
..GET LO BYTE OF INTERPOLATED FLOW RATE
..SUBTRACT INTERPOLATED VALUE FROM REQUESTED
..
..GET HI BYTE OF REQUESTED FLOW RATE AND STORE
..
..GET HI BYTE OF INTERPOLATED FLOW RATE
..SUBTRACT INTERPOLATED VALUE FROM REQUESTED..
..WITH ROROV
..
..COMPARE HI AND LO DIFFERENCE AS TEST
..IF DIFFERENCE > 0, WE HAVEN'T UPPER LIMIT..
..ALONG SYSTEM PRESSURE LINE
..
..AGAIN, SUBTRACT #0FHEX. FROM SYSTEM PRESS...
..TO PREVENT TABLE UNDERFLOW
..JUMP BACK TO UPPER PRESS. LINE
..STORE UPPER PRESSURE
..
..ADD #04 TO R7.0 ACCUMULATOR
..
..SEE IF THE # OF STEPS IN R7 IS - TO MAXIMUM..
..THE STEPM CAN TRAVEL. IF NLT, THEN JUMP TO KX
..IF R7 DOES = #50HEX., THEN PUT #50HEX. INTO..
..RA.0
..RETURN TO MAIN PROGRAM AS A DEFAULT

```


EP LINE ADDR B1 B2 B3 B4

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666 02CC 87	KX:	GLO R7	..DIVIDE R7.0 BY 2
667 02CD F6		SHR	..STORE RESULT ON STACK
668 02CE 52		STR R2	
669 02CF 89		GLO R9	
670 02D0 F4		ADD	..ADD THIS TO R9.0 TO GET POINTER TO NEXT..
671 02D1 A9		PLO R9	..2 BYTE FLOW RATE DATA
672 02D2 30 72	XZ:	BR RZ	..BEGIN DATA GATHERING AGAIN AT RZ
673 02D4 A7		GLO R7	..IF DIFFERENCE < 0 OR = 0, WE HAVE UPPER LIMIT..
674 02D5 52		STR R2	..ALONG SYSTEM PRESSURE LINE
675 02D6 F8 04		LDI 04H	..SUBTRACT #04 FROM ACCUMULATOR
676 02D8 F5		SD	..
677 02D9 A7		PLO R7	..
678 02DA AD		GLO RD	..AGAIN, SUBTRACT #0FHEX. FROM SYSTEM PRESS....
679 02DB FF 0F		SMI 0FH	..TO PREVENT TABLE UNDERFLOW
680 02DD 32 E5		BZ HH	
681 02DE 89		GLO R9	
682 02E0 52		STR R2	..JUMP BACK TO UPPER PRESS. LINE
683 02F1 F8 2B		LDI 02BH	
684 02E3 F4		ADD	
685 02E4 A9		PLO R9	..STORE NEW INTERPOLATED VALUE IN RF
686 02E5 9A	HH:	GHI RA	..
687 02F6 BF		PHI RF	..
688 02E7 8A		GLO RA	..
689 02E8 AF		PLO RF	..BEGIN BY STORING UPPER PRESS.
690 02E9 49		LDA R9	
691 02FA 73		STXD	
692 02EA 89		GLO R9	..DIVIDE R7.0 BY 2
693 02EC 52		STR R2	..
694 02ED 87		GLO R7	..MOVE TABLE POINTER BACK TO 2 BYTE..
695 02EE F6		SHR	..TABLE FLOW RATE ENCOUNTERED ON PREVIOUS..
696 02EF F4		ADD	..INTERPOLATION
697 02F0 A9		PLO R9	..STORE FLOW RATE DATA AGAIN
698 02F1 49		LDA R9	
699 02F2 73		STXD	
700 02F3 09		LDN R9	
701 02F4 73		STXD	
702 02F5 AD		GLO RD	..SUBTRACT #0FHEX. FROM SYSTEM PRESS. TO..
703 02F6 FF 0F 01		SMI 0FH	..PREVENT TABLE UNDERFLOW
704 02F7 C2 03 01		LBZ TR	..IF SYSTEM PRESS. = LOWEST TABLE VALUE, JUMP..
705 02F8 A9		GLO R9	..TO TR
706 02FC 52		STR R2	..JUMP BACK TO SAME POSITION IN LOWER PRESS....
707 02FD FA 2B		LDI 02BH	..LIMIT
708 02FF F5		SD	..STORE FLOW RATE DATA AGAIN
709 0300 A9	TR:	PLO R9	..
710 0301 29		DEC R9	

ER LINE ADDR B1 B2 B3 B4

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```

711 0302 49 LDA R9
712 0303 73 STXD
713 0304 09 LDN R9
714 0305 73 STXD
715 0306 A7 GLO R7
716 0307 F6 SHR
717 030A 52 STR R2
718 0309 F8 01 LDI 01H
719 0308 FA SHR
720 030C 89 GLO R9
721 030D F7 SM
722 030E A9 PLO R9
723 030F 29 DEC R9
724 0310 29 DEC R9
725 0311 09 LDN R9
726 0312 73 STXD
727 0313 CALL INTER
730 0316 60 IPX
731 0317 60 IRX
732 031A 60 IRX
733 0319 60 IRX
734 031A 60 IRX
735 031A 60 IRX
736 031C 9F GHI RF
737 031D 73 STXD
738 031E 8F GLO RF
739 031F 73 STXD
740 0320 87 GLO R7
741 0321 52 STR R2
742 0322 F8 04 LDI 04H
743 0324 F4 ADD
744 0325 73 STXD
745 0326 87 GLO R7
746 0327 73 STXD
747 0328 9A GHI RA
748 0329 73 STXD
749 032A 8A GLO RA
750 0329 73 STXD
751 032C CALL VALDI
754 032F 60 IRX
755 0330 60 IRX
756 0331 60 IRX
757 0332 60 IRX
758 0333 60 IRX
759 0334 60 IRX

..
..
..
..
..SET DF = 601

..STORE LOWER PRESS. VALUE
..CALL INTERPOLATION SUBROUTINE
..INCREMENT TO BOTTOM OF STACK
..
..
..
..
..STORE PREVIOUS INTERP. FLOW RATE ON STACK
..
..
..ADD #04 TO CURRENT VSD VALUE
..THIS VALUE CORRESPONDS TO VSD FOR OLD..
..INTERPOLATED FLOW RATE
..STORE OLD ACCUMULATED VSD
..STORE CURRENT VSD
..STORE CURRENT INTERPOLATED FLOW RATE
..
..
..CALL VALVE DISPLACEMENT SUBROUTINE
..INCREMENT TO BOTTOM OF STACK
..
..
..
..

```

ER LINE ADDR R1 B2 B3 B4

1202 ASSEMBLER VER 1.0MP

```

760 0135 C0 02 00      LRR RETR
761 011R FR 00          LDI 00H
762 033A AA             PLD FA
763 033R C0 02 00      LRR PETA
764 0400                PAGE
765 0400                PFTURN
767 0401 P0             CLOSE: GLD P9
768 0402 FC 02          ADI 02H
769 0404 A9             PLO R0
770 0405                CALL1 OUTPUT,03H
774 0409                CALL1 OUTPUT,02H
776 0400                CALL COUNT
781 0410 3R 05          PNF YV
782 0412 P9             CLD P9
783 0413 FF 02          SPT 02H
784 0415 A9             PLO P9
785 0416 FR 00          LDI 00H
786 041R AR             PLO RB
787 0419 RR             PHI RR
788 041A                CALL1 OUTPUT,0AH
792 041E 30 00          P9 XY
793 0420                PFTURN
795 0421 R9             OPEN:  GLD P9
796 0422 32 31          R7 AA
797 0424                CALL1 OUTPUT,01H
801 042R                CALL1 OUTPUT,00H
805 042C                CALL COUNT
808 042F 3R 24          RNF P0
809 0431 FR 00          LDI 00H
810 0432 AB             PLO RR
811 0434 RP             PHI R9
812 0435 30 20          PR PS
813 0437                PFTURN
815 043R 1R             PET4:  PFTURN
816 0439 F0             COUNT: INC RR
817 043A 73             GLD R9
818 043R 60             STXD
819 043C PR             IPY
820 043D F2             CLD RR
821 043F 3A 45          XOP
822 0440 FP 01          PNF PR
823 0442 FA             LDI 01H
824 0443 30 37          SHP
825 0445 FR 00          PP PET4
826 0447 FA             PR:  LDI 00H
                        SHP

```

..RETURN TO CALLING PROGRAM
 ..PUT #00 IN RA.0 AS DEFAULT DISPLACEMENT
 ..RETURN TO CALLING PROGRAM
 ..**CLOSE VALVE SURROUTINE**
 ..ADD #02 TO VSD IN R9.0
 ..
 ..
 ..OUTPUT #03 ON DATA BUS
 ..OUTPUT #02 ON DATA BUS
 ..CALL COUNTDOWN SURROUTINE
 ..JUMP TO XV IF COUNTDOWN NOT COMPLETE
 ..SUBTRACT #02 FROM MODIFIED VSD IN R9.0
 ..
 ..
 ..RELOAD #0000 INTO RR
 ..
 ..OUTPUT #0AHEX. ON DATA BUS
 ..RETURN TO CALLING PROGRAM
 ..**OPEN VALVE SURROUTINE**
 ..IF VSD IS = 0, JUMP TO AA
 ..
 ..OUTPUT #01 ON DATA BUS
 ..OUTPUT #00 ON DATA BUS
 ..CALL COUNTDOWN SURROUTINE
 ..JUMP TO P0 IF COUNTDOWN NOT COMPLETE
 ..PESTORE #0000 TO RB
 ..
 ..RETURN TO CALLING PROGRAM
 ..**COUNTDOWN SURROUTINE**
 ..INCFMENT PP
 ..
 ..COMPARE R9.0 TO PR.0
 ..IF PR <> R9, JUMP TO PR
 ..SET DF = #01
 ..
 ..RETURN TO CALLING PROGRAM
 ..SET DF = #00

```

1602 ASSEMBLER VER 1.0MR

ER LINE ADDR B1 B2 B3 B4

827 0448 30 37      BR RET4
828 044A          RET2: RETURN
829 044B          OUTPUT: LDA R6
830 044C          STXD
831 044D          IRX
832 044E          OUT 5
833 044F          DEC R2
834 0450          DELAY: GHI RF
835 0451          STXD
836 0452          GLO RF
837 0453          STXD
838 0454          XX: DEC RF
839 0455          GLO RF
840 0456          BNZ XX
841 0457          GHI RF
842 0458          BZ XX
843 0459          DEC RF
844 045A          RR XX
845 045B          IRX
846 045C          XA: LDYA
847 045D          PLO RF
848 045E          LDX
849 045F          PHI RF
850 0460          BR RET2
851 0461          PAGE
852 0462          RETS: RETURN
853 0463          INTER: IRX
854 0464          IRX
855 0465          LDXA
856 0466          PLO RE
857 0467          LDXA
858 0468          PLO RA
859 0469          PHI RA
860 0470          LDXA
861 0471          PLO RB
862 0472          LDXA
863 0473          PHI RB
864 0474          LDX
865 0475          PHI RE
866 0476          GLO RE
867 0477          SD
868 0478          SHR
869 0479          SHR
870 0480          SD
871 0481          F5
872 0482          F6
873 0483          F6

      **RETURN TO CALLING PROGRAM
      ***OUTPUT TO DATA LINES SUBROUTINE***
      **STORE ON STACK, DATA TO BE OUTPUT
      **OUTPUT DATA
      **DELAY SUB-SUBROUTINE**
      **STORE VALUE OF RF ON STACK
      **COUNTDOWN RF UNTIL IT = #0000
      **RELOAD ORIGINAL NUMBER INTO RF
      **RETURN TO CALLING PROGRAM
      ***INTERPOLATION SUBROUTINE***
      **MOVE STACK POINTER PAST ADDRESS LOCATIONS..
      **STORED THERE BY SCRT
      **LOAD STORED DATA INTO REGISTERS
      **PUT LOW PRESS. INTO RE.0
      **PUT LO BYTE OF LOW FLOW RATE IN RA.0
      **PUT HI BYTE OF LOW FLOW RATE IN RA.1
      **PUT LO BYTE OF HIGH FLOW RATE IN RB.0
      **PUT HI BYTE OF HIGH FLOW RATE IN RB.1
      **PUT HIGH PRESS. IN RE.1
      **SUBTRACT LOW PRESS. FROM HIGH PRESS.
      **DIVIDE DIFFERENCE BY 8
      **

```

ER LINF ADDR 81 B2 83 84

1902 ASSEMBLER VER 1.0MR

```

-874 0514 F6
875 0515 BC
876 0516 F8 01
877 051A F6
878 0519 22
879 051A 22
880 051A RA
881 051C F5
882 051D AB
883 051E 60
884 051F 9A
885 0520 75
886 0521 88
887 0522 22
888 0523 22
889 0524 22
890 0525 22
891 0526 22
892 0527 22
893 052A 22
894 0529 F8 00
895 052B RD
896 052C RA
897 052D F6
898 052E 38 38
899 0530 88
900 0531 FC 01
901 0533 AB
902 0534 9B
903 0535 7C 00
904 0537 BA
905 0538 F8 01
906 053A RD
907 053B 9B
908 053C F6
909 053D 8A
910 053F RA
911 053F 76
912 0540 AB
913 0541 RA
914 0542 F6
915 0543 3A 50
916 0545 BA
917 0546 FF 01
918 054A AB

RJI:
SHR RA
PHI RC
LDI 01H
SHR
DEC R2
DEC R2
GLD RA
SD
PLD RB
IRX RA
GHI RA
SDB
PHI RB
DEC R2
DEC R2
DEC R2
DEC R2
DEC R2
DEC R2
LDI 00H
PHI RD
GLO RB
SHR RJ
BNF RJ
GLO RB
ADI 01H
PLD RA
GHI RB
ANDI 00H
PHI RB
LDI 01H
PHI RD
GHI RB
SHR
PHI RB
GLD RA
SHRC
PLD RB
GLO RB
SHR
BNF RK
GLO RA
SHI 01H
PLD RB

..
..PUT PRESS, INCREMENT IN RC.1
..SET DF = #01
..DECREMENT STACK TO HIGH FLOW RATE, LO BYTE
..SUBTRACT FROM THIS, LOW FLOW RATE, LO BYTE
..PUT DIFFERENCE IN RB.0
..MOVE STACK POINTER TO HI FLOW RATE, HI BYTE
..SUBTRACT LO FLOW RATE, HI BYTE WITH BORROW
..PUT DIFFERENCE IN RB.1
..MOVE TO TOP OF STACK
..
..
..
..
..SET VALUE OF RD.1 = #00
..
..GET LO BYTE OF DIFFERENCE IN FLOW RATES..
..AND TEST TO SEE IF LEAST SIGNIFICANT BIT..
..(LSB) = 0. IF SO, JUMP TO RJ
..
..ADD #01 TO LO BYTE OF DIFFERENCE
..THIS WILL HELP COMPENSATE FOR DIVISION..
..TRUNCATION. THIS METHOD REDUCES CUMULATIVE..
..ERROR BY ALTERNATELY ADDING AND SUBTRACTING..
..#01 TO THE FLOW RATE DIFFERENCE
..SET RD.1 AS A CHECK FOR ADDITION OR SUBTR.
..
..DIVIDE HI BYTE OF DIFFERENCE BY 2
..
..DIVIDE LO BYTE OF DIFFERENCE BY 2 WITH CARRY
..
..
..TEST TO SEE IF NEW DIFFERENCE IS AN EVEN # ..
..IF IT IS, JUMP TO RK
..SUBTRACT #01 FROM LO BYTE OF DIFFERENCE
..

```

ER LINE ADDR B1 02 B3 B4

1802 ASSEMBLER VER 1.0MR

910	0549	0R	GHI RBSUBTRACT #00 FROM HI BYTE OF DIFF. WITH BORROW
920	054A	7F	SMI 00HSET RD.1 AS A CHECK FOR ADDITION OR SUBTR.
921	054C	8R	PHI RBDIVIDE HI BYTE OF NEW DIFFERENCE BY 2
922	054D	FA	LDI 00H
923	054F	8D	PHI RDDIVIDE LO BYTE OF NEW DIFFERENCE BY 2
924	0550	9R	GHI RB
925	0551	F6	SHR
926	0552	8R	PHI RB
927	0553	8B	GLO RB
928	0554	76	SHRC
929	0555	AB	PLO RB
930	0556	8B	GLO RB
931	0557	F6	SHR
932	0558	3B	BNF RLIF NEW DIFF. IS AN EVEN #, JUMP TO RL
933	055A	0D	GHI RDIF RD.1 <> 0, JUMP TO PT
934	055B	3A	BNZ PT
935	055D	8R	GLO RBADD #01 TO LO BYTE OF DIFFERENCE
936	055E	FC	ADI 01H
937	0560	AR	PLO RB
938	0561	9R	GHI RBADD #00 TO HI BYTE OF DIFF. WITH CARRY
939	0562	7C	ADCI 00H
940	0564	8R	PHI RBJUMP TO RL
941	0565	3D	RR RLTHIS SECTION IS TO ALTERNATE THE ADDITION..
942	0567	8R	GLO RBAND SUBTRACTION IN CASE THE LAST TEST..
943	056A	FF	SMI 01HSHOWED AN EVEN NUMBER IN THE LSB. BY DOING..
944	056A	AB	PLO RBTHIS, THE ADDITION OF #01 AGAIN, BEFORE..
945	056B	9B	GHI RB#01 IS SUBTRACTED, IS AVOIDED.
946	056C	7F	SMI 00H
947	056E	8B	PHI RBDIVIDE HI BYTE OF NEW DIFFERENCE BY 2
948	056F	9B	GHI RB
949	0570	F6	SHR
950	0571	8R	PHI RBDIVIDE LO BYTE OF NEW DIFF. BY 2 WITH CARRY
951	0572	8R	GLO RB
952	0573	76	SHRC
953	0574	AB	PLO RB
954	0575	8D	GLO RD
955	0576	52	STR R2SUBTRACT LOWER PRESS. FROM SYSTEM PRESS.
956	0577	8E	GLO RE
957	0578	F5	SDIF RESULT > 0, JUMP TO ZB
958	0579	33	RDF ZBTAKE THE 2'S COMPLIMENT OF RESULT
959	057B	FB	XPI OFFH
960	057D	FC	ADI 01HSTORE RESULT
961	057F	AC	PLO RCSET DF = #01
962	0580	FB	LDI 01H	..	
963	0582	F6	SHR	..	

PR LINE ADDR B1 B2 B3 B4

1802 ASSEMBLER VER 1.0MR

964 0583 8E	GLO RE	..ADD PRESS. INCREMENT TO LOWER PRESS.
965 0584 52	STR R2	..
966 0585 0C	GHI RC	
967 0586 F4	ADD	
968 0587 AF	PLN RE	
969 0588 RA	GLO RA	
970 0589 52	STR R2	..ADD LO FLOW RATE INCREMENT TO LOW FLOW..
971 058A 88	GLO R8	..RATE,LO BYTE
972 058B F4	ADD	
973 058C 0A	PLN RA	
974 058D 0A	GHI RA	
975 058E 57	STR R2	
976 058F 08	GHI R8	
977 0590 74	ADC	..ADD HI FLOW RATE INCREMENT TO LOW FLOW..
978 0591 BA	PHI RA	..RATE, HI BYTE
979 0592 8D	GLO RD	
980 0593 52	STR R2	
981 0594 AE	GLO RE	
982 0595 F5	SD	..SUBTR. MODIFIED LOWER PRESS.FROM SYSTEM PRESS.
983 0596 33 9C	RDF ZC	..IF RESULT IS > OR = 0, JUMP TO ZC
984 0598 F0 FF	YRI OFFH	..TAKE THE 2'S COMPLIMENT OF RESULT
985 059A FC 01	ADI 01H	
986 059C B0	PHI R9	..PUT RESULT IN R9.1
987 059D F8 01	LOI 01H	..SET DF = #01
988 059F F6	SHR	
989 05A0 8C	GLO RC	
990 05A1 52	STR R2	
991 05A2 99	GHI R9	
992 05A3 F5	SD	..SUBTRACT NEW DIFF. IN PRESS. FROM OLD DIFF.
993 05A4 38 AC	BNF ZZ	..IF RESULT IS < OR = 0, JUMP TO ZZ
994 05A6 37 AC	BZ ZZ	..
995 05A8 90	GHI R9	
996 05A9 AC	PLN RC	..STORE NEW DIFF.IN PRESS.IN OLD DIFF.HOLDING..
997 05AA 30 R3	RR RP	..AREA. JUMP TO RP
998 05AC 8A	GLO RA	
999 05AD 52	STR R2	
1000 05AE 88	GLO RB	
1001 05AF F5	SD	..SUBTRACT FLOW RATE INCREMENT FROM LATEST ..
1002 05B0 AA	PLN RA	..MODIFIED LO FLOW RATE
1003 05B1 9A	GHI RA	..
1004 05B2 52	STR R2	..
1005 05B3 98	GHI R8	..
1006 05B4 75	SD8	..RESULTING INTERPOLATED FLOW RATE STORED IN RA
1007 05B5 BA	PHI RA	
1008 05B6 F8 00	LOI 00H	

ER LINE ADDR B1 B2 B3 B4

1802 ASSEMBLER VER 1.0MR

```

1009 0388 AB      PLO RB
1010 0589 RR      PHI RB
1011 058A F8 08   LDI A,1(FLODAT)
1012 059C B9      PHI R9
1013 058D 30 00   RR RET5
1014 0600         PAGE
1015 0600         RET3: RETURN
1017 0601 60      VALDI: IRX
1018 0602 60      IRX
1019 0603 60      IRX
1020 0604 72      LDXA
1021 0605 AA      PLO RA
1022 0606 72      LDXA
1023 0607 9A      PHI RA
1024 0608 72      LDXA
1025 0609 AE      PLO RE
1026 060A 72      LDXA
1027 060B BE      PHI RE
1028 060C 72      LDXA
1029 060D A4      PLO RB
1030 060F F0      LDX
1031 060F RB      PHI RB
1032 0610 22      DEC R2
1033 0611 F8 01   LDI 01H
1034 0613 F6      SHR
1035 0614 8A      GLO RA
1036 0615 F5      SN
1037 0616 AB      PLO RB
1038 0617 60      IRX
1039 0618 9A      GHI RA
1040 0619 75      SOB
1041 061A RR      PHI RB
1042 061A RR      GLO RB
1043 061C F6      SHR
1044 061D 38 27   BNF AJ
1045 061F RB      GLO RB
1046 0620 FC 01   ADI 01H
1047 0622 AB      PLO RB
1048 0623 9A      GHI RB
1049 0624 7C 00   ADCI 00H
1050 0626 88      PHI RB
1051 0627 9A      GHI RB
1052 0628 F6      SHR
1053 0629 RR      PHI RB
1054 062A 88      GLO RB

1009 0388 AB      ..SET RB = #0000
1010 0589 RR      ..RELOAD ADDRESS OF FLOW RATE TABLE BACK..
1011 058A F8 08   ..INTO R9.1
1012 059C B9      ..RETURN TO CALLING PROGRAM
1013 058D 30 00   ..
1014 0600         **VALVE DISPLACEMENT SUBROUTINES**
1015 0600         **INTERPOLATES THE CORRECT VALVE DISPLACEMENT
1017 0601 60      **LOAD NECESSARY DATA INTO REGISTER*
1018 0602 60      **PUT LOWER FLOW RATE, LO BYTE, IN RA.0
1019 0603 60      **PUT LOWER FLOW RATE, HI BYTE, IN RA.1
1020 0604 72      **PUT LOWER VSD IN RE.0
1021 0605 AA      **PUT UPPER VSD IN RE.1
1022 0606 72      **PUT UPPER FLOW RATE, LO BYTE, IN RB.0
1023 0607 9A      **PUT UPPER FLOW RATE, HI BYTE, IN RB.1
1024 0608 72      **MOVE STK.POINTER TO LO BYTE OF UPPER FLOW RATE
1025 0609 AE      **SET OF = #01
1026 060A 72      ..
1027 060B BE      **SUBTR. FROM THIS THE LOWER FLOW RATE, LO BYTE
1028 060C 72      **MOVE STK.POINTER TO HI BYTE OF UPPER FLOW RATE
1029 060D A4      **SUBTR. FROM THIS THE LOWER FLOW RATE, HI BYTE
1030 060F F0      **CHECK THE LSB IN RB FOR #00
1031 060F RB      **IF LSB IN RB = #00, JUMP TO AJ
1032 0610 22      **OTHERWISE, ADD #01 TO RB TO PREVENT..
1033 0611 F8 01   **DIVISION TRUNCATION.
1034 0613 F6      ..
1035 0614 8A      ..
1036 0615 F5      ..
1037 0616 AB      ..
1038 0617 60      ..
1039 0618 9A      ..
1040 0619 75      ..
1041 061A RR      ..
1042 061A RR      ..
1043 061C F6      ..
1044 061D 38 27   ..
1045 061F RB      ..
1046 0620 FC 01   ..
1047 0622 AB      ..
1048 0623 9A      ..
1049 0624 7C 00   ..
1050 0626 88      ..
1051 0627 9A      ..
1052 0628 F6      ..
1053 0629 RR      ..
1054 062A 88      ..

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AJ:

FW LIST ADDR B1 B2 B3 B4

100- ASSEMBLER VER 1.0MR

1055 0628 76	SHRC	..DIVIDE FLOW RATE DIFF., LO BYTE, BY 2 W/ CARRY
1056 062C AR	PL0 RB	
1057 062D AR	GLO RB	
1058 062F F6	SHR	
1059 062F 39	BNF AK	..IF LSB IN RB = #00, JUMP TO AK
1060 0631 PR	GLO RB	
1061 0632 FF 01	SMT 01H	..OTHERWISE, SUBTRACT #01 FROM RB TO PREVENT..
1062 0634 AP	PL0 RB	..DIVISION TRUNCATION.
1063 0635 9R	GHI RB	..
1064 0636 7F 00	SMBI 00H	..
1065 0638 8B	PHI RB	..
1066 0639 9B	GHI RB	..DIVIDE HI BYTE OF FLOW RATE DIFFERENCE BY 2
1067 063A F6	SHR	..
1068 063B BR	PHI RB	
1069 063C PR	GLO RB	
1070 063D 76	SHRC	..DIVIDE LO BYTE OF FLOW RATE DIFF.BY 2 W/ CARRY
1071 063F AB	PL0 RB	..MOVE STACK POINTER TO UPPER VSD
1072 063F 22	DEC R2	..
1073 0640 22	DEC R2	
1074 0641 RE	GLO RE	
1075 0642 F5	SD	..SUBTRACT FROM THIS THE LOWER VSD
1076 0643 F6	SHR	..DIVIDE DIFFERENCE BY 4
1077 0644 F6	SHR	..
1078 0645 RE	PHI RE	..GO TO TOP OF STACK
1079 0646 22	DEC R2	..
1080 0647 22	DEC R2	..
1081 0648 22	DEC R2	..
1082 0649 22	DEC R2	..
1083 064A 22	DEC R2	..
1084 064B 22	DEC R2	..GET LO BYTE OF REQUESTED FLOW RATE
1085 064C AB	GLO RB	..SUBTR. FROM THIS THE LOWER FLOW RATE, LO BYTE
1086 064D 52	STR R2	..
1087 064F PA	GLO RA	..STORE RESULT IN RC.0
1088 064F F5	SD	..GET HI BYTE OF REQUESTED FLOW RATE
1089 0650 AC	PL0 RC	..
1090 0651 9B	GHI RB	..STORE RESULT IN RC.0
1091 0652 52	STR R2	..GET HI BYTE OF REQUESTED FLOW RATE
1092 0653 0A	GHI RA	..
1093 0654 75	SDB	..SUBTR. FROM THIS THE LOWER FLOW RATE, HI BYTE
1094 0655 BC	PHI RC	..STORE RESULT IN RC.1
1095 0656 F8 01	LDI 01H	..SET OF = #01
1096 0658 F6	SHR	
1097 0659 RA	GLO RA	
1098 065A 52	STR R2	
1099 065B 8B	GLO RB	..ADD LOWER FLOW RATE INCREMENT TO LOWER..

AK:

PP:

1802 ASSEMBLER VER 1.0MR

ER LINE ADDR B1 B2 B3 B4

1100	065C	F4	ADD RA
1101	065D	AA	PL0 RA
1102	065E	QA	GHI RA
1103	065F	52	STR R2
1104	0660	98	GHI R8
1105	0661	74	ADC
1106	0662	RA	PHI RA
1107	0663	8E	GLO RE
1108	0664	52	STR R2
1109	0665	9E	GHI RE
1110	0666	F4	ADD
1111	0667	AE	PL0 RE
1112	0668	8B	GLO R8
1113	0669	52	STR R2
1114	066A	8A	GLO RA
1115	066B	F5	SD
1116	066C	A7	PL0 R7
1117	066D	98	CHI R3
1118	066E	52	STR R2
1119	066F	9A	GHI RA
1120	0670	75	SDB
1121	0671	A7	PHI R7
1122	0672	33 80	BDF XH
1123	0674	A7	GLO R7
1124	0675	F8 FF	XRI OFFH
1125	0677	FC 01	AND OIH
1126	0679	A7	PL0 R7
1127	067A	97	GHI R7
1128	067B	F8 FF	XRI OFFH
1129	067D	7C 00	ANDI OOH
1130	067F	B7	PHI R7
1131	0680	F8 01	LDI OIH
1132	0692	F6	SHR
1133	0693	AC	GLO RC
1134	06A4	52	STR R2
1135	06A5	87	GLC R7
1136	06A6	F5	SD
1137	06A7	73	STXD
1138	068A	9C	GHI RC
1139	0699	52	STR R2
1140	069A	97	GHI R7
1141	06A8	75	SDB
1142	06A9	60	TRX
1143	06AD	3R 9A	BKF GG
1144	06AF	3A 94	BNZ LJ

..FLOW RATE, LO BYTE
 ..
 ..ADD UPPER FLOW RATE INCREMENT TO LOWER..
 ..FLOW RATE, HI BYTE
 ..
 ..ADD VSD INCREMENT TO LOWER VSD
 ..
 ..SUBTRACT LO BYTE OF MODIFIED LOWER FLOW..
 ..RATE FROM REQUESTED LO BYTE
 ..
 ..SUBTRACT HI BYTE OF MODIFIED LOWER FLOW..
 ..RATE FROM REQUESTED HI BYTE, WITH BORROW
 ..
 ..IF DIFFERENCE > 0, JUMP TO XH
 ..TAKE THE 2'S COMPLIMENT OF RESULT LO BYTE
 ..
 ..TAKE THE 2'S COMPLIMENT OF RESULT HI BYTE
 ..
 ..SET DF = #01
 ..
 ..SUBTRACT NEW DIFF. FROM OLD DIFF., LO BYTE
 ..
 ..SUBTRACT NEW DIFF. FROM OLD DIFF., HI BYTE..
 ..WITH BORROW
 ..
 ..IF DIFFERENCE < 0, JUMP TO GG
 ..IF DIFFERENCE > 0, JUMP TO LJ

ER LINE ADDR B1 R2 B3 B4

1802 ASSEMBLER VER 1.0MR

```

1145 0501 F0
1146 0632 32 9A
1147 0694 87
1148 0695 AC
1149 0696 97
1150 0697 BC
1151 0698 30 59
1152 0699 8F
1153 069B 52
1154 069C 9F
1155 069D F5
1156 069F AA
1157 069F F8 00
1158 06A1 8A
1159 06A2 8B
1160 06A3 8A
1161 06A4 FF 08
1162 06A6 89
1163 06A7 30 00
1164 0700
1165 0700
1167 0701 6F
1168 0702 A7
1169 0703 8F
1170 0704 F3
1171 0705 32 00
1172 0707 8A
1173 0708 8B
1174 0709 F8 01
1175 0709 F6
1176 070C F8 09
1177 070E 8A
1178 070F 08
1179 0710 52
1180 0711 AE
1181 0712 87
1182 0713 F5
1183 0714 33 14
1184 0716 F8 FF
1185 0718 FC 01
1186 071A 52
1187 071B F6 03
1188 071D F5
1189 071F 38 00
1190 0720 32 00

LJ:
1145 0501 F0
1146 0632 32 9A
1147 0694 87
1148 0695 AC
1149 0696 97
1150 0697 BC
1151 0698 30 59
1152 0699 8F
1153 069B 52
1154 069C 9F
1155 069D F5
1156 069F AA
1157 069F F8 00
1158 06A1 8A
1159 06A2 8B
1160 06A3 8A
1161 06A4 FF 08
1162 06A6 89
1163 06A7 30 00
1164 0700
1165 0700
1167 0701 6F
1168 0702 A7
1169 0703 8F
1170 0704 F3
1171 0705 32 00
1172 0707 8A
1173 0708 8B
1174 0709 F8 01
1175 0709 F6
1176 070C F8 09
1177 070E 8A
1178 070F 08
1179 0710 52
1180 0711 AE
1181 0712 87
1182 0713 F5
1183 0714 33 14
1184 0716 F8 FF
1185 0718 FC 01
1186 071A 52
1187 071B F6 03
1188 071D F5
1189 071F 38 00
1190 0720 32 00

GG:
1145 0501 F0
1146 0632 32 9A
1147 0694 87
1148 0695 AC
1149 0696 97
1150 0697 BC
1151 0698 30 59
1152 0699 8F
1153 069B 52
1154 069C 9F
1155 069D F5
1156 069F AA
1157 069F F8 00
1158 06A1 8A
1159 06A2 8B
1160 06A3 8A
1161 06A4 FF 08
1162 06A6 89
1163 06A7 30 00
1164 0700
1165 0700
1167 0701 6F
1168 0702 A7
1169 0703 8F
1170 0704 F3
1171 0705 32 00
1172 0707 8A
1173 0708 8B
1174 0709 F8 01
1175 0709 F6
1176 070C F8 09
1177 070E 8A
1178 070F 08
1179 0710 52
1180 0711 AE
1181 0712 87
1182 0713 F5
1183 0714 33 14
1184 0716 F8 FF
1185 0718 FC 01
1186 071A 52
1187 071B F6 03
1188 071D F5
1189 071F 38 00
1190 0720 32 00

RET7:
1145 0501 F0
1146 0632 32 9A
1147 0694 87
1148 0695 AC
1149 0696 97
1150 0697 BC
1151 0698 30 59
1152 0699 8F
1153 069B 52
1154 069C 9F
1155 069D F5
1156 069F AA
1157 069F F8 00
1158 06A1 8A
1159 06A2 8B
1160 06A3 8A
1161 06A4 FF 08
1162 06A6 89
1163 06A7 30 00
1164 0700
1165 0700
1167 0701 6F
1168 0702 A7
1169 0703 8F
1170 0704 F3
1171 0705 32 00
1172 0707 8A
1173 0708 8B
1174 0709 F8 01
1175 0709 F6
1176 070C F8 09
1177 070E 8A
1178 070F 08
1179 0710 52
1180 0711 AE
1181 0712 87
1182 0713 F5
1183 0714 33 14
1184 0716 F8 FF
1185 0718 FC 01
1186 071A 52
1187 071B F6 03
1188 071D F5
1189 071F 38 00
1190 0720 32 00

LVDT:
1145 0501 F0
1146 0632 32 9A
1147 0694 87
1148 0695 AC
1149 0696 97
1150 0697 BC
1151 0698 30 59
1152 0699 8F
1153 069B 52
1154 069C 9F
1155 069D F5
1156 069F AA
1157 069F F8 00
1158 06A1 8A
1159 06A2 8B
1160 06A3 8A
1161 06A4 FF 08
1162 06A6 89
1163 06A7 30 00
1164 0700
1165 0700
1167 0701 6F
1168 0702 A7
1169 0703 8F
1170 0704 F3
1171 0705 32 00
1172 0707 8A
1173 0708 8B
1174 0709 F8 01
1175 0709 F6
1176 070C F8 09
1177 070E 8A
1178 070F 08
1179 0710 52
1180 0711 AE
1181 0712 87
1182 0713 F5
1183 0714 33 14
1184 0716 F8 FF
1185 0718 FC 01
1186 071A 52
1187 071B F6 03
1188 071D F5
1189 071F 38 00
1190 0720 32 00

LM:
1145 0501 F0
1146 0632 32 9A
1147 0694 87
1148 0695 AC
1149 0696 97
1150 0697 BC
1151 0698 30 59
1152 0699 8F
1153 069B 52
1154 069C 9F
1155 069D F5
1156 069F AA
1157 069F F8 00
1158 06A1 8A
1159 06A2 8B
1160 06A3 8A
1161 06A4 FF 08
1162 06A6 89
1163 06A7 30 00
1164 0700
1165 0700
1167 0701 6F
1168 0702 A7
1169 0703 8F
1170 0704 F3
1171 0705 32 00
1172 0707 8A
1173 0708 8B
1174 0709 F8 01
1175 0709 F6
1176 070C F8 09
1177 070E 8A
1178 070F 08
1179 0710 52
1180 0711 AE
1181 0712 87
1182 0713 F5
1183 0714 33 14
1184 0716 F8 FF
1185 0718 FC 01
1186 071A 52
1187 071B F6 03
1188 071D F5
1189 071F 38 00
1190 0720 32 00

LDX
RZ GG
GLD R7
PLO RC
GHI R7
PHI RC
PR PP
GLD RE
STR R2
GHI RE
SD
PLO RA
LDI 00H
PLN RB
PHI RB
PHI RA
LDI A.1(FLODAT)
PHI R9
BR RET3
PAGE
RETURN
INP 7
PLO R7
GLO RE
XDR
BZ RET7
GLD RA
PLN RB
LDI 01H
SHR
LDI A.1(VTDAT)
PHI RB
LDN RB
STR R2
PLO RE
GLO R7
SD
BDF LM
XPI OFFH
ADI 01H
STR R2
LDI 03H
SD
BNF RET7
BZ RET7

..IF DIFFERENCE = 0, JUMP TO GG
..STORE NEW DIFF. IN OLD DIFF. HOLDING AREA
..
..JUMP TO PP
..
..SUBTRACT VSD INCREMENT FROM LATEST MODIFIED..
..LOWER VSD
..STORE INTERPOLATED VSD IN RA.0
..PUT #00 IN RB AND RA.1
..
..
..LOAD ADDRESS OF FLOW RATE TABLE..
..BACK INTO R9
..RETURN TO CALLING PROGRAM
..
..**VALVE STEM POSITION SENSING SUBROUTINE**
..GET INPUT OF POSITION SENSOR
..STORE IN R7.0
..GET STORED VALUE OF PREVIOUS POSITION
..COMPARE CURRENT VALUE TO PREVIOUS VALUE
..IF EQUAL, RETURN TO CALLING PROGRAM
..STORE INTERPOLATED VSD IN ..
..RB.0
..SET DF = #01
..
..LOAD POSITION SENSING DATA TABLE ADDRESS..
..INTO RB
..
..LOAD TABLE READING OF CURRENT VSD INTO RE.0
..SUBTRACT FROM THIS ACTUAL POSITION READING
..IF DIFFERENCE > 0, JUMP TO LM
..TAKE THE 2'S COMPLIMENT OF THE RESULT
..
..STORE DIFFERENCE ON STACK
..
..SUBTRACT #03 FROM DIFFERENCE
..IF RESULT < 0, RETURN TO CALLING PROGRAM
..IF RESULT = 0, RETURN TO CALLING PROGRAM

```

ER LINE ADDR B1 B2 B3 B4

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```

1191 0722 8A      GLD RA
1192 0723 A9      PLO R9
1193 0724 FF 10   LDI 010H
1194 0726 FF      PHI RF
1195 0727         CALL1 OUTPUT,044H
1199 072A         CALL1 OUTPUT,04H
1203 072F FR 00   LDI 00H
1204 0731 FR      PHI RB
1205 0732 AP      PLO R8
1206 0733 89      GLO R9
1207 0734 32 39   RZ LK
1208 073A         CALL CLOSE
1211 0739 FF 18   LDI 18H
1212 073B BF      PHI RF
1213 073C         CALL1 OUTPUT,014H
1217 0740         CALL1 OUTPUT,04H
1221 0744 FR 10   LDI 010H
1222 0746 BF      PHI RF
1223 0747         CALL OPEN
1226 074A         CALL1 OUTPUT,084H
1230 074E         CALL1 OUTPUT,04H
1234 0752 30 00   BR RET7
1235 0750         PAGE
1236 0800 OF 00 00 00 00 00 00 00 00 00 00 00 00 00 00 00 00
1237 0P04 46 00 87
1238 0R07 00 RQ 00 E6
1239 0R0R 01 OE
1240 0P0D 01 36 01 54
1241 0R11 C1 72
1242 0P13 01 R6 01 9A
1243 0P17 01 A9
1244 0R19 01 RD 01 CC
1245 01D 01 D6
1246 0P1F 01 E0 01 E0
1247 0123 01 F0
1248 0P25 01 E0 01 E0
1249 0P22 01 E0
1250 0R2R 17 00 00 00
1251 0R2F 69 00 84
1252 0R32 00 FA 01 40
1253 0R36 01 7C
1254 0R3R 01 A9 01 D6
1255 0R3C 01 FE
1256 0R3E 02 1C 02 3A
1257 0842 02 4F

1191 0722 8A      **STORE VSD IN R9.0
1192 0723 A9      **LOAD #10HEX. INTO RF.1 FOR COUNTDOWN
1193 0724 FF 10   **
1194 0726 FF      **CLOSE UPSTREAM SOLENOID
1195 0727         **CUT OFF POWER TO ACTUATOR
1199 072A         **SET RB = #0000
1203 072F FR 00   **
1204 0731 FR      **CHECK TO SEE IF VSD = 0
1205 0732 AP      **IF SO, JUMP TO LK
1206 0733 89      **CLOSE VALVE COMPLETELY
1207 0734 32 39   **LOAD #18HEX. INTO RF.1 FOR COUNTDOWN
1208 073A         **
1211 0739 FF 18   **RESET LINEAR ENCODER INTERFACE CIRCUIT
1212 073B BF      **CUT OFF POWER TO ACTUATOR
1213 073C         **LOAD #10HEX. INTO RF.1 FOR COUNTDOWN
1217 0740         **
1221 0744 FR 10   **OPEN VALVE TO CORRECT DISPLACEMENT
1222 0746 BF      **OPEN UPSTREAM SOLENOID
1223 0747         **CUT OFF POWER TO ACTUATOR
1226 074A         **RETURN TO CALLING PROGRAM
1230 074E         **
1234 0752 30 00   **
1235 0750         **
1236 0800 OF 00 00 00 00 00 00 00 00 00 00 00 00 00 00 00 00
1237 0P04 46 00 87
1238 0R07 00 RQ 00 E6
1239 0R0R 01 OE
1240 0P0D 01 36 01 54
1241 0R11 C1 72
1242 0P13 01 R6 01 9A
1243 0P17 01 A9
1244 0R19 01 RD 01 CC
1245 01D 01 D6
1246 0P1F 01 E0 01 E0
1247 0123 01 F0
1248 0P25 01 E0 01 E0
1249 0P22 01 E0
1250 0R2R 17 00 00 00
1251 0R2F 69 00 84
1252 0R32 00 FA 01 40
1253 0R36 01 7C
1254 0R3R 01 A9 01 D6
1255 0R3C 01 FE
1256 0R3E 02 1C 02 3A
1257 0842 02 4F

FLD DAT: DC OFH, 00H, 00H, 00H, 46H, 00H, 87H      **FLOW RATE DATA TABLE..15 PSI LINE
DC 00H, 089H, 00H, 0E6H, 01H, 0EH
DC 01H, 36H, 01H, 54H, 01H, 72H
DC 01H, 86H, 01H, 9AH, 01H, 0A9H
DC 01H, 08DH, 01H, 0CCH, 01H, 0D6H
DC 01H, 0E0H, 01H, 0E0H, 01H, 0E0H
DC 01H, 0E0H, 01H, 0E0H, 01H, 0E0H
DC 17H, 00H, 00H, 00H, 69H, 00H, 084H      **23 PSI LINE
DC 00H, 0FAH, 01H, 40H, 01H, 7CH
DC 01H, 0A9H, 01H, 0D6H, 01H, 0FEH
DC 02H, 1CH, 02H, 3AH, 02H, 4EH

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ER	LINE	ADDR	B1 B2 B3 B4
	1258	0A44	02 6C 02 78
	1259	0A48	02 8A
	1260	0A4A	02 90 02 A3
	1261	0A4F	02 A8
	1262	0B50	02 A8 02 A8
	1263	0A54	02 A8
	1264	0B56	1F 00 00 00
	1265	0A5A	AC 00 E6
	1266	0A5D	01 45 01 9A
	1267	0A61	01 E0
	1268	0A63	02 21 02 58
	1269	0A67	02 A5
	1270	0A69	02 AD 02 D0
	1271	0A6D	02 F9
	1272	0A6F	03 07 03 20
	1273	0A73	03 34
	1274	0A75	03 43 03 52
	1275	0A79	03 5C
	1276	0A7A	03 5C 03 5C
	1277	0A7F	03 5C
	1278	0A81	27 00 00 00
	1279	0A85	AF 01 1D
	1280	0A88	01 88 01 F4
	1281	0A9C	02 44
	1282	0A8F	02 94 02 D5
	1283	0A92	03 0C
	1284	0A94	03 34 03 5C
	1285	0A98	03 7F
	1286	0A9A	03 A2 03 88
	1287	0A9F	03 D4
	1288	0A80	03 E8 03 F7
	1289	0A84	04 01
	1290	0A86	04 01 04 01
	1291	0A8A	04 01
	1292	0A8C	2F 00 00 00
	1293	0A8D	07 01 4F
	1294	0A93	01 01 02 3F
	1295	0A87	02 A8
	1296	0A89	02 FD 03 4D
	1297	0A8D	03 80
	1298	0A8F	03 88
	1299	0A83	04 10
	1300	0A85	04 38 04 56
	1301	0A89	04 74
	1302	0A8B	04 88 04 9C

DC	02H,6CH,02H,78H,02H,8AH	
DC	02H,90H,02H,0A3H,02H,0A8H	
DC	02H,0A3H,02H,0A8H,02H,0A8H	
DC	1FH,00H,00H,00H,8CH,00H,0E6H	..31 PSI LINE
DC	01H,45H,01H,9AH,01H,0E0H	
DC	02H,21H,02H,58H,02H,85H	
DC	02H,0ADH,02H,0D0H,02H,0E9H	
DC	03H,07H,03H,20H,03H,34H	
DC	03H,43H,03H,52H,03H,5CH	
DC	03H,5CH,03H,5CH,03H,5CH	
DC	27H,00H,00H,00H,0AFH,01H,1DH	..39 PSI LINE
DC	01H,88H,01H,0F4H,02H,44H	
DC	02H,94H,02H,0D5H,03H,0CH	
DC	03H,34H,03H,5CH,03H,7FH	
DC	03H,0A2H,03H,088H,03H,0D4H	
DC	03H,0E8H,03H,0F7H,04H,01H	
DC	04H,01H,04H,01H,04H,01H	
DC	2FH,00H,00H,00H,007H,01H,4FH	..47 PSI LINE
DC	01H,0D1H,02H,3FH,02H,0A8H	
DC	02H,0FDH,03H,4DH,03H,89H	
DC	03H,088H,03H,0E8H,04H,1DH	
DC	04H,38H,04H,56H,04H,74H	
DC	04H,88H,04H,9CH,04H,0A1H	

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```

ER  LINE ADDR  B1 B2 B3 B4
1303 08CF 04 A1
1304 0801 04 A1 04 A1
1305 0875 04 A1
1306 0900
1307 0900 00 02 04 05
1308 0804 07 08 0C 0E
1309 0908 10 13 15 18
1310 090C 1A 1D 20 22
1311 0910 25 27 2A 2C
1312 0914 2E 31 33 36
1313 0918 3A 3D 40
1314 091C 42 45 47 4A
1315 0920 4C 4E 51 53
1316 0924 55 58 5A 5D
1317 0928 5F 61 64 66
1318 092C 6A 6C 6E
1319 0930 70 72 75 77
1320 0934 79 7B 7F 80
1321 0938 82 85 89 8A
1322 093C 8D 8F 92 94
1323 0940 96 98 9A 9C
1324 0944 9E 9F A0 A0
1325 0948 A1 A2 A3 A3
1326 094C A4 A5 A6 A6
1327 0950 A7
1328 0951

```

DC 04H,0A1H,04H,0A1H,04H,0A1H

PAGE

VTDAT: DC 0CH,02H,04H,05H,07H,09H,0CH,0EH ..DISPLACEMENT DATA TABLE

DC 10H,13H,15H,18H,1AH,1DH,20H,22H

DC 25H,27H,2AH,2CH,2EH,31H,33H,36H

DC 38H,3BH,3DH,40H,42H,45H,47H,4AH

DC 4CH,4EH,51H,53H,55H,58H,5AH,5DH

DC 5FH,61H,64H,66H,68H,6AH,6CH,6EH

DC 70H,72H,75H,77H,79H,7BH,7EH,80H

DC 82H,85H,88H,8AH,8DH,8FH,92H,94H

DC 96H,98H,9AH,9CH,9EH,9FH,0A0H,0A0H

DC 0A1H,0A2H,0A3H,0A4H,0A5H,0A6H,0A6H,0A7H

END

ASSEMBLER ERRORS = 0

APPENDIX D

DATA

Contained in the following pages are:

Table D.1 Preliminary calibration data

Table D.2 Linear encoder calibration table

Table D.1 Preliminary calibration data

$P_u = 30$ psig				$P_u = 40$ psig				$P_u = 50$ psig				$P_u = 60$ psig				$P_u = 70$ psig			
x_s	Δp	Q_o	L_s	Δp	Q_o	L_s	(steps) (psig) (SCCS) (volts)	Δp	Q_o	L_s	(steps) (psig) (SCCS) (volts)	Δp	Q_o	L_s	(steps) (psig) (SCCS) (volts)	Δp	Q_o	L_s	(steps) (psig) (SCCS) (volts)
0	28	0	11.28	38	0	11.27		49	0	11.27		59	0	11.33	69	0	11.28		
4	28	241	11.32	38	293	11.31		47	359	11.32		57	411	11.37	67	496	11.32		
8	26	481	11.39	35	599	11.38		44	732	11.38		54	901	11.44	62	1038	11.39		
12	24	698	11.46	32	883	11.45		41	1067	11.45		49	1270	11.51	57	1430	11.46		
16	21	883	11.53	29	1128	11.52		36	1359	11.52		44	1576	11.57	51	1784	11.53		
20	19	1005	11.59	26	1298	11.58		33	1562	11.59		39	1812	11.64	45	2058	11.59		
24	17	1114	11.66	23	1416	11.66		29	1713	11.66		35	1996	11.71	40	2270	11.67		
28	15	1189	11.73	20	1506	11.72		26	1798	11.73		31	2133	11.78	36	2426	11.73		
32	13	1237	11.80	18	1576	11.79		23	1888	11.80		28	2228	11.84	33	2534	11.80		
36	12	1284	11.86	17	1624	11.86		21	1949	11.86		26	2289	11.90	30	2596	11.87		
40	12	1312	11.92	16	1652	11.92		20	1996	11.92		24	2331	11.96	28	2657	11.93		
44	11	1326	11.98	15	1680	11.98		19	2025	11.98		23	2364	12.02	26	2700	11.99		
48	10	1340	12.04	14	1713	12.04		18	2044	12.04		22	2412	12.09	25	2733	12.05		
52	10	1359	12.10	14	1713	12.10		17	2072	12.10		21	2440	12.15	24	2761	12.11		
56	10	1373	12.16	13	1727	12.15		16	2086	12.16		20	2440	12.20	23	2794	12.17		
60	9	1373	12.21	13	1741	12.21		16	2086	12.22		19	2440	12.25	22	2808	12.22		
64	9	1388	12.27	12	1756	12.26		15	2105	12.26		19	2473	12.30	21	2822	12.28		
68	9	1388	12.32	12	1770	12.32		15	2119	12.32		18	2487	12.35	21	2841	12.33		
72	8	1388	12.37	12	1770	12.37		15	2133	12.37		18	2501	12.40	20	2841	12.38		
76	8	1388	12.38	12	1770	12.38		15	2133	12.39		18	2501	12.41	20	2841	12.38		
80	8	1388	12.39	12	1770	12.39		15	2133	12.39		18	2501	12.42	20	2855	12.39		

Table D.2 Linear encoder calibration table

x_s	E_s	x_s	E_s
(steps)	(counts)	(steps)	(counts)
00	00	41	97
01	02	42	100
02	04	43	102
03	05	44	104
04	07	45	106
05	09	46	108
06	12	47	110
07	14	48	112
08	16	49	114
09	19	50	117
10	21	51	119
11	24	52	121
12	26	53	123
13	29	54	126
14	32	55	128
15	34	56	130
16	37	57	133
17	39	58	136
18	42	59	138
19	44	60	141
20	46	61	143
21	49	62	146
22	51	63	148
23	54	64	150
24	56	65	152
25	59	66	154
26	61	67	156
27	64	68	158
28	66	69	159
29	69	70	160
30	71	71	160
31	74	72	161
32	76	73	162
33	78	74	163
34	81	75	163
35	83	76	164
36	85	77	165
37	88	78	166
38	90	79	166
39	93	80	167
40	95		

APPENDIX E

SYSTEM STABILITY

The developed assembly was a static system and therefore inherently stable in response to changes in the requested volumetric flow rate. There were two reasons for this design:

- (1) Because of the indirect control techniques used, a finite amount of computation time was required by the microprocessor system to calculate the valve stem displacement for each command change. In classical controls, such a delay in the response could cause a system to become unstable [9]. Therefore, the developed system was designed to operate in a non-continuous mode. The control algorithm was written such that during computation, the actuator position was held stationary and the microprocessor system would not accept any more inputs. Thus, the developed system is considered static.
- (2) As mentioned in Chap. 3, an upstream assembly was added to interrupt the flow while the valve stem was repositioned. This was because the actuator was not capable of supplying sufficient force to move the valve stem as long as there was flow through the valve. This was the major reason for designing the assembly as a static system.

It should be noted that in cases where the requested volumetric flow rate remained unchanged, the system would reposition in the presence of upstream pressure fluctuations. Due to the long computational time requirement, the assembly would not be able to respond to any fast changing pressure fluctuation. However, it is to be expected that the source pressure fluctuation would be minimal. It was noticed that perturbations in the source pressure, due to the repositioning, were present in the system; but were damped out before the end of the control algorithm cycle.